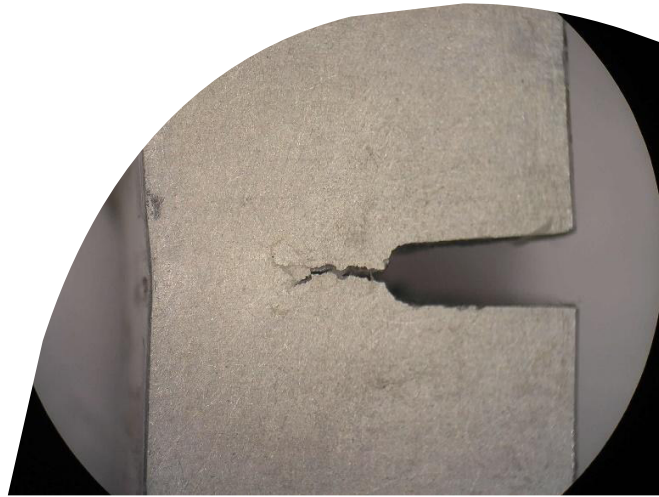
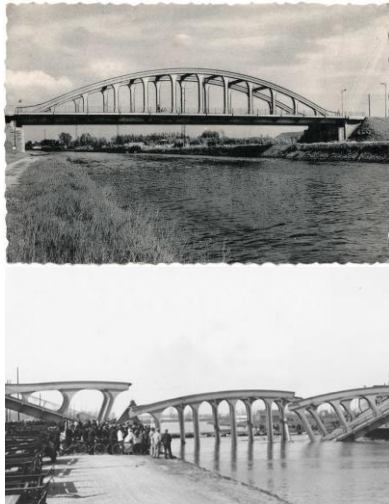




**EUROPEAN CONVENTION
FOR CONSTRUCTIONAL
STEELWORK
Bridge Committee**



**How to properly select the steel
grade concerning toughness
requirements and how to extend
the fatigue life of steel and steel
composite bridges**

ECCS Bridge Committee

Mike Tibolt, Francesco Profico, Mohammad Al-Emrani, Zuheir Barsoum,
Bertram Kühn, Pavel Ryjáček

PREFACE

An ECCS webinar was held on 4 November 2025 and gives the basis for this eBook that focuses on the on toughness and steel grade selection in prEN1993-1-10, that is included now in the 2nd generation of Eurocodes and explain the TTV tool, that can be for this application used. It also explains the basis of the HFMI (High Frequency Mechanical Impact) treatment, describe the advantages and possibilities of this method and give a practical example of the structural analysis of the bridge, designed with this method and related savings. Also, the quality control methods for the HFMI are described and explained.

A handwritten signature in blue ink, appearing to read 'Pavel Ryjáček', is positioned above the printed name.

Pavel Ryjáček
Bridge committee ECCS, Chair

CONTENTS

1. Choice of steel grade concerning toughness requirements: Background and TTV tool demonstration,
(Mike Tibolt & Francesco Profico, ArcelorMittal)
2. Fatigue life improvement of welded bridge details using High Frequency Mechanical Impact treatment
(Mohammad Al-Emrani, Chalmers University of Technology)
3. HFMI for Fatigue Life Extension: Guidelines, Digital QA, and Bridge Applications
(Zuheir Barsoum, KTH Royal Institute of Technology)
4. The development and changes in prEN1993-1-10
(Bertram Kühn, Technische Hochschule Mittelhessen)

1. CHOICE OF STEEL GRADE CONCERNING TOUGHNESS REQUIREMENTS: BACKGROUND AND TTV TOOL DEMONSTRATION, (MIKE TIBOLT & FRANCESCO PROFICO)

ArcelorMittal Europe



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EKS**



ArcelorMittal

Choice of steel grade concerning toughness requirements:

Background and TTV tool demonstration

Mike Tibolt

Francesco Profico

4 November 2025



Speakers



Mike TIBOLT

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Senior Structural Engineer, Steligen Engineering

Mostly active in bridges, lattice towers and fracture mechanics problems.

Member standardization committee EN1993-1-9 and EN1993-1-10.



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Structural Engineer, Steligen Engineering

Mostly active in bridges and composite structures.

EN 1993-1-10:2005

**Background of the verification method
against brittle fracture
& Concept of TTV Tool**

Brittle fracture – Discovering the phenomena



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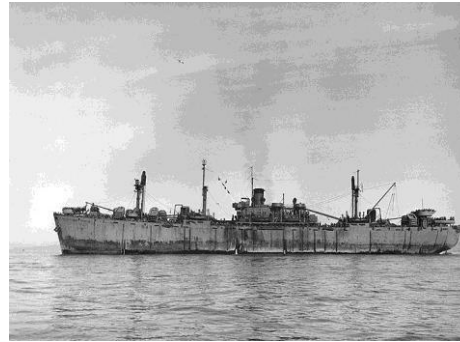


1919-1945: Sudden structural failures engaged by crack formation in different steel structures

Pont Hasselt, Belgium, 1938



Liberty ships, US, 1941-45



Molasse tank, Boston, US, 1919



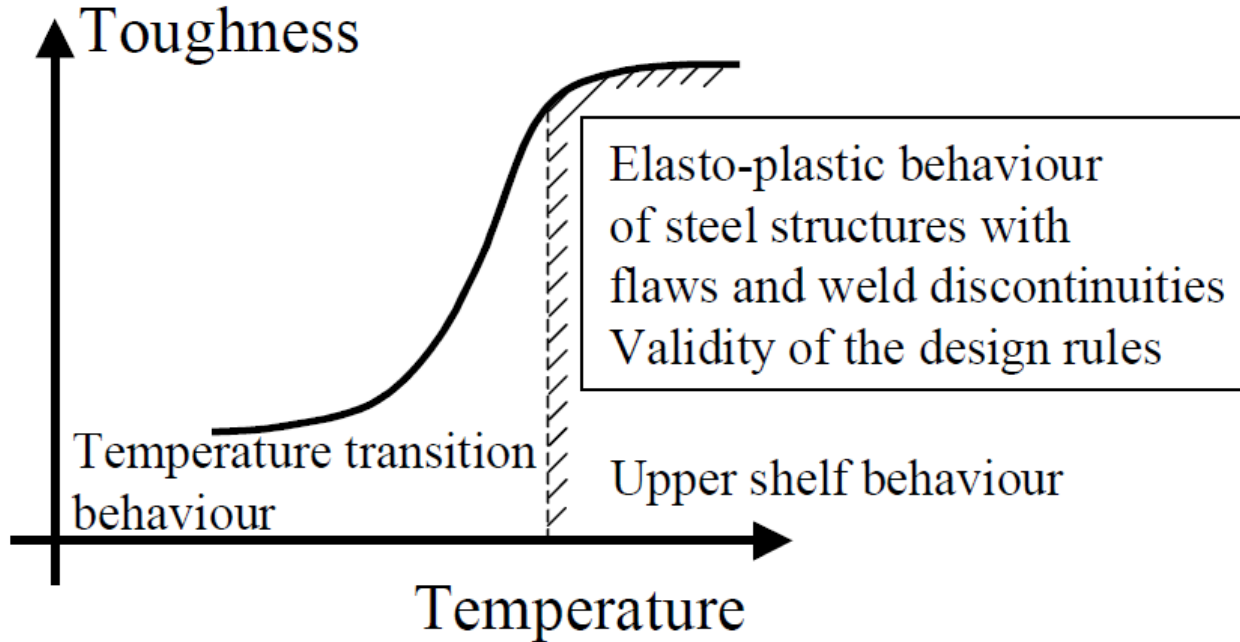
Toughness – The key property to avoid brittle fracture



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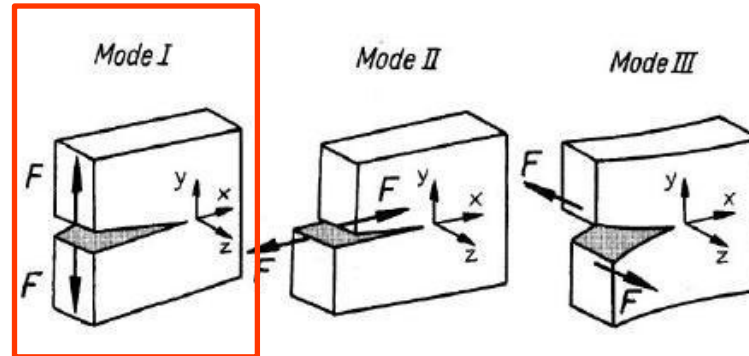


Verification according to LEFM:

$$K_{\text{appl,d}}^* \leq K_{\text{mat,d}}(T_{\text{Ed}})$$

$K_{\text{appl,d}}^*$ Stress intensity factor, mode I [MPam^{1/2}]

$K_{\text{mat,d}}(T_{\text{Ed}})$ Fracture toughness [MPam^{1/2}]



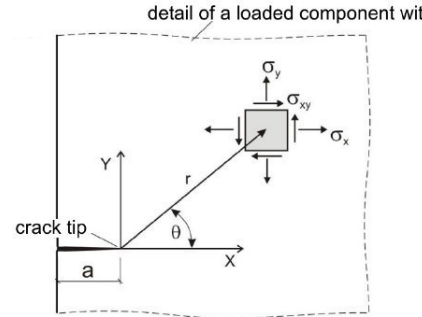


Verification according to LEFM:

$$K^*_{\text{appl,d}} \leq K_{\text{mat,d}}(T_{\text{Ed}})$$

Stress intensity factor $K^*_{\text{appl,d}}$:

$$K^*_{\text{appl,d}} = \frac{\sigma_{\text{Ed}} \cdot \sqrt{\pi a_d} \cdot Y \cdot M_K}{k_{R6} - \rho}$$



$$\sigma_x = \sigma \sqrt{\frac{a}{2r}} \cdot \cos \frac{\theta}{2} \left[1 - \left(\sin \frac{\theta}{2} \cdot \sin \frac{3\theta}{2} \right) \right] = \frac{K}{\sqrt{2\pi r}} \cdot f_x(\theta)$$

$$\sigma_y = \sigma \sqrt{\frac{a}{2r}} \cdot \cos \frac{\theta}{2} \left[1 + \left(\sin \frac{\theta}{2} \cdot \sin \frac{3\theta}{2} \right) \right] = \frac{K}{\sqrt{2\pi r}} \cdot f_y(\theta)$$

$$\sigma_{xy} = \sigma \sqrt{\frac{a}{2r}} \cdot \cos \frac{\theta}{2} \cdot \sin \frac{\theta}{2} \cdot \cos \frac{3\theta}{2} = \frac{K}{\sqrt{2\pi r}} \cdot f_{xy}(\theta)$$

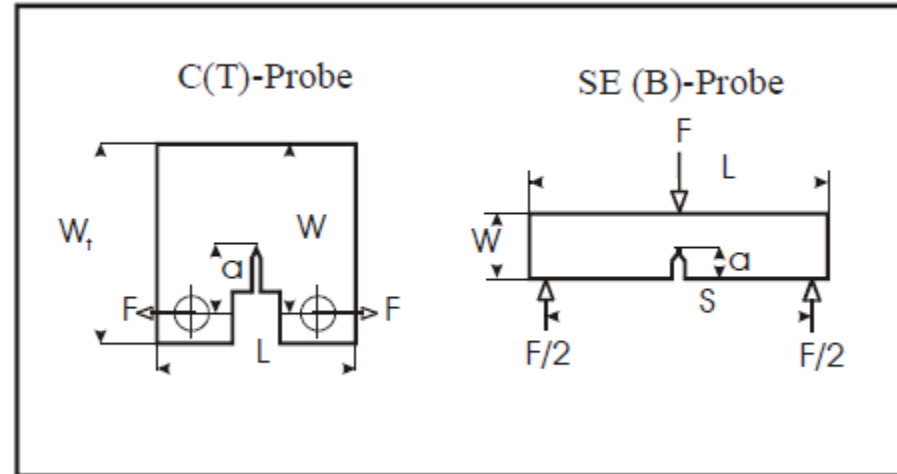
| | | |
|----------------------|--|-------|
| σ_{Ed} | Nominal design stress in the member | [MPa] |
| a_d | Design size of the crack | [mm] |
| Y | Correction function for crack shape and position | [-] |
| M_K | Correction function for structural detail | [-] |
| k_{R6} | Plasticity correction factor | [-] |
| ρ | Factor for local residual stresses | [-] |

Verification according to LEFM:

$$K_{\text{appl,d}}^* \leq K_{\text{mat,d}}(T_{\text{Ed}})$$

Fracture toughness $K_{\text{mat,d}}(T_{\text{Ed}})$:

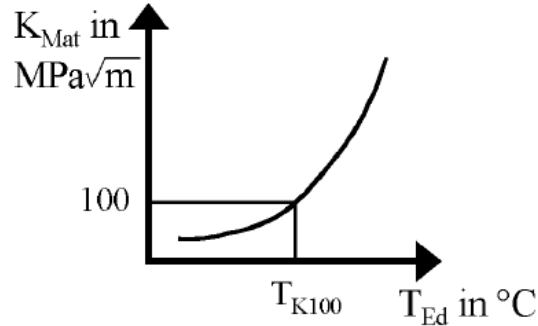
- Material property
- Temperature dependent
- Determination by tests



Practical verification method is required

1. Step: Master-Curve approach from Wallin

- Shape of the fracture toughness **transition** curve practically independent from material and yield strength
- Description as a function of temperature, specimen thickness and fracture probability
- Fracture probability of 50% -> mean value approach



$$K_{Mat}(T_{Ed}) = 20 + [70 \left\{ \exp \frac{T_{Ed} - T_{K100} + \Delta T_R}{52} \right\} + 10] \cdot \left(\frac{25}{b_{eff}} \right)^{1/4}$$

T_{Ed} Test temperature [$^{\circ}\text{C}$]

T_{K100} Test temperature at which the measured fracture toughness is $100 \text{ MPa}\sqrt{\text{m}}$ [$^{\circ}\text{C}$]

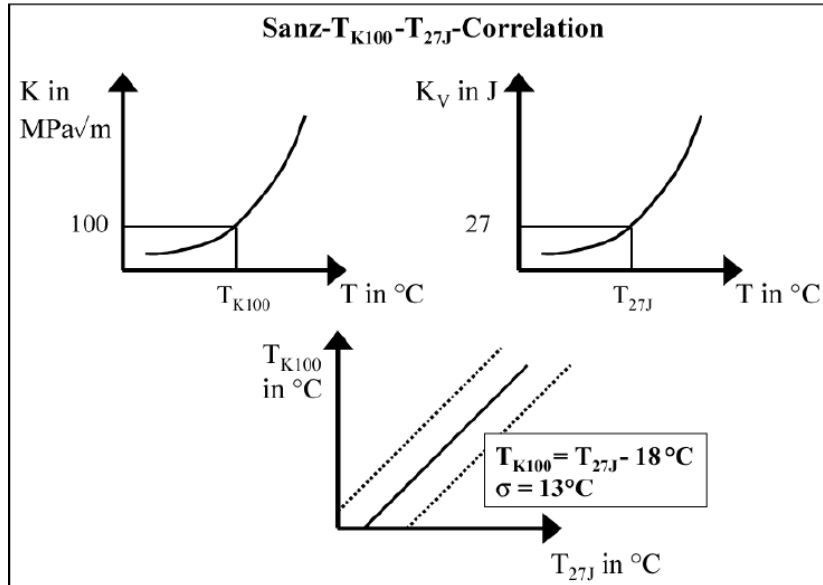
ΔT_R Additive safety element [$^{\circ}\text{C}$]

$\left(\frac{25}{b_{eff}} \right)^{1/4}$ Effect of the crack front on the failure probability [$^{\circ}\text{C}$]



2. Step: Correlation of Sanz

- Correlation between T_{K100} and T_{27J}



$$20 + [70 \left\{ \exp \frac{T_{Ed} - T_{K100} + \Delta T_R}{52} \right\} + 10] \cdot \left(\frac{25}{b_{eff}} \right)^{1/4}$$



$$T_{K100} = T_{27J} - 18^{\circ}\text{C}$$

$$\sigma = 13^{\circ}\text{C}$$

$$20 + [70 \left\{ \exp \frac{T_{Ed} - T_{27J} + 18^{\circ}\text{C} + \Delta T_R}{52} \right\} + 10] \cdot \left(\frac{25}{b_{eff}} \right)^{1/4}$$



3. Step: Expression in terms of temperatures

$$K^*_{\text{appl},d} \leq K_{\text{mat},d}(T_{\text{Ed}})$$

$$K^*_{\text{appl},d} = \frac{\sigma_{\text{Ed}} \cdot \sqrt{\pi a_d} \cdot Y \cdot M_K}{k_{R6-\rho}} \leq 20 + [70 \left\{ \exp \frac{T_{\text{Ed}} - T_{27J} + 18^\circ\text{C} + \Delta T_R}{52} \right\} + 10] \cdot \left(\frac{25}{b_{\text{eff}}} \right)^{1/4}$$

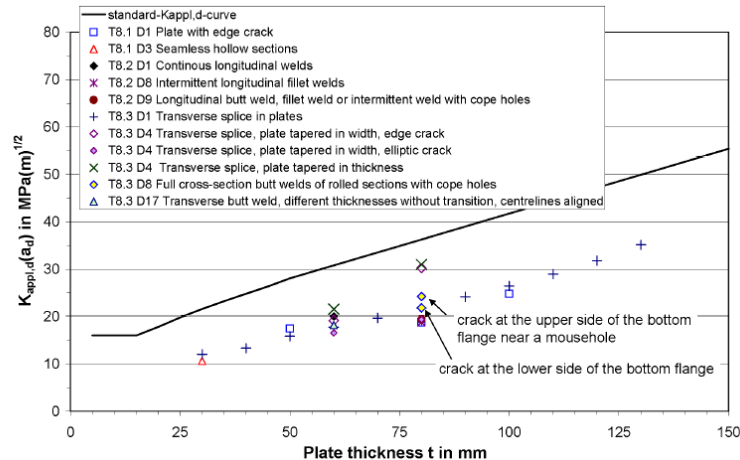
$$T_{\text{Ed}} - 52 \cdot \ln \left(\frac{(K^*_{\text{appl},d} - 20) \cdot \left(\frac{b_{\text{eff}}}{25} \right)^{0,25} - 10}{70} \right) + \Delta T_R \geq T_{27J} - 18^\circ\text{C}$$

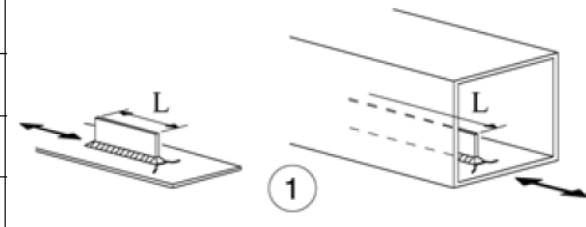
ΔT_σ :
function of the detail geometry
and
crack shape

ΔT_R :
Additive safety
element

4. Step: Definition of a standard detail and crack shape

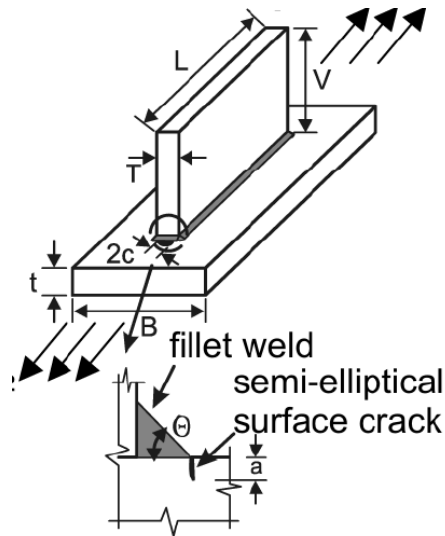
- Worst detail from EN 1993-1-9 and crack shape with regards to stress intensity factor $K_{appl,d}^*$
- Detail 1 in Table 4.1 in EN1993-1-9, Detail category 56



| Detail category | Constructional detail | |
|-----------------|----------------------------|---|
| 80 | $L \leq 50\text{mm}$ |  |
| 71 | $50 < L \leq 80\text{mm}$ | |
| 63 | $80 < L \leq 100\text{mm}$ | |
| 56 | $L > 100\text{mm}$ | |

4. Step: Definition of a standard detail and crack shape

- Standardization of the geometry and crack shape and size



$$L/t = 8,20$$

$$a_0/c_0 = 0,40$$

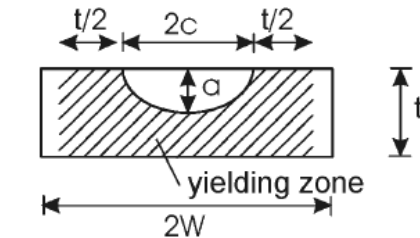
$$T/t = 0,15$$

$$a_0 = 0,5 \cdot \ln \left(1 + \frac{t}{t_0} \right), t < 15 \text{ mm}, t_0 = 1 \text{ mm}$$

$$B/t = 7,50$$

$$a_0 = 0,5 \cdot \ln \left(\frac{t}{t_0} \right), t \geq 15 \text{ mm}, t_0 = 1 \text{ mm}$$

$$\Theta = 45^\circ$$



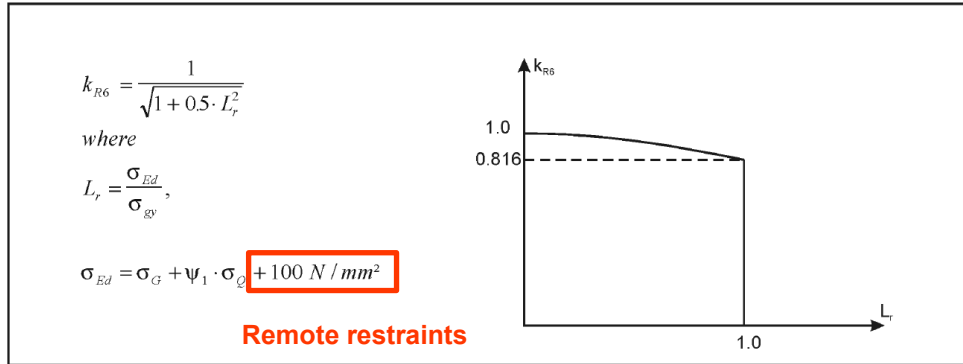
4. Step: Definition of a standard detail and crack shape

- Determination of $\Delta T_{\sigma} = 52 \cdot \ln \left(\frac{(K_{appl,d}^* - 20) \cdot \left(\frac{b_{eff}}{25}\right)^{0,25} - 10}{70} \right)$ with $K_{appl,d}^* = \frac{\sigma_{Ed} \cdot \sqrt{\pi a_d} \cdot Y \cdot M_K}{k_{R6} - \rho}$

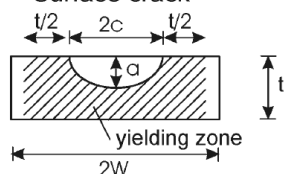
| Case | Function M_k | Source | Case | Function Y | Source | |
|---|--|-----------|--|---|---|--|
| <p>fillet weld semielliptical surface crack</p> | $M_k = C \cdot \left(\frac{a}{t}\right)^k$ and $M_k \geq 1$ | Hobbacher | <p>Surface crack</p> | $Y = \frac{F_s}{\sqrt{Q}}$ $Q = 1 + 1464 \left(\frac{a}{c}\right)^{165}$ $F_s = \left[M_1 + M_2 \cdot \left(\frac{a}{t}\right)^2 + M_3 \cdot \left(\frac{a}{t}\right)^4 \right] \cdot g \cdot f_{\phi} \cdot f_w$ | Raju - Newman | |
| <p>range of validity</p> $0.5 \leq \frac{L}{t} \leq 40$ $0.15 \leq \frac{T}{t} \leq 2$ $2.5 \leq \frac{B}{t} \leq 40$ $30^\circ \leq \Theta \leq 60^\circ$ | $C = 0.9089 - 0.2357 \cdot \frac{T}{t} + 0.0249 \cdot \left(\frac{L}{t}\right)$ $-0.00038 \left(\frac{L}{t}\right)^2 + 0.0186 \cdot \frac{B}{t} - 0.1414 \cdot \frac{\Theta}{45^\circ}$ $k = -0.02285 + 0.0167 \cdot \frac{T}{t}$ $-0.3863 \cdot \frac{\Theta}{45^\circ} + 0.1230 \cdot \left(\frac{\Theta}{45^\circ}\right)^2$ | | <p>range of validity</p> $0 \leq \frac{a}{c} \leq 1$ $\frac{2c}{B} \leq 0.5$ $0 \leq \phi \leq \pi$ $0 \leq \frac{a}{t} \leq 1$ | $M_1 = 113 - 0.09 \cdot \left(\frac{a}{c}\right)$ $M_2 = 0.5 - \frac{1}{0.65 + \frac{a}{c}}$ $M_3 = 14 \cdot \left(1 - \frac{a}{c}\right)^{24}$ $f_{\phi} = \left[\left(\frac{a}{c}\right)^2 \cdot \cos^2 \phi + \sin^2 \phi \right]^{\frac{1}{4}}$ | $M_2 = -0.54 + \frac{0.89}{0.2 + \frac{a}{c}}$ $g = 1 + \left[0.1 + 0.35 \left(\frac{a}{t}\right)^2 \right] \cdot (1 - \sin \phi)^2$ $f_w = \left[\frac{1}{\cos \left(\frac{\pi \cdot c}{B} \sqrt{\frac{a}{t}} \right)} \right]^{\frac{1}{2}}$ | |

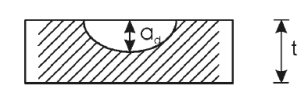
4. Step: Definition of a standard detail and crack shape

- Determination of $\Delta T_\sigma = 52 \cdot \ln \left(\frac{(K_{appl,d}^* - 20) \cdot \left(\frac{b_{eff}}{25}\right)^{0,25} - 10}{70} \right)$ with $K_{appl,d}^* = \frac{\sigma_{Ed} \cdot \sqrt{\pi a_d} \cdot Y \cdot M_K}{k_{R6} - \rho}$



| Definition of ρ | |
|---|---|
| $L_r \leq 0,8$ | $\rho = \rho_1$ |
| $0,8 \leq L_r \leq 1,05$ | $\rho = 4 \rho_1 (1,05 - L_r)$ |
| $1,05 \leq L_r$ | $\rho = 0$ |
| Definition of ρ_1 | |
| $\psi = \frac{\sigma_S L_r}{\sigma_p} \leq 0$ | $\rho_1 = 0$ |
| $\psi = \frac{\sigma_S L_r}{\sigma_p} \leq 5,2$ | $\rho_1 = 0,1 \psi^{0,714} - 0,007 \psi^2 + 0,00003 \psi^5$ |
| $\psi = \frac{\sigma_S L_r}{\sigma_p} > 5,2$ | $\rho_1 = 0,25$ |

| Case | Function Y | Source |
|--|--|----------|
| <p>Surface crack</p>  | $\sigma_{gy}(t) = f_y(t) \left(1 - \frac{\pi \cdot 2,5 \cdot a_d^2}{2 \cdot t \cdot (5 \cdot a_d + t)} \right)$ | Harrison |

| Case | b_{eff} [mm] |
|---|----------------|
| <p>Surface crack</p>  | $5 a_d$ |



4. Step: Definition of a standard detail and crack shape

- Consideration of the crack growing due to fatigue by *Paris-Erdogan law*:

$$\frac{da}{dN} \approx \frac{\Delta a}{\Delta N} = C \cdot \Delta K^m$$

$$\Delta a_i = C \cdot \Delta K_i^m \cdot \Delta N \quad \text{with} \quad \Delta K_i = \Delta \sigma \cdot \sqrt{\pi \cdot a_i} \cdot Y_i \cdot M_{K,i}$$

$$\Delta \sigma = 56 \text{ MPa}$$

$$\Delta N = 500.000$$

$$C = 1,8 \cdot 10^{-13}$$

$$m = 3,0$$

} For ferritic/pearlitic steels

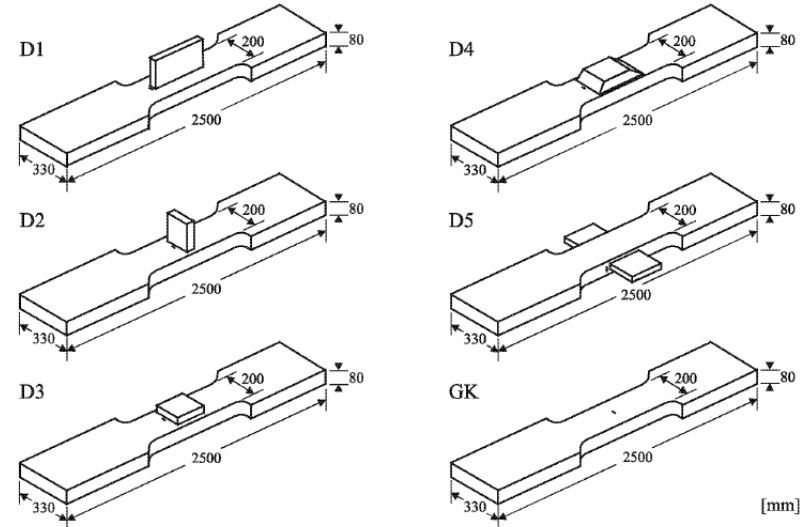
$$a_d = a_0 + \sum a_i$$

$$a_i/c_i = 0,40$$



5. Step: Global Safety level

- Statistical evaluation of large-scale tests to determine ΔT_R :
 1. Nominal values from product standards (EN 10025):
 $\Delta T_R = + 7 \text{ K}$
 2. Measured values from proper test campaign:
 $\Delta T_R = - 38 \text{ K}$
- Effect from residual stresses from welding included in ΔT_R





6. Step: Verification in the temperature domain

$$T_{Ed} - 52 \cdot \ln \left(\frac{(K_{appl,d}^* - 20) \cdot \left(\frac{b_{eff}}{25}\right)^{0,25} - 10}{70} \right) + \Delta T_R \geq T_{27J} - 18^\circ C$$

$$T_{md} + T_r + \Delta T_\sigma + \Delta T_R \geq T_{27J} - 18^\circ C$$

T_{md} lowest air temperature (see EN 1991-1-5)

ΔT_r radiation losses ($-5^\circ C$)

$$T_{md} + T_r + \Delta T_\sigma + \Delta T_R + (\Delta T_\dot{\epsilon} + \Delta T_{ecf}) \geq T_{27J} - 18^\circ C + \Delta T_t \quad \Rightarrow \quad T_{Ed} \geq T_{Rd}$$

$\Delta T_\dot{\epsilon}$ adjustment for strain rates other than $4 \cdot 10^{-4}$ /sec, EN 1993-1-10, 2.3, (2.3)

ΔT_{ecf} adjustment for cold forming, EN 1993-1-10, 2.3, (2.4), $\Delta T_{ecf} = 0$ K if DCF ≤ 2 %

ΔT_t adjustment for thick plates

Towards a temperature-based verification

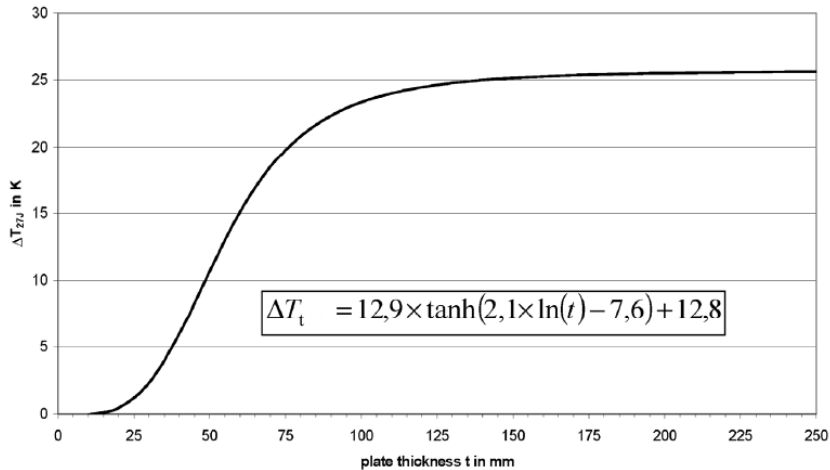


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Adjustment for thick plates ΔT_t



Applies if crack tip is located in core area

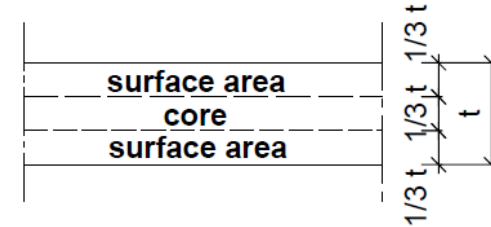




Table 2.1: Maximum permissible values of element thickness t in mm

| Steel grade | Sub-grade | Charpy energy CVN at T [°C] | J _{min} | Reference temperature T _{Ed} [°C] | | | | | | | | | | | | | | | | | | | | |
|-------------|-----------|-----------------------------|------------------|--|-----|-----|-----|-----|-----|-----|-----|-----|-----|-----|-----|-----|-----|-----|-----|-----|-----|-----|-----|-----|
| | | | | 10 | 0 | -10 | -20 | -30 | -40 | -50 | 10 | 0 | -10 | -20 | -30 | -40 | -50 | 10 | 0 | -10 | -20 | -30 | -40 | -50 |
| S235 | JR | 20 | 27 | 60 | 50 | 40 | 35 | 30 | 25 | 20 | 90 | 75 | 65 | 55 | 45 | 40 | 35 | 135 | 115 | 100 | 85 | 75 | 65 | 60 |
| | J0 | 0 | 27 | 90 | 75 | 60 | 50 | 40 | 35 | 30 | 125 | 105 | 90 | 75 | 65 | 55 | 45 | 175 | 155 | 135 | 115 | 100 | 85 | 75 |
| | J2 | -20 | 27 | 125 | 105 | 90 | 75 | 60 | 50 | 40 | 170 | 145 | 125 | 105 | 90 | 75 | 65 | 200 | 200 | 175 | 155 | 135 | 115 | 100 |
| S275 | JR | 20 | 27 | 55 | 45 | 35 | 30 | 25 | 20 | 15 | 80 | 70 | 55 | 50 | 40 | 35 | 30 | 125 | 110 | 95 | 80 | 70 | 60 | 55 |
| | J0 | 0 | 27 | 75 | 65 | 55 | 45 | 35 | 30 | 25 | 115 | 95 | 80 | 70 | 55 | 50 | 40 | 165 | 145 | 125 | 110 | 95 | 80 | 70 |
| | J2 | -20 | 27 | 110 | 95 | 75 | 65 | 55 | 45 | 35 | 155 | 130 | 115 | 95 | 80 | 70 | 55 | 200 | 190 | 165 | 145 | 125 | 110 | 95 |
| | M,N | -20 | 40 | 135 | 110 | 95 | 75 | 65 | 55 | 45 | 180 | 155 | 130 | 115 | 95 | 80 | 70 | 200 | 200 | 190 | 165 | 145 | 125 | 110 |
| | ML,NL | -50 | 27 | 185 | 160 | 135 | 110 | 95 | 75 | 65 | 200 | 200 | 180 | 155 | 130 | 115 | 95 | 230 | 200 | 200 | 200 | 190 | 165 | 145 |
| S355 | JR | 20 | 27 | 40 | 35 | 25 | 20 | 15 | 10 | 65 | 55 | 45 | 40 | 30 | 25 | 25 | 110 | 95 | 80 | 70 | 60 | 55 | 45 | |
| | J0 | 0 | 27 | 60 | 50 | 40 | 35 | 25 | 20 | 15 | 95 | 80 | 65 | 55 | 45 | 40 | 30 | 150 | 130 | 110 | 95 | 80 | 70 | 60 |
| | J2 | -20 | 27 | 90 | 75 | 60 | 50 | 40 | 35 | 25 | 135 | 110 | 95 | 80 | 65 | 55 | 45 | 200 | 175 | 150 | 130 | 110 | 95 | 80 |
| | K2,M,N | -20 | 40 | 110 | 90 | 75 | 60 | 50 | 40 | 35 | 155 | 135 | 110 | 95 | 80 | 65 | 55 | 200 | 200 | 175 | 150 | 130 | 110 | 95 |
| | ML,NL | -50 | 27 | 155 | 130 | 110 | 90 | 75 | 60 | 50 | 200 | 180 | 155 | 135 | 110 | 95 | 80 | 210 | 200 | 200 | 200 | 175 | 150 | 130 |
| S420 | M,N | -20 | 40 | 95 | 80 | 65 | 55 | 45 | 35 | 30 | 140 | 120 | 100 | 85 | 70 | 60 | 50 | 200 | 185 | 160 | 140 | 120 | 100 | 85 |
| | ML,NL | -50 | 27 | 135 | 115 | 95 | 80 | 65 | 55 | 45 | 190 | 165 | 140 | 120 | 100 | 85 | 70 | 200 | 200 | 200 | 185 | 160 | 140 | 120 |
| S460 | Q | -20 | 30 | 70 | 60 | 50 | 40 | 30 | 25 | 20 | 110 | 95 | 75 | 65 | 55 | 45 | 35 | 175 | 155 | 130 | 115 | 95 | 80 | 70 |
| | M,N | -20 | 40 | 90 | 70 | 60 | 50 | 40 | 30 | 25 | 130 | 110 | 95 | 75 | 65 | 55 | 45 | 200 | 175 | 155 | 130 | 115 | 95 | 80 |
| | QL | -40 | 30 | 105 | 90 | 70 | 60 | 50 | 40 | 30 | 155 | 130 | 110 | 95 | 75 | 65 | 55 | 200 | 200 | 175 | 155 | 130 | 115 | 95 |
| | ML,NL | -50 | 27 | 125 | 105 | 90 | 70 | 60 | 50 | 40 | 180 | 155 | 130 | 110 | 95 | 75 | 65 | 200 | 200 | 200 | 175 | 155 | 130 | 115 |
| | QL1 | -60 | 30 | 150 | 125 | 105 | 90 | 70 | 60 | 50 | 200 | 180 | 155 | 130 | 110 | 95 | 75 | 215 | 200 | 200 | 200 | 175 | 155 | 130 |
| S690 | Q | 0 | 40 | 40 | 30 | 25 | 20 | 15 | 10 | 10 | 65 | 55 | 45 | 35 | 30 | 20 | 20 | 120 | 100 | 85 | 75 | 60 | 50 | 45 |
| | Q | -20 | 30 | 50 | 40 | 30 | 25 | 20 | 15 | 10 | 80 | 65 | 55 | 45 | 35 | 30 | 20 | 140 | 120 | 100 | 85 | 75 | 60 | 50 |
| | QL | -20 | 40 | 60 | 50 | 40 | 30 | 25 | 20 | 15 | 95 | 80 | 65 | 55 | 45 | 35 | 30 | 165 | 140 | 120 | 100 | 85 | 75 | 60 |
| | QL | -40 | 30 | 75 | 60 | 50 | 40 | 30 | 25 | 20 | 115 | 95 | 80 | 65 | 55 | 45 | 35 | 190 | 165 | 140 | 120 | 100 | 85 | 75 |
| | QL1 | -40 | 40 | 90 | 75 | 60 | 50 | 40 | 30 | 25 | 135 | 115 | 95 | 80 | 65 | 55 | 45 | 200 | 190 | 165 | 140 | 120 | 100 | 85 |
| | QL1 | -60 | 30 | 110 | 90 | 75 | 60 | 50 | 40 | 30 | 160 | 135 | 115 | 95 | 80 | 65 | 55 | 200 | 200 | 190 | 165 | 140 | 120 | 100 |

T_{Ed}, σ_{Ed}, f_y(t), CVN



Table 2.1



t_{max}



t ≤ t_{max}

Interpolation but no extrapolation

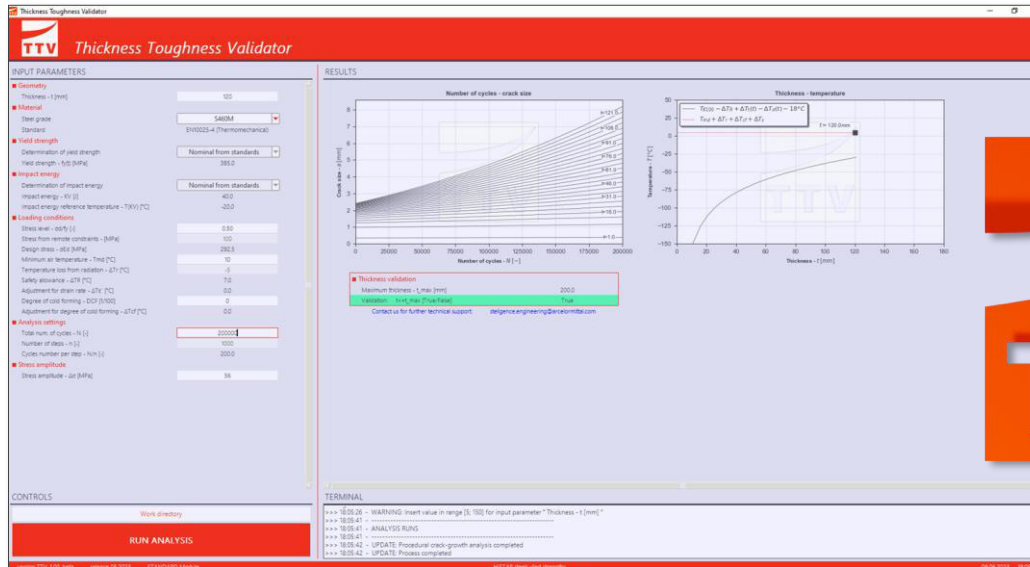
Use of TTV - Thickness Toughness Validator

A free application for toughness requirements checks according to Eurocode



Introduction

- Is a tool developed by **ArcelorMittal, Stelience Engineering**
- It allows to perform **toughness checks acc. EN 1993-1-10**
- It is available for **free online**



Scope and limitations of TTV



- Know **scopes and limits** of software and applications
- Know what the software does (**implementations + background**)

Scope and limitations of TTV



What it is conceived for

- More **advanced toughness checks compared to the “table” method**
- Requires **input which must be mastered** by the structural engineer user
- Use well-established **analytical formulas** framed by the BGD of Eurocode
- **New constructions** designed according Eurocode

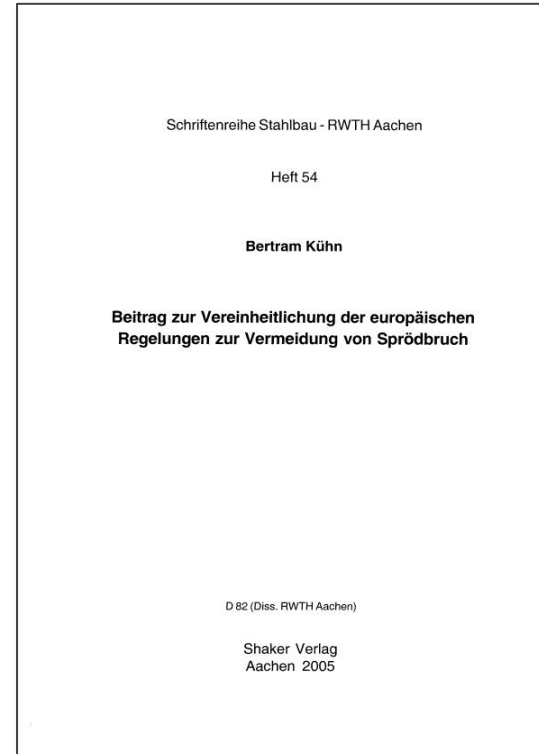
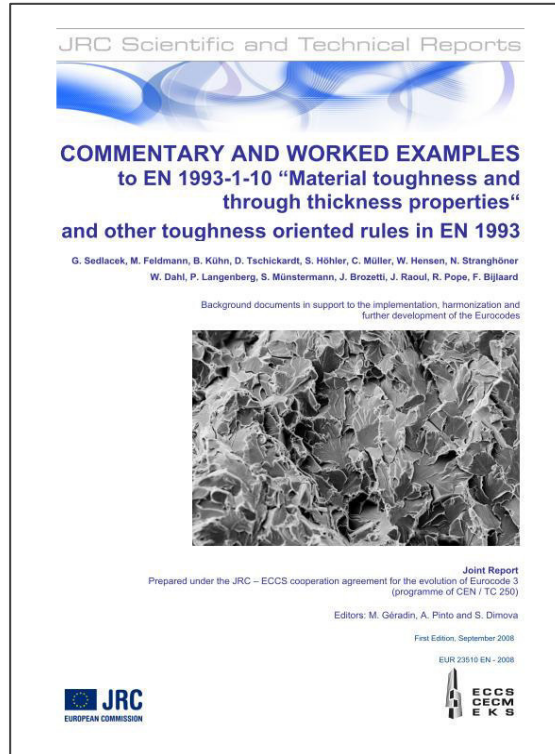


What it doesn't do

- **Steel grades not covered** by the design standards
- **Execution imperfections or cracks**
- **Existing old structures** where other safety factors may apply

Other advanced methods are possible and may deliver more refined results (e.g. *numerical simulations with explicit crack propagation in non-linear fracture mechanics*)

Technical background and references



Download and first use

1. Go to:
constructalia.arcelormittal.com/en/tools/software
2. Download the folder
3. Un-zip the folder
4. In the un-zipped folder you will find the .exe file

The screenshot shows the Constructalia website page for the TTV (Thickness Toughness Validator) software. The page title is "TTV (Thickness Toughness Validator)" and it describes the tool as being developed for validating steel construction details according to EN 1993-1-10. A blue arrow points to the download link "TTV (Thickness Toughness Validator)".


Below the website screenshot is a file explorer view showing the contents of the downloaded folder. The files and folders are listed in a table:

| Name | Date modified | Type | Size |
|-------------------------|------------------|-------------|-----------|
| database | 05/07/2023 15:38 | File folder | |
| fonts | 25/07/2023 14:54 | File folder | |
| images | 04/05/2023 16:02 | File folder | |
| ttv.exe | 26/07/2023 11:02 | Application | 86.909 KB |
| TTV_1.00-User_guide.pdf | 20/07/2023 15:27 | PDF File | 4.108 KB |

A blue arrow points to the 'ttv.exe' file in the file explorer.

Download and first use

5. Double click
6. Wait 20 seconds
7. Start window pops up



| Name | Date modified | Type | Size |
|-------------------------|------------------|-------------|-----------|
| database | 05/07/2023 15:38 | File folder | |
| fonts | 25/07/2023 14:54 | File folder | |
| images | 04/05/2023 16:02 | File folder | |
| ttv.exe | 26/07/2023 11:02 | Application | 86.909 KB |
| TTV_1.00-User_guide.pdf | 20/07/2023 15:27 | PDF File | 4.108 KB |



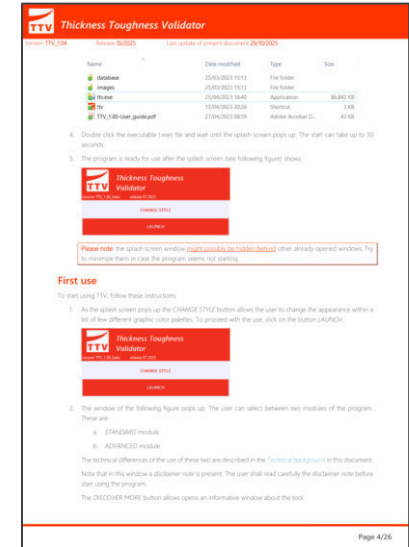
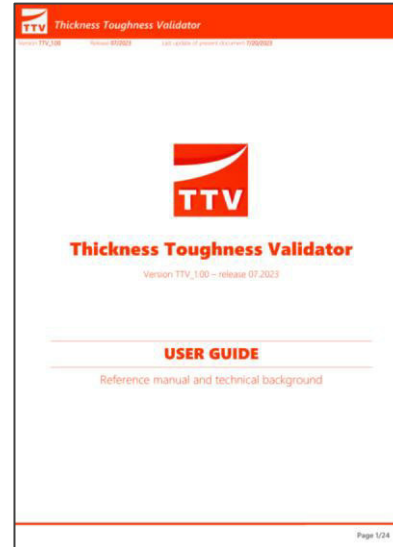
User guide

In this file you will find

- How to install
- First use
- Quick technical background
- Some examples of calculation and validation



| Name | Date modified | Type | Size |
|-------------------------|------------------|-------------|-----------|
| database | 05/07/2023 15:38 | File folder | |
| fonts | 25/07/2023 14:54 | File folder | |
| images | 04/05/2023 16:02 | File folder | |
| ttv.exe | 26/07/2023 11:02 | Application | 96,000 KB |
| TTV_1.00-User_guide.pdf | 20/07/2023 15:27 | PDF File | 4,108 KB |



Modules

The screenshot shows a window titled "Procedure selection" with two radio button options: "STANDARD - Simplified method according to EN 1993-1-10, 2.3" (selected) and "ADVANCED - Evaluation based on fracture mechanics methods (EN 1993-1-10, 2.4)". Below the options is a paragraph of text explaining the standard method. A "DISCLAIMER NOTE" section follows, detailing the tool's intended use for professional users and its limitations. At the bottom, there are two buttons: "DISCOVER MORE" and "PROCEED".

STANDARD Clause 2.3, EN1993-1-10

2.3 Maximum permitted thickness values

2.3.1 General

(1) Table 2.1 gives the maximum permissible element thickness appropriate to a steel grade, its toughness quality in terms of K_{IC} KV-value K_{IC} , the reference stress level $[\sigma_{ref}]$ and the reference temperature $[T_{ref}]$.

ADVANCED Clause 2.4, EN1993-1-10

2.4 Evaluation using fracture mechanics


(1) For numerical evaluation using fracture mechanics the toughness requirement and the design toughness property of the materials may be expressed in terms of CTOD values, J-integral values, K_{IC} values, or K_{IC} KV-values K_{IC} and comparison should be made using suitable fracture mechanics methods.

(2) The following condition for the reference temperature should be met:

$$T_{ref} \geq T_{Rd} \quad (2.7)$$

where T_{Rd} is the temperature at which a safe level of fracture toughness can be relied upon under the conditions being evaluated

Report printing


Thickness-Toughness Validator

version TTV_1.00 release 07.2023 ADVANCED Module

ANALYSIS REPORT

Thursday, 27 July 2023 - 08:51

Author name : Mike Tibolt
 Company : ArcelorMittal - Stelgence
 Project notes : _____

Disclaimer note
 Given the complexity of the calculation methods, this tool is only intended for professional users active in the sector of steel constructions (who are fully aware of the possibilities, limits and its adequacy thereof for specific practical cases). The User shall use the tool under his own responsibility and at his own risks. The tool does not enable to analyse all situations and to make in an exhaustive way all relevant calculations needed for a study of execution which requires in every case the advice of an external Engineering Office. The tool may be used free of charge. No right is granted to the User of the software, the property and intellectual rights of which continue to belong exclusively to ArcelorMittal Commercial Sections S.A. (or, depending on the case, to the company of the ArcelorMittal Group who is owner of these rights.) No warranty is granted to the User. ArcelorMittal Commercial Sections S.A. and/or any other subsidiaries of the ArcelorMittal Group cannot be held liable for any loss or damage directly and/or indirectly sustained as a result of the use of the software. The User undertakes to hold ArcelorMittal Commercial Sections S.A. free and harmless from any claim and any direct, indirect and/or consequential damages, in particular those resulting from an incorrect or inappropriate use or a use made for an inadequate or inappropriate purpose of the software.

INPUT

Geometry

| | |
|---|------------------------------|
| Case : | Surface crack, weld detail 2 |
| Thickness - t [mm] : | 50 |
| Width - B [mm] : | 300 |
| Weld width - W [mm] : | 5 |
| Weld height - H [mm] : | 5.0 |
| Thickness - T [mm] : | 50 |
| Initial crack size specified by user [True/False] : | 1.0 |
| Initial crack size - a0 [mm] : | 1.95601 |
| a/c [-] : | 0.15 |
| Crack width - c [mm] : | 13.04007 |
| Phi - phi [rad] : | 1.5708 |

Material

| | |
|---------------|--------|
| Steel grade : | S460JR |
|---------------|--------|

Support : stelgence.engineering@arcelormittal.com Page 1 / 4

Example 1: butt weld joint – Clause 2.3



Context

- longitudinal butt-weld detail ($t_f=60\text{mm}$). The steel grade is S355J2.

- The stress level in the frequent design combination (EN 1993-1-10, 2.2(4), with temperature as leading action) is 0,63 fy.

$$\sigma_{\Delta T} = 50 \text{ MPa}$$

$$\sigma_G = 85 \text{ MPa},$$

$$\sigma_Q = 100 \text{ MPa},$$

$$\psi_1 = 0,80$$

- The minimum air temperature is $T_{md} = -25^\circ\text{C}$

$$(T_{Ed} = T_{md} - 5^\circ\text{C} = -30^\circ\text{C})$$

- Cold forming can be neglected.

Example 1: butt weld joint – Clause 2.3

Linear interpolation

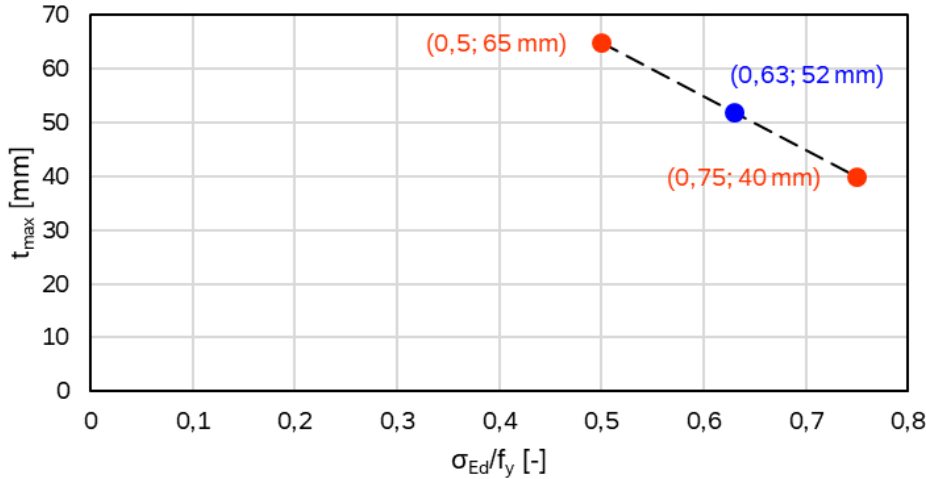



Table 2.1: Maximum permissible values of element thickness t in mm

| Steel grade | Sub-grade | EKV ⁶⁵ at T [°C] | J _{min} | Reference temperature T _{ref} [°C] | | | | | | | | | | | | | | | | | | | | |
|-------------|-----------|--------------------------------|------------------|---|-----|-----|-----|------------------------|-----|-----|-----|------------------------|-----|-----|-----|-----|-----|-----|-----|-----|-----|-----|-----|-----|
| | | | | σ _{Ed} = 0,7 | | | | σ _{Ed} = 0,50 | | | | σ _{Ed} = 0,25 | | | | | | | | | | | | |
| | | | | 10 | 0 | -10 | -20 | -30 | -40 | -50 | 10 | 0 | -10 | -20 | -30 | -40 | -50 | 10 | 0 | -10 | -20 | -30 | -40 | -50 |
| S235 | JR | 20 | 27 | 60 | 50 | 40 | 35 | 30 | 25 | 20 | 90 | 75 | 65 | 55 | 45 | 40 | 35 | 135 | 115 | 100 | 85 | 75 | 65 | 60 |
| | J0 | 0 | 27 | 90 | 75 | 60 | 50 | 40 | 35 | 30 | 125 | 105 | 90 | 75 | 65 | 55 | 45 | 175 | 155 | 135 | 115 | 100 | 85 | 75 |
| | J2 | -20 | 27 | 125 | 105 | 90 | 75 | 60 | 50 | 40 | 170 | 145 | 125 | 105 | 90 | 75 | 65 | 200 | 200 | 175 | 155 | 135 | 115 | 100 |
| S275 | JR | 20 | 27 | 55 | 45 | 35 | 30 | 25 | 20 | 15 | 80 | 70 | 55 | 50 | 40 | 35 | 30 | 125 | 110 | 95 | 80 | 70 | 60 | 55 |
| | J0 | 0 | 27 | 75 | 65 | 55 | 45 | 35 | 30 | 25 | 115 | 95 | 80 | 70 | 55 | 50 | 40 | 165 | 145 | 125 | 110 | 95 | 80 | 70 |
| | J2 | -20 | 27 | 110 | 95 | 75 | 65 | 55 | 45 | 35 | 155 | 130 | 115 | 95 | 80 | 70 | 55 | 200 | 190 | 165 | 145 | 125 | 110 | 95 |
| | M,N | -20 | 40 | 135 | 110 | 95 | 75 | 65 | 55 | 45 | 180 | 155 | 130 | 115 | 95 | 80 | 70 | 200 | 200 | 190 | 165 | 145 | 125 | 110 |
| S355 | ML,NL | -50 | 27 | 185 | 160 | 135 | 110 | 95 | 75 | 65 | 200 | 200 | 180 | 155 | 130 | 115 | 95 | 230 | 200 | 200 | 200 | 190 | 165 | 145 |
| | JR | 20 | 27 | 40 | 35 | 25 | 20 | 15 | 15 | 10 | 65 | 55 | 45 | 40 | 30 | 25 | 25 | 110 | 95 | 80 | 70 | 60 | 55 | 45 |
| | J0 | 0 | 27 | 60 | 50 | 40 | 35 | 25 | 20 | 15 | 95 | 80 | 65 | 55 | 45 | 40 | 30 | 150 | 130 | 110 | 95 | 80 | 70 | 60 |
| S420 | J2 | -20 | 27 | 90 | 75 | 60 | 50 | 40 | 35 | 25 | 135 | 110 | 95 | 80 | 65 | 55 | 45 | 200 | 175 | 150 | 130 | 110 | 95 | 80 |
| | K2,M,N | -20 | 40 | 110 | 90 | 75 | 60 | 50 | 40 | 35 | 155 | 135 | 110 | 95 | 80 | 65 | 55 | 200 | 200 | 175 | 150 | 130 | 110 | 95 |
| | ML,NL | -50 | 27 | 155 | 130 | 110 | 90 | 75 | 60 | 50 | 200 | 180 | 155 | 135 | 110 | 95 | 80 | 210 | 200 | 200 | 200 | 175 | 150 | 130 |
| S460 | M,N | -20 | 40 | 95 | 80 | 65 | 55 | 45 | 35 | 30 | 140 | 120 | 100 | 85 | 70 | 60 | 50 | 200 | 185 | 160 | 140 | 120 | 100 | 85 |
| | ML,NL | -50 | 27 | 135 | 115 | 95 | 80 | 65 | 55 | 45 | 190 | 165 | 140 | 120 | 100 | 85 | 70 | 200 | 200 | 200 | 185 | 160 | 140 | 120 |
| | Q | -20 | 30 | 70 | 60 | 50 | 40 | 30 | 25 | 20 | 110 | 95 | 75 | 65 | 55 | 45 | 35 | 175 | 155 | 130 | 115 | 95 | 80 | 70 |
| S690 | M,N | -20 | 40 | 90 | 70 | 60 | 50 | 40 | 30 | 25 | 130 | 110 | 95 | 75 | 65 | 55 | 45 | 200 | 175 | 155 | 130 | 115 | 95 | 80 |
| | QL | -40 | 30 | 105 | 90 | 70 | 60 | 50 | 40 | 30 | 155 | 130 | 110 | 95 | 75 | 65 | 55 | 200 | 175 | 155 | 130 | 115 | 95 | 80 |
| | ML,NL | -50 | 27 | 125 | 105 | 90 | 70 | 60 | 50 | 40 | 180 | 155 | 130 | 110 | 95 | 75 | 65 | 200 | 200 | 200 | 175 | 155 | 130 | 115 |
| S690 | QL1 | -60 | 30 | 150 | 125 | 105 | 90 | 70 | 60 | 50 | 200 | 180 | 155 | 130 | 110 | 95 | 75 | 215 | 200 | 200 | 200 | 175 | 155 | 130 |
| | Q | 0 | 40 | 40 | 30 | 25 | 20 | 15 | 10 | 10 | 65 | 55 | 45 | 35 | 30 | 20 | 20 | 120 | 100 | 85 | 75 | 60 | 50 | 45 |
| | Q | -20 | 30 | 50 | 40 | 30 | 25 | 20 | 15 | 10 | 80 | 65 | 55 | 45 | 35 | 30 | 20 | 140 | 120 | 100 | 85 | 75 | 60 | 50 |
| | QL | -20 | 40 | 60 | 50 | 40 | 30 | 25 | 20 | 15 | 95 | 80 | 65 | 55 | 45 | 35 | 30 | 165 | 140 | 120 | 100 | 85 | 75 | 60 |
| | QL | -40 | 30 | 75 | 60 | 50 | 40 | 30 | 25 | 20 | 115 | 95 | 80 | 65 | 55 | 45 | 35 | 190 | 165 | 140 | 120 | 100 | 85 | 75 |

$$t = 60\text{mm} \leq t_{max} = 52\text{mm} \rightarrow \text{False}$$

not verified

Example 1: butt weld joint – Clause 2.3



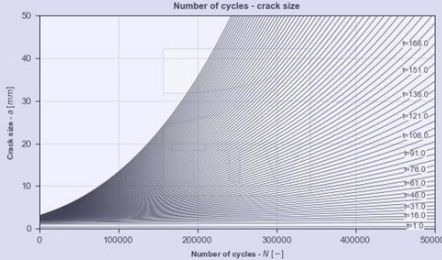
Thickness Toughness Validator

— □ ✕

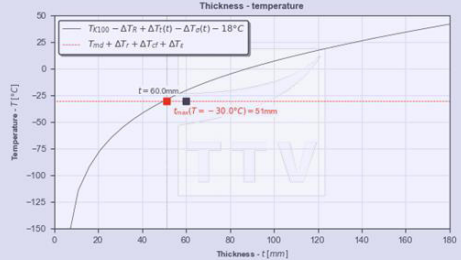
INPUT PARAMETERS

- Geometry**
 - Thickness - t [mm]
- Material**
 - Steel grade
 - Standard EN10025-2 (Non-Alloyed)
- Yield strength**
 - Determination of yield strength
- Impact energy**
 - Determination of impact energy:
 - Impact energy - KV [J]
 - Impact energy reference temperature - T(KV) [°C] T27J [°C]
- Loading conditions**
 - Stress from temperature action - $\sigma(Te)$ [MPa]
 - Stress - $\sigma(G)$ [MPa]
 - Stress - $\sigma(Q)$ [MPa]
 - Combination coefficient - $\psi1$ [-]
 - Stress - σd [MPa]
 - Stress level - $\sigma d / \sigma_y$ [-]
 - Stress from remote constraints - [MPa]
 - Design stress - σd [MPa]
 - Minimum air temperature - T_{mid} [°C]
 - Temperature loss from radiation - ΔTr [°C]
 - Safety allowance - ΔTr [°C]
 - Adjustment for strain rate - ΔTe [°C]
 - Degree of cold forming - DCF [1/100]
 - Adjustment for degree of cold forming - ΔTcf [°C]
- Analysis settings**
 - Total num. of cycles - N [-]
 - Number of steps - n [-]
 - Cycles number per step - N/n [-]
- Stress amplitude**
 - Stress amplitude - $\Delta \sigma$ [MPa]

RESULTS



Number of cycles - crack size



Thickness - temperature

Thickness validation

| | |
|--|-------|
| Maximum thickness - t_{max} [mm] | 51.0 |
| Validations - t_{ok_max} (True/False) | False |

Contact us for further technical support: delignee.engineering@arcormittal.com

CONTROLS

Work directory

Insert metadata

RUN ANALYSIS

TERMINAL

```

>>> 10:02:06 - UPDATE: Procedural crack-growth analysis completed
>>> 10:02:06 - UPDATE: Process completed
>>> 10:02:51 - .....
>>> 10:02:51 - ANALYSIS RUNS
>>> 10:02:51 - .....
>>> 10:02:51 - UPDATE: Input values validated
>>> 10:02:54 - UPDATE: Procedural crack-growth analysis completed
>>> 10:02:54 - UPDATE: Process completed
                    
```

version TTV_1.01
release 09.2024
STANDARD Module
Read carefully the DISCLAIMER note
13.09.2024 10:27:32

Example 1: butt weld joint – Clause 2.3

Linear interpolation

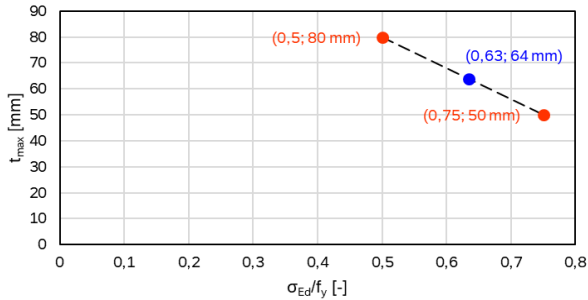
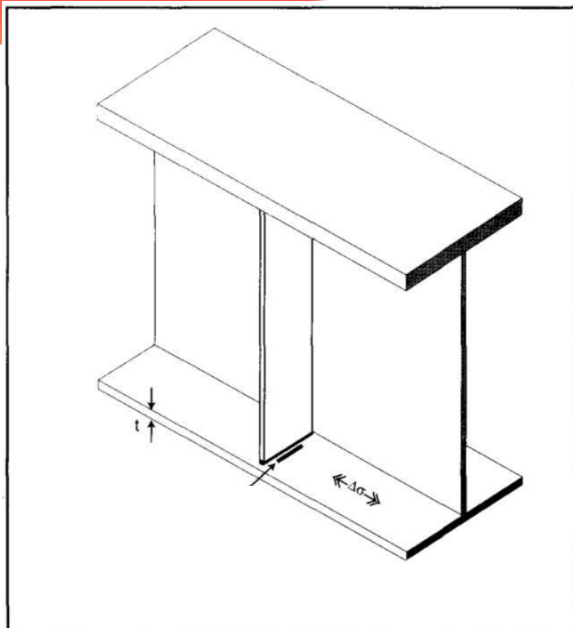


Table 2.1: Maximum permissible values of element thickness t in mm

| Steel grade | Sub-grade | E [KV] | t _{min} [mm] | Reference temper. t _{ref} [°C] | | | | | | | | | | | | | | | | | | | | |
|-------------|-----------|--------|-----------------------|---|-----|-----|-----|-----|-----|----|-----|-----|-----|-----|-----|----|-----|-----|-----|-----|-----|-----|-----|-----|
| | | | | 0 | -10 | -20 | -30 | -40 | -50 | 0 | -10 | -20 | -30 | -40 | -50 | | | | | | | | | |
| S235 | JR | 20 | 27 | 60 | 50 | 40 | 35 | 30 | 25 | 20 | 90 | 75 | 65 | 55 | 45 | 40 | 35 | 135 | 115 | 100 | 85 | 75 | 65 | 60 |
| | JO | 0 | 27 | 90 | 75 | 60 | 50 | 40 | 35 | 30 | 125 | 105 | 90 | 75 | 65 | 55 | 45 | 175 | 155 | 135 | 115 | 100 | 85 | 75 |
| S275 | JR | 20 | 27 | 55 | 45 | 35 | 30 | 25 | 20 | 15 | 80 | 70 | 55 | 50 | 45 | 35 | 30 | 125 | 110 | 95 | 80 | 70 | 60 | 55 |
| | JO | 0 | 27 | 75 | 65 | 55 | 45 | 35 | 30 | 25 | 115 | 95 | 80 | 70 | 55 | 50 | 40 | 165 | 145 | 125 | 110 | 95 | 80 | 70 |
| S355 | JR | 20 | 27 | 40 | 35 | 25 | 20 | 15 | 10 | 5 | 65 | 55 | 45 | 40 | 30 | 25 | 110 | 95 | 80 | 70 | 60 | 55 | 45 | |
| | JO | 0 | 27 | 60 | 50 | 40 | 35 | 25 | 20 | 15 | 95 | 80 | 65 | 55 | 45 | 40 | 30 | 150 | 130 | 110 | 95 | 80 | 70 | 60 |
| S420 | M.N. | -20 | 40 | 95 | 80 | 65 | 55 | 45 | 35 | 30 | 140 | 120 | 100 | 85 | 70 | 60 | 50 | 200 | 185 | 160 | 140 | 120 | 100 | 85 |
| | ML.NL | -50 | 27 | 135 | 115 | 95 | 80 | 65 | 55 | 45 | 190 | 165 | 140 | 120 | 100 | 85 | 70 | 200 | 200 | 200 | 185 | 160 | 140 | 120 |
| S460 | Q | -20 | 30 | 70 | 60 | 50 | 40 | 30 | 25 | 20 | 110 | 95 | 75 | 65 | 55 | 45 | 35 | 175 | 155 | 130 | 115 | 95 | 80 | 70 |
| | M.N. | -20 | 40 | 90 | 70 | 60 | 50 | 40 | 30 | 25 | 130 | 110 | 95 | 75 | 65 | 55 | 45 | 200 | 175 | 155 | 130 | 115 | 95 | 80 |
| S690 | Q | -20 | 30 | 50 | 40 | 30 | 25 | 20 | 15 | 10 | 80 | 65 | 55 | 45 | 35 | 30 | 20 | 140 | 120 | 100 | 85 | 75 | 60 | 50 |
| | Q | -20 | 40 | 60 | 50 | 40 | 30 | 25 | 20 | 15 | 95 | 80 | 65 | 55 | 45 | 35 | 30 | 165 | 140 | 120 | 100 | 85 | 75 | 60 |

Results for **changing to S355K2**: the toughness check for the detail is **satisfied** with both procedures with a difference in t_{max} of 4mm, likely due to linear interpolation and 5mm rounding of the EN 1993-1-10 table.

Example 2: transverse stiffener – Clause 2.4



Context: a stiffener plate is welded on the side of a bridge girder ($t_f=65\text{mm}$, $t_{\text{stiff}}=15\text{mm}$, $b=300\text{mm}$). The steel grade is **S460J0**.

The stress level in the frequent design combination (EN 1993-1-10, 2.2(4), with temperature as leading action) is 0,656 fy

$$\begin{aligned}\sigma_{\Delta T} &= 50 \text{ MPa} \\ \sigma_G &= 185 \text{ MPa}, \\ \sigma_Q &= 70 \text{ MPa}, \\ \psi_1 &= 0,80\end{aligned}$$

The minimum air temperature is $T_{md} = -35^\circ\text{C}$. Cold forming can be neglected.

Every 25 years an inspection is foreseen to detect eventual cracks.

The stress range is $\Delta\sigma = 56\text{MPa}$

Example 2: transverse stiffener – Clause 2.4

Thickness Toughness Validator
TTV Thickness Toughness Validator

INPUT PARAMETERS

- Geometry**
 - Case: Surface crack_weld detail 2
 - Thickness - t [mm]: 65
 - Width - B [mm]: 300
 - Weld width - W [mm]: 10
 - Weld height - H [mm]: 10.0
 - Thickness - T [mm]: 15
 - Initial crack size specified by user [True/False]:
 - Initial crack size - a0 [mm]: 2.087
 - a/c [-]: 0.15
 - Crack width - c [mm]: 13.9146
 - Phi - phi [rad]: 1.5708
- Material**
 - Steel grade: S460J0
 - Standard: EN10025-2 (Non-Alloyed)
- Yield strength**
 - Determination of yield strength: Default from background meth
- Impact energy**
 - Determination of impact energy: Nominal from standards
 - Impact energy - KV [J]: 27.0
 - Impact energy reference temperature - T(KV) [°C]: 0.0
 - T27J [°C]: 0.0
- Loading conditions**
 - Stress from temperature action - $\sigma(T_{Ed})$ [MPa]: 50
 - Stress - $\sigma(G)$ [MPa]: 185
 - Stress - $\sigma(Q)$ [MPa]: 70
 - Combination coefficient - ψ_1 [-]: 0.8
 - Stress - σ_d [MPa]: 2910
 - Stress level - σ_d/f_y [-]: 0.65577
 - Stress from remote constraints - [MPa]: 100
 - Design stress - σ_{Ed} [MPa]: 3910
 - Minimum air temperature - T_{md} [°C]: -35
 - Temperature loss from radiation - ΔT_r [°C]: -5
 - Safety allowance - ΔT_R [°C]: 7.0
 - Adjustment for strain rate - ΔT_e [°C]: 0.0
 - Degree of cold forming - DCF [1/100]: 0
 - Adjustment for degree of cold forming - ΔT_{cf} [°C]: 0.0

RESULTS

Detail sketch

Number of cycles - crack size

Thickness validation

| | |
|---|---------|
| Parameter $\rho - \rho$ [-] | 0.022 |
| Parameter $L_r - L_r$ [-] | 0.892 |
| $\sigma_y - \sigma_y$ [MPa] | 438.126 |
| Stress intensification factor - $K_{app,d}$ [MPa.m ^{0.5}] | 70.426 |
| Corrected stress intens. factor - $K_{app,d,correct}$ [MPa.m ^{0.5}] | 85.471 |
| $\Delta T_o - \Delta T_o$ [°C] | 14.585 |
| Temperature - T [°C] | -40.0 |
| Minimum temperature - T_{min} [°C] | -39.6 |
| Validation: T_a [min] [True/False] | False |

[Contact us for further technical support](#) steilence.engineering@arcelormittal.com

Can change to S460J2 help?

TERMINAL

```

>>> 13:46:15 - ANALYSIS RUNS
>>> 13:46:15 - -----
>>> 13:46:15 - WARNING: ratio H/t is not included in range [0.2; 1]
>>> 13:46:15 - WARNING: ratio W/t is not included in range [0.2; 1]
>>> 13:46:15 - UPDATE: Input values validated
>>> 13:46:15 - WARNING: ratio H/t is not included in range [0.2; 1]
>>> 13:46:15 - WARNING: ratio W/t is not included in range [0.2; 1]
>>> 13:46:15 - UPDATE: Process completed
                    
```

version TTV_101 release 09.2023 ADVANCED Module
--- You can change the interface GRAPHIC STYLE in the splash ---
14.09.2023 13:46:40

Example 2: transverse stiffener – Clause 2.4

TTV Thickness Toughness Validator

CHANGE MODULE

INPUT PARAMETERS

Initial crack size specified by user [True/False] Tick if true

Initial crack size - a0 [mm]

a/c [-]

Crack width - c [mm]

Phi - phi [rad]

Material

Steel grade

Standard

Yield strength

Determination of yield strength

Impact energy

Determination of impact energy

Impact energy - KV [J]

Impact energy reference temperature - TKV [°C]

T27J [°C]

Loading conditions

Stress from temperature action - σ(TEd) [MPa]

Stress - σ(G) [MPa]

Stress - σ(Q) [MPa]

Combination coefficient - ψ1 [-]

Stress - σd [MPa]

Stress level - σd/σy [-]

Stress from remote constraints - [MPa]

Design stress - σEd [MPa]

Minimum air temperature - Tmd [°C]

Temperature loss from radiation - ΔTr [°C]

Safety allowance - ΔTR [°C]

Adjustment for strain rate - ΔTe [°C]

Degree of cold forming - DCF [1/100]

Adjustment for degree of cold forming - ΔTcf [°C]

Analysis settings

Total num. of cycles - N [-]

Number of steps - n [-]

Cycles number per step - N/n [-]

Stress amplitude

Cyclic stress range - Δσ [MPa]

RESULTS

Detail sketch

Thickness validation

| | |
|--|---------|
| Parameter ρ - ρ [-] | 0.022 |
| Parameter Lr - Lr [-] | 0.892 |
| ogy - ogy [MPa] | 438.126 |
| Stress intensification factor - K _{applied} [MPa.m ^{0.5}] | 70.426 |
| Corrected stress intens. factor - K _{applied} correct [MPa.m ^{0.5}] | 85.471 |
| ΔTo - ΔTo [°C] | 14.585 |
| Temperature - T [°C] | -40.0 |
| Minimum temperature - Tmin [°C] | -59.6 |
| Validation - TaTmin [True/False] | True |

Contact us for further technical support: steijence.engineering@arcelormittal.com

Number of cycles - crack size

CONTROLS

Work directory

Insert metadata

RUN ANALYSIS

TERMINAL

```

>>> 16:12:55 - ANALYSIS RUNS
>>> 16:12:55 - .....
>>> 16:12:55 - WARNING: ratio H/t is not included in range [0.2; 1]
>>> 16:12:55 - WARNING: ratio W/t is not included in range [0.2; 1]
>>> 16:12:55 - UPDATE: Input values validated
>>> 16:12:55 - WARNING: ratio H/t is not included in range [0.2; 1]
>>> 16:12:55 - WARNING: ratio W/t is not included in range [0.2; 1]
>>> 16:12:55 - UPDATE: Process completed
            
```

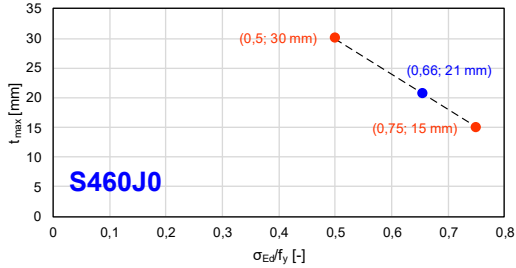
ADVANCED,
S460J2

version TTV_103
release 06.2024
ADVANCED Module
Insert the inputs, run the analysis and VALID
09.05.2024 16:13:48

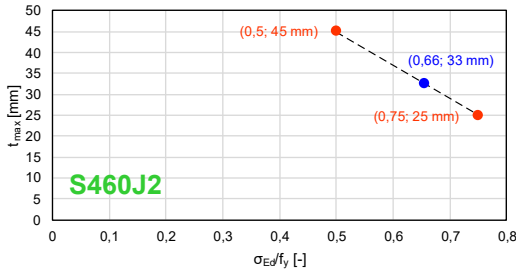
Example 2: transverse stiffener – Clause 2.4

Application of Clause 2.3, EN 1993-1-10 as comparison

Linear interpolation



With TTV
(STANDARD
module) $t_{max} = 21\text{mm}$



With TTV
(STANDARD
module) $t_{max} = 31\text{mm}$

| Steel grade | Quality | KV | Reference Temperature T _r [°C] | | | | | | | | | | | | | | | | | | | | | | | | | | |
|-------------|---------|----|---|-----|-----|-----|-----|-----|-----|-----|------|-----|-----|-----|---------------------|-----|-----|-----|-----|------|-----|-----|-----|-----|-----|-----|-----|-----|------|
| | | | T _r [°C] | | | | | | | | | | | | T _r [°C] | | | | | | | | | | | | | | |
| | | | 10 | 0 | -10 | -20 | -30 | -40 | -50 | -80 | -120 | 10 | 0 | -10 | -20 | -30 | -40 | -50 | -80 | -120 | 10 | 0 | -10 | -20 | -30 | -40 | -50 | -80 | -120 |
| JR | 20 | 27 | 30 | 25 | 20 | 15 | 10 | 10 | 5 | 5 | - | 55 | 45 | 35 | 30 | 25 | 20 | 15 | 10 | 5 | 95 | 80 | 70 | 60 | 50 | 45 | 40 | 25 | 15 |
| J0 | 0 | 27 | 50 | 40 | 30 | 25 | 20 | 15 | 10 | 5 | - | 80 | 65 | 55 | 45 | 35 | 30 | 25 | 15 | 5 | 130 | 115 | 95 | 80 | 70 | 60 | 50 | 35 | 20 |
| J2 | -20 | 27 | 75 | 60 | 50 | 40 | 30 | 25 | 20 | 10 | 5 | 110 | 95 | 80 | 65 | 55 | 45 | 35 | 20 | 10 | 175 | 150 | 130 | 115 | 95 | 80 | 70 | 45 | 25 |
| J4 | -40 | 27 | 105 | 90 | 75 | 60 | 50 | 40 | 30 | 15 | 5 | 155 | 130 | 110 | 95 | 80 | 65 | 55 | 30 | 15 | 200 | 200 | 175 | 150 | 130 | 115 | 95 | 60 | 35 |
| K2,M,N | -20 | 40 | 90 | 75 | 60 | 50 | 40 | 30 | 25 | 10 | 5 | 130 | 110 | 95 | 80 | 65 | 55 | 45 | 25 | 10 | 200 | 175 | 150 | 130 | 115 | 95 | 80 | 50 | 30 |
| J5,M,N,L | -50 | 27 | 125 | 105 | 90 | 75 | 60 | 50 | 40 | 20 | 5 | 180 | 155 | 130 | 110 | 95 | 80 | 65 | 35 | 15 | 200 | 200 | 200 | 175 | 150 | 130 | 115 | 70 | 40 |
| Q | -20 | 30 | 75 | 60 | 50 | 40 | 30 | 25 | 20 | 10 | 5 | 110 | 95 | 80 | 65 | 55 | 45 | 35 | 20 | 10 | 175 | 150 | 130 | 115 | 95 | 80 | 70 | 45 | 25 |
| QL | -40 | 30 | 105 | 90 | 75 | 60 | 50 | 40 | 30 | 15 | 5 | 155 | 130 | 110 | 95 | 80 | 65 | 55 | 30 | 15 | 200 | 200 | 175 | 150 | 130 | 115 | 95 | 60 | 35 |
| QL1 | -60 | 30 | 150 | 125 | 105 | 90 | 75 | 60 | 50 | 25 | 10 | 200 | 180 | 155 | 130 | 110 | 95 | 80 | 45 | 20 | 200 | 200 | 200 | 200 | 175 | 150 | 130 | 80 | 45 |
| J0 | 0 | 27 | 45 | 35 | 30 | 20 | 20 | 15 | 10 | 5 | - | 70 | 60 | 50 | 40 | 35 | 25 | 20 | 10 | 5 | 125 | 105 | 90 | 80 | 65 | 55 | 50 | 30 | 20 |
| K2,M,N | -20 | 40 | 80 | 65 | 55 | 45 | 35 | 30 | 20 | 10 | 5 | 125 | 105 | 85 | 70 | 60 | 50 | 40 | 20 | 10 | 195 | 170 | 145 | 125 | 105 | 90 | 80 | 50 | 25 |
| M,N,L | -50 | 27 | 120 | 100 | 80 | 65 | 55 | 45 | 35 | 20 | 10 | 170 | 145 | 125 | 105 | 85 | 70 | 60 | 35 | 15 | 250 | 220 | 195 | 170 | 145 | 125 | 105 | 65 | 35 |
| Q | 0 | 40 | 55 | 45 | 35 | 30 | 20 | 20 | 15 | 5 | - | 85 | 70 | 60 | 50 | 40 | 35 | 25 | 15 | 5 | 145 | 125 | 105 | 90 | 80 | 65 | 55 | 35 | 20 |
| Q | -20 | 30 | 65 | 55 | 45 | 35 | 30 | 20 | 20 | 10 | 5 | 105 | 85 | 70 | 60 | 50 | 40 | 35 | 20 | 10 | 170 | 145 | 125 | 105 | 90 | 80 | 65 | 40 | 25 |
| QL | -20 | 40 | 80 | 65 | 55 | 45 | 35 | 30 | 20 | 10 | 5 | 125 | 105 | 85 | 70 | 60 | 50 | 40 | 20 | 10 | 195 | 170 | 145 | 125 | 105 | 90 | 80 | 50 | 25 |
| QL | -40 | 30 | 100 | 80 | 65 | 55 | 45 | 35 | 30 | 15 | 5 | 145 | 125 | 105 | 85 | 70 | 60 | 50 | 25 | 10 | 220 | 195 | 170 | 145 | 125 | 105 | 90 | 55 | 30 |
| QL1 | -40 | 40 | 120 | 100 | 80 | 65 | 55 | 45 | 35 | 20 | 5 | 170 | 145 | 125 | 105 | 85 | 70 | 60 | 35 | 15 | 250 | 220 | 195 | 170 | 145 | 125 | 105 | 65 | 35 |
| QL1 | -60 | 30 | 140 | 120 | 100 | 80 | 65 | 55 | 45 | 25 | 10 | 195 | 170 | 145 | 125 | 105 | 85 | 70 | 40 | 20 | 250 | 250 | 220 | 195 | 170 | 145 | 125 | 80 | 40 |

- S460J0 $t = 65\text{mm} \leq t_{max} = 21\text{mm} \rightarrow \text{False}$ not verified
- S460J2 $t = 65\text{mm} \leq t_{max} = 33\text{mm} \rightarrow \text{False}$ not verified

Thanks for your attention



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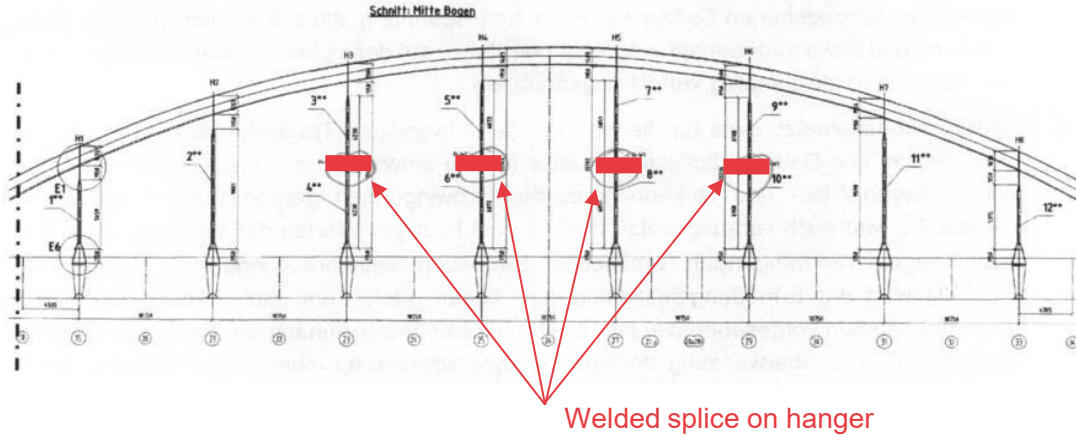


APPENDICE SLIDES

Example 3: butt weld on hanger – Clause 2.4



Source: Section 2.4.2 - COMMENTARY AND WORKED EXAMPLES to EN 1993-1-10 “Material toughness and through thickness properties” and other toughness oriented rules in EN 1993 - G. Sedlacek, M. Feldmann, B. Kühn, D. Tschickardt, S. Höhler, C. Müller, W. Hensen, N. Stranghöner, W. Dahl, P. Langenberg, S. Münstermann, J. Brozetti, J. Raoul, R. Pope, F. Bijlaard



Context: in an arch bridge the central hangers have a welded splice in the middle. The hangers are $\Phi 220\text{mm}$ rods in **S420NL**. The section in front of the splice constitutes one of the critical sections that needs to be checked.

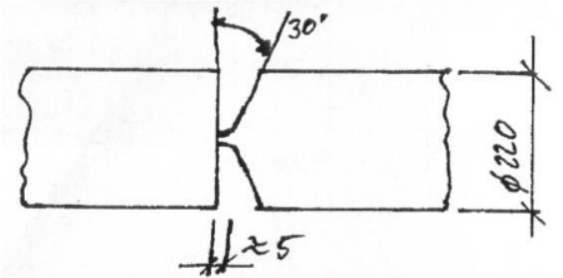
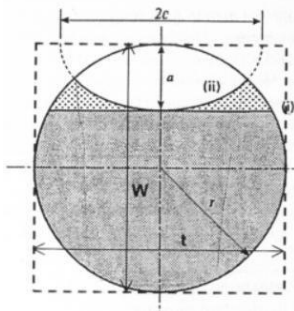


Fig. 2-73: Detail of welded splice



Example 3: butt weld on hanger – Clause 2.4



| a) surface crack at hanger splice | |
|-------------------------------------|-----------------------------|
| Plate thickness t_{max} | 220 mm |
| yield strength f_y | 420 N/mm ² |
| frequent stress σ_p | 0,55 f_y (t) |
| residual stress σ_s | 100 MPa |
| initial crack depth a_0 | 6,00 mm |
| inhomogeneity $\Delta 1_{27J}$ | 0 K |
| $\Delta\sigma_p$ | ≤ 15 N/mm ² |
| $\Delta K < \Delta K_{th}$ | 2,8 MPa $\sqrt{m} < 5$ |
| Crack growth Δa | 0 mm |
| Design crack depth a_d | 6 mm |
| a/w | 0,027 |
| Y for crack (Murakami) | 1,1211 |
| M_y for plate | 1,0 |
| f_y (t) | 320 N/mm ² |
| σ_p | 176 N/mm ² |
| $\sigma_{Ed} = \sigma_p + \sigma_s$ | 276 N/mm ² |
| $K_{appl,d} (a_i, \sigma_{Ed})$ | 42 MPa \sqrt{m} |
| W | 220 mm |
| σ_{gy} | 311,3 N/mm ² |
| $L_y = \sigma_{Ed}/\sigma_{gy}$ | 0,89 |
| K_{RB} | 0,85 |
| Residual stresses ψ | 0,32 |
| ρ_1 | 0,0437 |
| ρ_2 | 0,0286 |
| $K_{appl,d}$ | 51,89 MPa \sqrt{m} |
| t_{eff} | 220 mm |
| ΔT_g | + 23,1 K |

The section is modelled via an equivalent square section 220x220 mm and a single edge crack is considered with depth 6mm.

The $f_y(t)$ value is according to EN 10025-3

The stress level in the frequent design combination (EN 1993-1-10, 2.2(4), with temperature as leading action) is 0,55 f_y

$$\sigma_{\Delta T} = 22^\circ C \times 1,2 \cdot 10^{-5} C^{-1} \times 210000 MPa = 55,5 MPa \quad \text{Estimated}$$

$$\sigma_G = 75 MPa,$$

$$\sigma_Q = 57 MPa,$$

$$\psi_1 = 0,80$$

The minimum air temperature is $T_{md} = -25^\circ C$

No cold forming is present.

Example 3: butt weld on hanger – Clause 2.4

Thickness Toughness Validator

— □ ✕

INPUT PARAMETERS

Geometry

Case: Single edge crack

Initial crack size - a0 [mm]: 6

Thickness - t [mm]: 220

Plate width - W [mm]: 220

Welds are present [True/False]: Tick if true

Material

Steel grade: S420NL

Standard: V10025-3 (Normalized/Normalized-Role)

Yield strength

Determination of yield strength: Nominal from standards

Impact energy

Determination of impact energy: Nominal from standards

Impact energy - KV [J]: 27.0

Impact energy / reference temperature - T(KV) [°C]: -50.0

T27J [°C]: -50.0

Loading conditions

Stress from temperature action - $\sigma(T_{E0})$ [MPa]: 55.5

Stress - $\sigma(S)$ [MPa]: 75

Stress - $\sigma(Q)$ [MPa]: 51

Combination coefficient - ψ_1 [-]: 0.8

Stress - σ_d [MPa]: 176.1

Stress level - σ_d/f_y [-]: 0.55031

Stress from remote constraints - [MPa]: 100

Design stress - σ_{Ed} [MPa]: 276.1

Minimum air temperature - T_{min} [°C]: -25

Temperature loss from radiation - ΔT_r [°C]: -5

Safety allowance - ΔTR [°C]: 7.0

Adjustment for strain rate - ΔT_e [°C]: 0.0

Degree of cold forming - DCF [1/100]: 0

Adjustment for degree of cold forming - ΔT_d [°C]: 0.0

Analysis settings

Total num. of cycles - N [-]: 0

Number of steps - n [-]: 1000

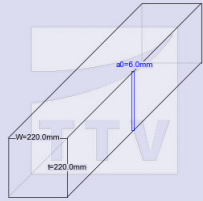
Cycles number per step - N/n [-]: 0.0

Stress amplitude

Stress amplitude - $\Delta \sigma$ [MPa]: 56

RESULTS

Detail sketch

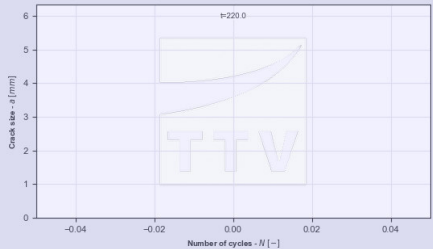


Thickness validation

| | |
|--|---------|
| Parameter p - p [-] | 0.029 |
| Parameter Lr - Lr [-] | 0.887 |
| $\sigma_{py} - \sigma_{py}$ [MPa] | 311.273 |
| Stress intensification factor - K _{app,d} [MPa ^m *0.5] | 42.498 |
| Corrected stress intens. factor - K _{app,d,correct} [MPa ^m *0.5] | 51.913 |
| $\Delta T_e - \Delta T_e$ [°C] | 23.016 |
| Temperature - T [°C] | -30.0 |
| Minimum temperature - T _{min} [°C] | -98.0 |
| Validation - T _a T _{min} [True/False] | True |

Contact us for further technical support: steigence.engineering@arcelormittal.com

Number of cycles - crack size



TERMINAL

```

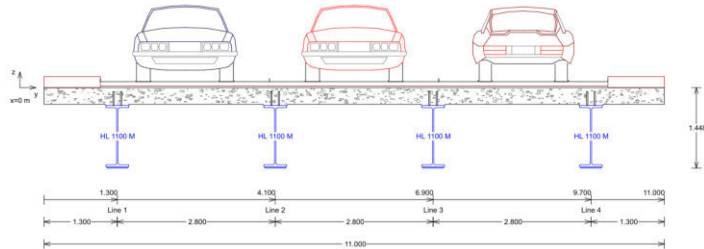
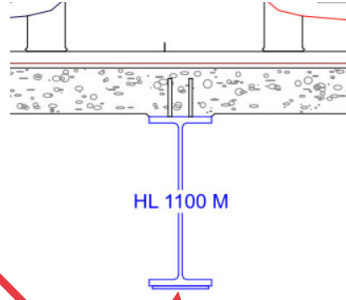
>>> 17:12:31 - .....
>>> 17:12:31 - UPDATE: Input values validated
>>> 17:12:32 - UPDATE: Process completed
>>> 17:29:44 - .....
>>> 17:29:44 - ANALYSIS RUNS
>>> 17:29:44 - .....
>>> 17:29:44 - UPDATE: Input values validated
>>> 17:29:44 - UPDATE: Process completed
                    
```

version TTV_101 release 09.2023 ADVANCED Module

--- Read carefully the DISCLAIMER note ---

12.09.2023 17:31:19

Example 4: plate on bottom flange – Clause 2.4



Strengthening plate at bottom flange

Context: a strengthening plate is welded on the bottom flange of a box girder out of HL 1100 M profiles in S460M. ($t_f=40\text{mm}$, $t_p=20\text{mm}$, $b_{tot}=400\text{mm}$)

The $f_y(t)$ value is according to EN 10025-4

The stress level in the frequent design combination (with temperature as leading action) is 0,284 f_y

$$\begin{aligned}\sigma_{\Delta T} &= 45 \text{ MPa} \\ \sigma_G &= 40 \text{ MPa}, \\ \sigma_Q &= 50 \text{ MPa}, \\ \psi_1 &= 0,80\end{aligned}$$

The minimum air temperature is $T_{md} = -35\text{ }^\circ\text{C}$

Every 25 years an inspection is foreseen to detect eventual cracks.

The stress range is $\Delta\sigma = 50\text{ MPa}$.

Example 4: plate on bottom flange – Clause 2.4

Thickness Toughness Validator

✖

INPUT PARAMETERS

| | |
|--|---------------------------------------|
| Plate width - b [mm] | 400 |
| Initial crack size specified by user [True/False] | <input type="checkbox"/> Tick if true |
| Initial crack size - a0 [mm] | 1.844 |
| K/C [-] | 0.15 |
| Crack width - c [mm] | 12.29627 |
| Phi - phi [rad] | 1.5708 |
| Material | |
| Steel grade | S460M |
| Standard | EN10225-4 (Thermomechanical) |
| Yield strength | |
| Determination of yield strength | Nominal from standards |
| Impact energy | |
| Determination of impact energy | Nominal from standards |
| Impact energy - KV [J] | 40.0 |
| Impact energy reference temperature - T(KV) [°C] | -20.0 |
| T27 [°C] | -29.0 |
| Loading conditions | |
| Stress from temperature action - $\sigma(Ted)$ [MPa] | 45 |
| Stress - $\sigma(G)$ [MPa] | 40 |
| Stress - $\sigma(Q)$ [MPa] | 50 |
| Combination coefficient - $\Psi1$ [-] | 0.8 |
| Stress - σd [MPa] | 125.0 |
| Stress level - $\sigma d/fy$ [-] | 0.28409 |
| Stress from remote constraints - [MPa] | 100 |
| Design stress - σd [MPa] | 225.0 |
| Minimum air temperature - Tmd [°C] | -35 |
| Temperature loss from radiation - ΔTr [°C] | -5 |
| Safety allowance - ΔTr [°C] | 7.0 |
| Adjustment for strain rate - ΔTe [°C] | 0.0 |
| Degree of cold forming - DCF [1/100] | 0 |
| Adjustment for degree of cold forming - ΔTf [°C] | 0.0 |
| Analysis settings | |
| Total num. of cycles - N [-] | 500000 |
| Number of steps - n [-] | 1000 |
| Cycles number per step - N/n [-] | 500.0 |
| Stress amplitude | |
| Stress amplitude - $\Delta \sigma$ [MPa] | 50 |

RESULTS

Detail sketch

Thickness validation

| | |
|--|---------|
| Parameter $p - p$ [-] | 0.035 |
| Parameter $Lr - Lr$ [-] | 0.516 |
| ogy - ogy [MPa] | 436.343 |
| Stress intensification factor - K_{appld} [MPa $\cdot m^{0.5}$] | 19.707 |
| Corrected stress intens. factor - $K_{appld,correct}$ [MPa $\cdot m^{0.5}$] | 21.777 |
| $\Delta To - \Delta To$ [°C] | 120.0 |
| Temperature - T [°C] | -40.0 |
| Minimum temperature - Tmin [°C] | -174.4 |
| Validation: $TsTmin$ [True/False] | True |

Contact us for further technical support: stelgence.engineering@arcelormittal.com

Number of cycles - crack size

CONTROLS

Work directory

Insert metadata

RUN ANALYSIS

TERMINAL

```

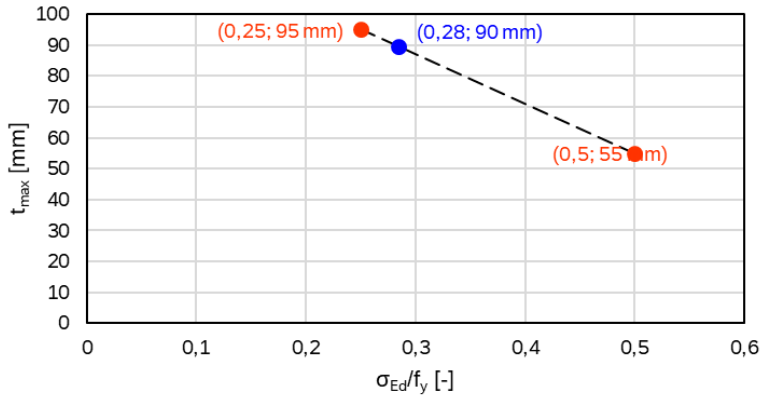
>>> 16:40:26 - WARNING: ratio T/t is not included in range [0.64; 2.0]
>>> 16:40:27 - -----
>>> 16:40:27 - ANALYSIS RUNS
>>> 16:40:27 - -----
>>> 16:40:27 - WARNING: ratio T/t is not included in range [0.64; 2.0]
>>> 16:40:27 - UPDATE: Input values validated
>>> 16:40:27 - WARNING: ratio T/t is not included in range [0.64; 2.0]
>>> 16:40:27 - UPDATE: Process completed
                    
```

version TTV_101 release 09.2023 ADVANCED Module
--- Analyse the results by opening the CSV OUTPUT FILES in the workdirectory ---
13.09.2023 16:41:11

Example 4: plate on bottom flange – Clause 2.4

Application of Clause 2.3, EN 1993-1-10 as comparison

Linear interpolation



With **TTV (STANDARD module)** → $t_{max} = 87\text{mm}$

Table 2.1: Maximum permissible values of element thickness t in mm

| Steel grade | Sub-grade | ES KV at T [°C] | J _{min} | Reference temperature T_{Ed} [°C] | | | | | | | | | | | | | | | | | | | | | | | | | | | | |
|-------------|-----------|-----------------|------------------|-------------------------------------|-----|-----|-----|-----|-----|-----|-----|-----|-----|-----|-----|-----------------------------|-----|-----|-----|-----|-----|-----------------------------|-----|-----|-----|-----|-----|-----|-----|-----|-----|-----|
| | | | | 10 | 0 | -10 | -20 | -30 | -40 | -50 | 10 | 0 | -10 | -20 | -30 | -40 | -50 | 10 | 0 | -10 | -20 | -30 | -40 | -50 | | | | | | | | |
| | | | | $\sigma_{Ed} = 0.75 f_y(t)$ | | | | | | | | | | | | $\sigma_{Ed} = 0.50 f_y(t)$ | | | | | | $\sigma_{Ed} = 0.25 f_y(t)$ | | | | | | | | | | |
| S235 | JR | 20 | 27 | 60 | 50 | 40 | 35 | 30 | 25 | 20 | 90 | 75 | 65 | 55 | 45 | 40 | 35 | 30 | 25 | 20 | 15 | 100 | 85 | 75 | 65 | 60 | 135 | 115 | 100 | 85 | 75 | |
| | J0 | 0 | 27 | 90 | 75 | 60 | 50 | 40 | 35 | 30 | 125 | 105 | 90 | 75 | 65 | 55 | 45 | 35 | 25 | 15 | 10 | 175 | 155 | 135 | 115 | 100 | 200 | 175 | 155 | 135 | 115 | |
| | J2 | -20 | 27 | 125 | 105 | 90 | 75 | 60 | 50 | 40 | 170 | 145 | 125 | 105 | 90 | 75 | 65 | 55 | 45 | 35 | 30 | 200 | 175 | 155 | 135 | 115 | 220 | 200 | 180 | 160 | 145 | |
| S275 | JR | 20 | 27 | 55 | 45 | 35 | 30 | 25 | 20 | 15 | 80 | 70 | 55 | 45 | 40 | 35 | 30 | 25 | 20 | 15 | 10 | 125 | 110 | 95 | 80 | 70 | 160 | 145 | 125 | 110 | 95 | |
| | J0 | 0 | 27 | 75 | 65 | 55 | 45 | 35 | 30 | 25 | 115 | 95 | 80 | 70 | 55 | 50 | 40 | 30 | 20 | 15 | 10 | 165 | 145 | 125 | 110 | 95 | 200 | 180 | 160 | 145 | 125 | |
| | J2 | -20 | 27 | 110 | 95 | 75 | 65 | 55 | 45 | 35 | 155 | 130 | 115 | 95 | 80 | 70 | 55 | 45 | 35 | 25 | 20 | 200 | 180 | 165 | 145 | 125 | 220 | 200 | 180 | 165 | 145 | |
| | M,N | -20 | 40 | 135 | 110 | 95 | 75 | 65 | 55 | 45 | 180 | 155 | 130 | 115 | 95 | 80 | 70 | 200 | 200 | 190 | 165 | 145 | 225 | 200 | 180 | 165 | 145 | 250 | 225 | 200 | 180 | 165 |
| S355 | JR | 20 | 27 | 40 | 35 | 25 | 20 | 15 | 10 | 5 | 65 | 55 | 45 | 40 | 30 | 25 | 20 | 15 | 10 | 5 | 5 | 110 | 95 | 80 | 70 | 60 | 145 | 125 | 110 | 95 | 80 | |
| | J0 | 0 | 27 | 60 | 50 | 40 | 35 | 25 | 20 | 15 | 95 | 80 | 65 | 55 | 45 | 40 | 30 | 20 | 15 | 10 | 5 | 150 | 130 | 110 | 95 | 80 | 200 | 175 | 150 | 130 | 110 | |
| | J2 | -20 | 27 | 90 | 75 | 60 | 50 | 40 | 35 | 25 | 135 | 110 | 95 | 80 | 65 | 55 | 45 | 35 | 25 | 15 | 10 | 200 | 175 | 150 | 130 | 110 | 220 | 200 | 180 | 160 | 140 | |
| | M,N | -20 | 40 | 110 | 90 | 75 | 60 | 50 | 40 | 35 | 155 | 135 | 110 | 95 | 80 | 65 | 55 | 200 | 200 | 190 | 175 | 150 | 200 | 180 | 165 | 145 | 125 | 225 | 200 | 180 | 165 | 145 |
| S420 | M,N | -20 | 40 | 95 | 80 | 65 | 55 | 45 | 35 | 30 | 140 | 120 | 100 | 85 | 70 | 60 | 50 | 200 | 200 | 185 | 160 | 140 | 200 | 180 | 165 | 140 | 120 | 220 | 200 | 180 | 165 | 140 |
| | ML,NL | -50 | 27 | 135 | 115 | 95 | 80 | 65 | 55 | 45 | 190 | 165 | 140 | 120 | 100 | 85 | 70 | 200 | 200 | 200 | 185 | 160 | 220 | 200 | 180 | 165 | 140 | 240 | 220 | 200 | 185 | 160 |
| | Q | -20 | 30 | 70 | 60 | 50 | 40 | 30 | 25 | 20 | 110 | 95 | 75 | 65 | 55 | 45 | 35 | 175 | 155 | 130 | 115 | 95 | 200 | 180 | 165 | 140 | 120 | 220 | 200 | 180 | 165 | 140 |
| | M,N | -20 | 40 | 90 | 70 | 60 | 50 | 40 | 30 | 25 | 140 | 110 | 95 | 75 | 65 | 55 | 45 | 200 | 175 | 155 | 140 | 115 | 200 | 180 | 165 | 140 | 120 | 220 | 200 | 180 | 165 | 140 |
| S460 | Q | -20 | 30 | 70 | 60 | 50 | 40 | 30 | 25 | 20 | 110 | 95 | 75 | 65 | 55 | 45 | 35 | 175 | 155 | 130 | 115 | 95 | 200 | 180 | 165 | 140 | 120 | 220 | 200 | 180 | 165 | 140 |
| | M,N | -20 | 40 | 90 | 70 | 60 | 50 | 40 | 30 | 25 | 140 | 110 | 95 | 75 | 65 | 55 | 45 | 200 | 175 | 155 | 140 | 115 | 200 | 180 | 165 | 140 | 120 | 220 | 200 | 180 | 165 | 140 |
| | QL | -40 | 30 | 105 | 90 | 70 | 60 | 50 | 40 | 30 | 155 | 130 | 110 | 95 | 75 | 65 | 55 | 200 | 200 | 175 | 155 | 130 | 200 | 180 | 165 | 140 | 120 | 220 | 200 | 180 | 165 | 140 |
| | ML,NL | -50 | 27 | 125 | 105 | 90 | 70 | 60 | 50 | 40 | 180 | 155 | 130 | 110 | 95 | 75 | 65 | 200 | 200 | 200 | 175 | 155 | 200 | 180 | 165 | 140 | 120 | 220 | 200 | 180 | 165 | 140 |
| S690 | QL1 | -60 | 30 | 150 | 125 | 105 | 90 | 70 | 60 | 50 | 200 | 180 | 155 | 130 | 110 | 95 | 75 | 215 | 200 | 200 | 175 | 155 | 200 | 180 | 165 | 140 | 120 | 220 | 200 | 180 | 165 | 140 |
| | Q | 0 | 40 | 40 | 30 | 25 | 20 | 15 | 10 | 10 | 65 | 55 | 45 | 35 | 30 | 20 | 20 | 120 | 100 | 85 | 75 | 60 | 50 | 45 | 40 | 30 | 25 | 60 | 50 | 40 | 30 | 25 |
| | Q | -20 | 30 | 50 | 40 | 30 | 25 | 20 | 15 | 10 | 80 | 65 | 55 | 45 | 35 | 30 | 20 | 140 | 120 | 100 | 85 | 75 | 60 | 50 | 40 | 30 | 25 | 70 | 60 | 50 | 40 | 30 |
| | QL | -20 | 40 | 60 | 50 | 40 | 30 | 25 | 20 | 15 | 95 | 80 | 65 | 55 | 45 | 35 | 30 | 165 | 140 | 120 | 100 | 85 | 75 | 60 | 50 | 40 | 30 | 85 | 75 | 60 | 50 | 40 |
| S690 | QL | -40 | 30 | 75 | 60 | 50 | 40 | 30 | 25 | 20 | 115 | 95 | 80 | 65 | 55 | 45 | 35 | 190 | 165 | 140 | 120 | 100 | 85 | 75 | 60 | 50 | 40 | 100 | 85 | 75 | 60 | 50 |
| | QL1 | -40 | 40 | 90 | 75 | 60 | 50 | 40 | 30 | 25 | 135 | 115 | 95 | 80 | 65 | 55 | 45 | 200 | 190 | 165 | 140 | 120 | 100 | 85 | 75 | 60 | 50 | 120 | 100 | 85 | 75 | 60 |
| | QL1 | -40 | 40 | 90 | 75 | 60 | 50 | 40 | 30 | 25 | 135 | 115 | 95 | 80 | 65 | 55 | 45 | 200 | 190 | 165 | 140 | 120 | 100 | 85 | 75 | 60 | 50 | 120 | 100 | 85 | 75 | 60 |
| | QL1 | -60 | 30 | 110 | 90 | 75 | 60 | 50 | 40 | 30 | 160 | 135 | 115 | 95 | 80 | 65 | 55 | 200 | 200 | 190 | 165 | 140 | 120 | 100 | 85 | 75 | 60 | 140 | 120 | 100 | 85 | 70 |

$$t = 40\text{mm} \leq t_{max} = 90\text{mm} \rightarrow \text{True}$$

verified

**2. FATIGUE LIFE IMPROVEMENT OF WELDED BRIDGE
DETAILS USING HIGH FREQUENCY MECHANICAL IMPACT
TREATMENT (MOHAMMAD AL-EMRANI, CHALMERS
UNIVERSITY OF TECHNOLOGY)**

Fatigue life improvement of welded bridge details

using High Frequency Mechanical Impact treatment



Mohammad Al-Emrani, Chalmers University of Technology

mohammad.al-emrani@chalmers.se

2025-11-04

Agenda

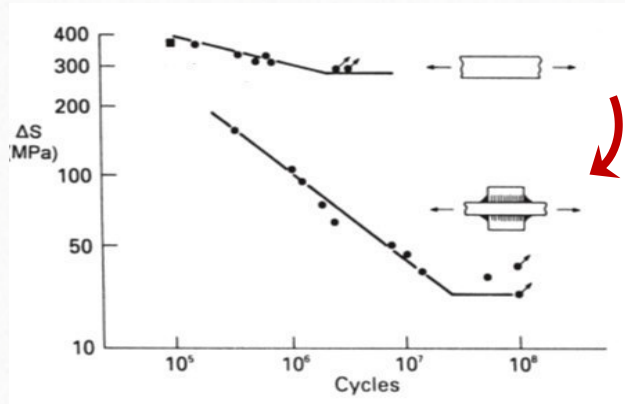
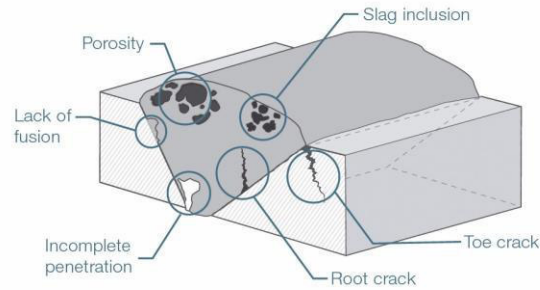


- What is HFMI and how it works
- Implications on design procedure and fatigue verification
- Design rules in prEN 1993-1-9 (Annex F)
- Potential weight and cost saving employing HFMI – a case study
- Design example

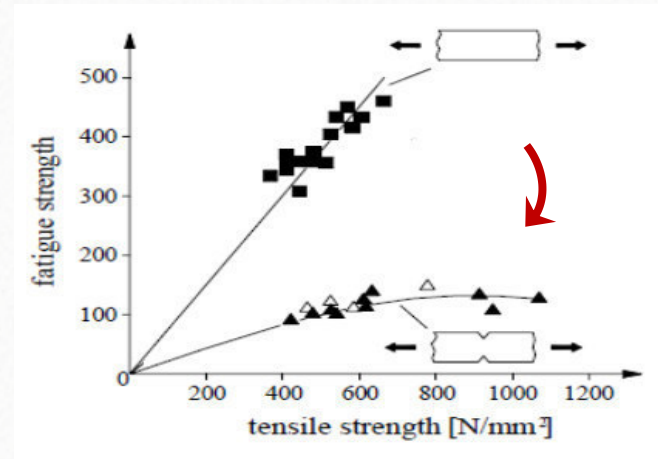
Welded details

Two serious implications of welding

- Weld defects (e.g. undercut)
- Residual stresses



- "No" crack initiation phase
- Substantially reduced fat. strength



- Fatigue strength (practically) independent on:
 - Steel strength
 - Mean stress

Post-weld treatment

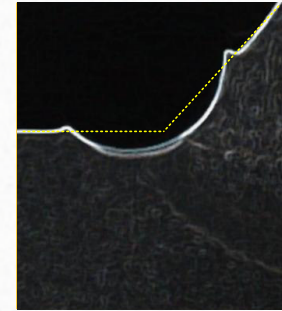
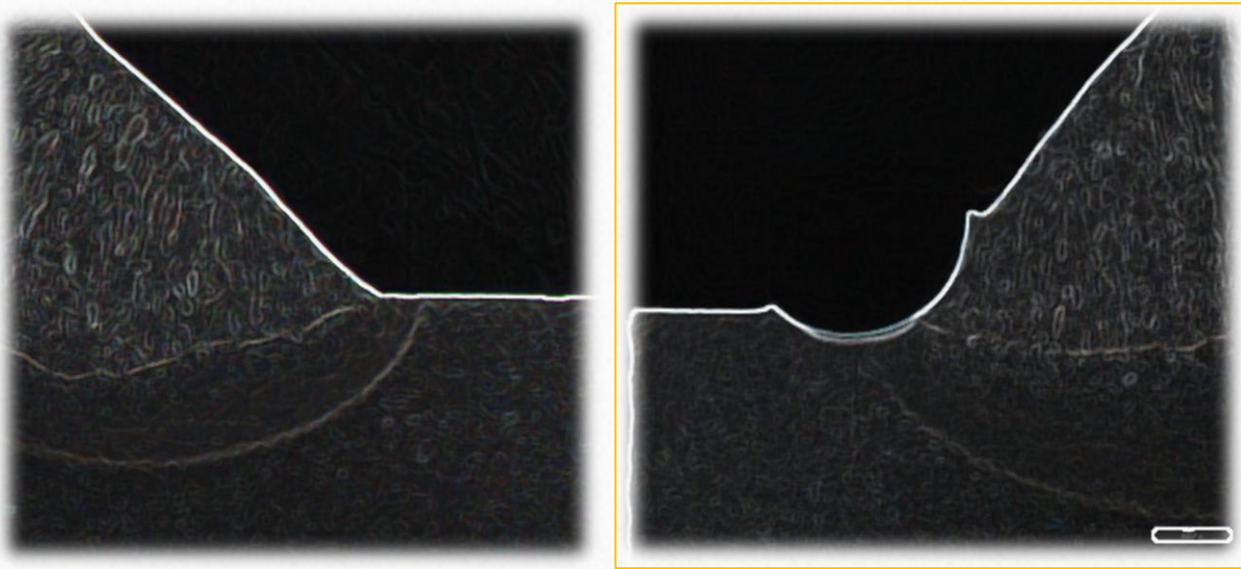
High Frequency Mechanical Impact treatment (HFMI)



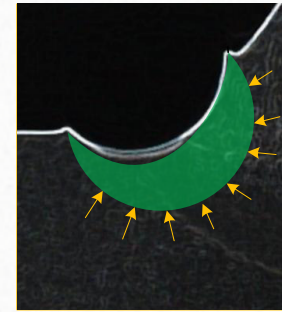
Post-weld treatment

High Frequency Mechanical Impact treatment (HFMI)

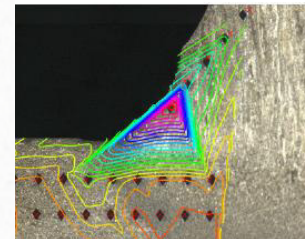
Cold-working of the material at weld toe gives ..



Reduced stress concentration



Local compressive residual stress

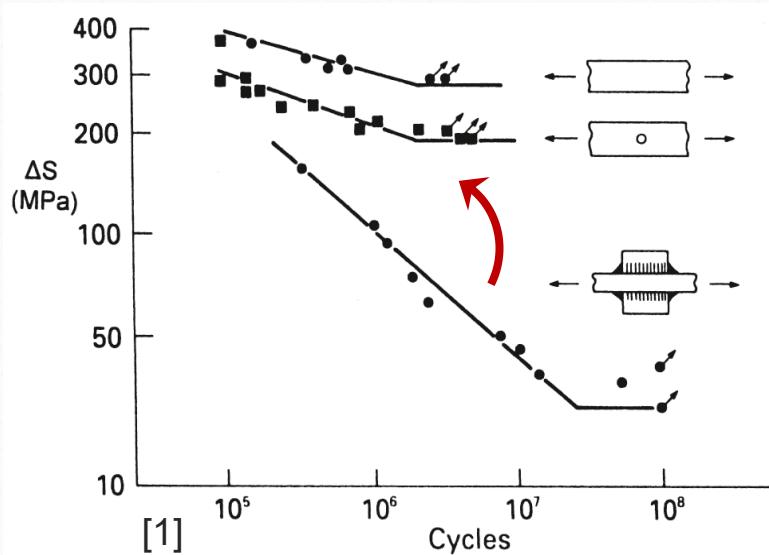


Increased hardness

HFMI treatment – Increase fatigue strength



0



“Restore” the fatigue strength of the steel metal
Considerable part of the fatigue life is **crack initiation**

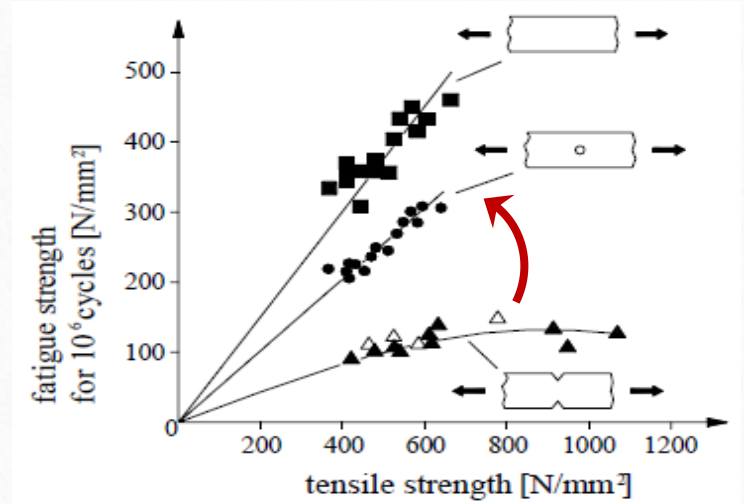
→ higher fatigue strength
& shallower slope of the S-N curve

HFMI treatment – Increase fatigue strength

Particularities: f_y dependency 1

→ Restore the fatigue strength dependency on material strength.

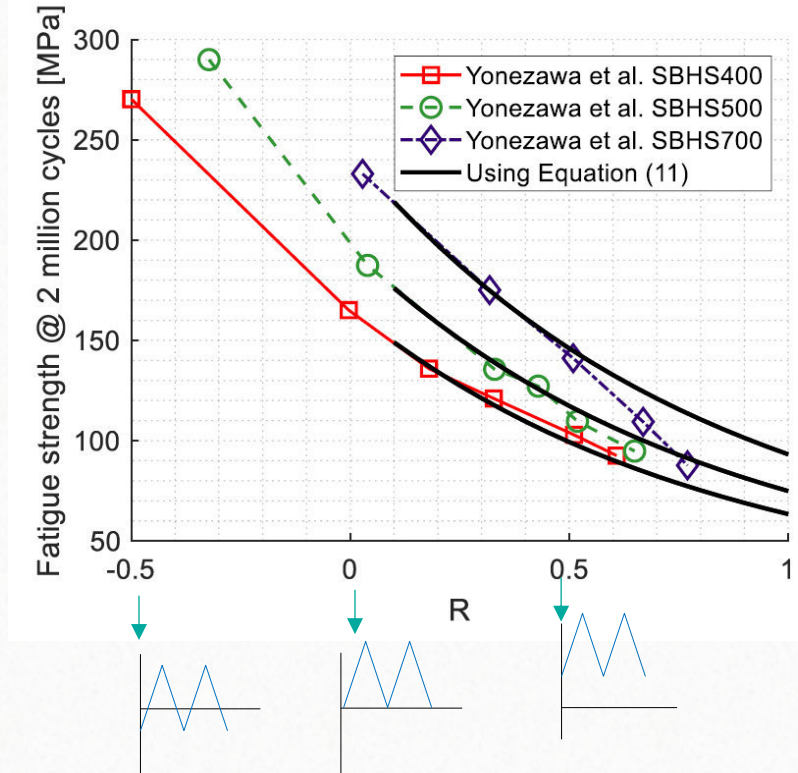
(HSS will have higher fatigue strength)



HFMI treatment – Increase fatigue strength, **but**

Particularities: *Fatigue strength dependency on the **R-ratio** (mean stress)* **2**

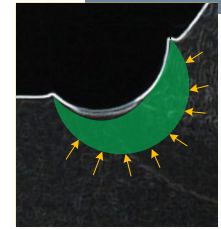
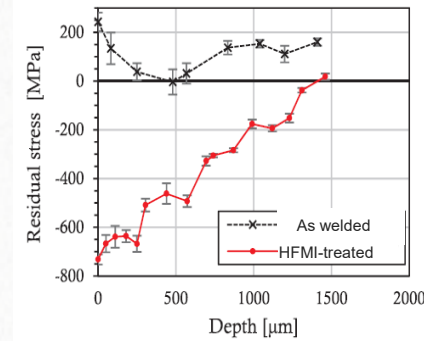
Load cycles with high **R-ratio**
(mean stress) are more damaging!



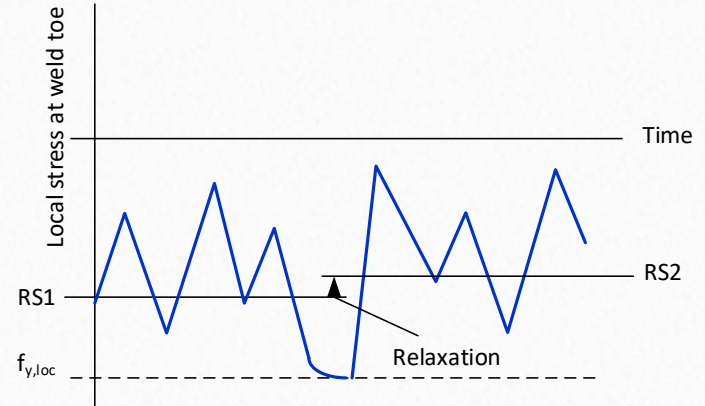
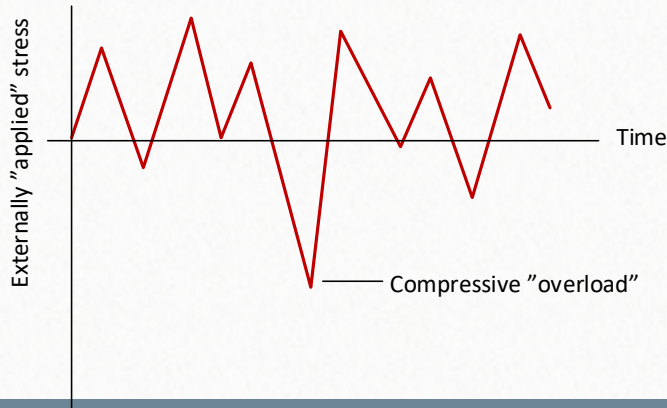
HFMI treatment - Increase fatigue strength, **but**

Particularities: *Limitations on **Max./Min.** allowable stress* **3**

An "overload" that is high enough to cause relaxation of these residual stresses would reduce the efficiency of HFMI



Compressive overloads are particularly harmful



Design of HFMI-treated welds

Rules in Annex F of prEN 1993-1-9

0 1 $\Delta\sigma_{C, HFMI} = f(f_y)$

2 $\Delta\sigma_{Ed} = f(R)$

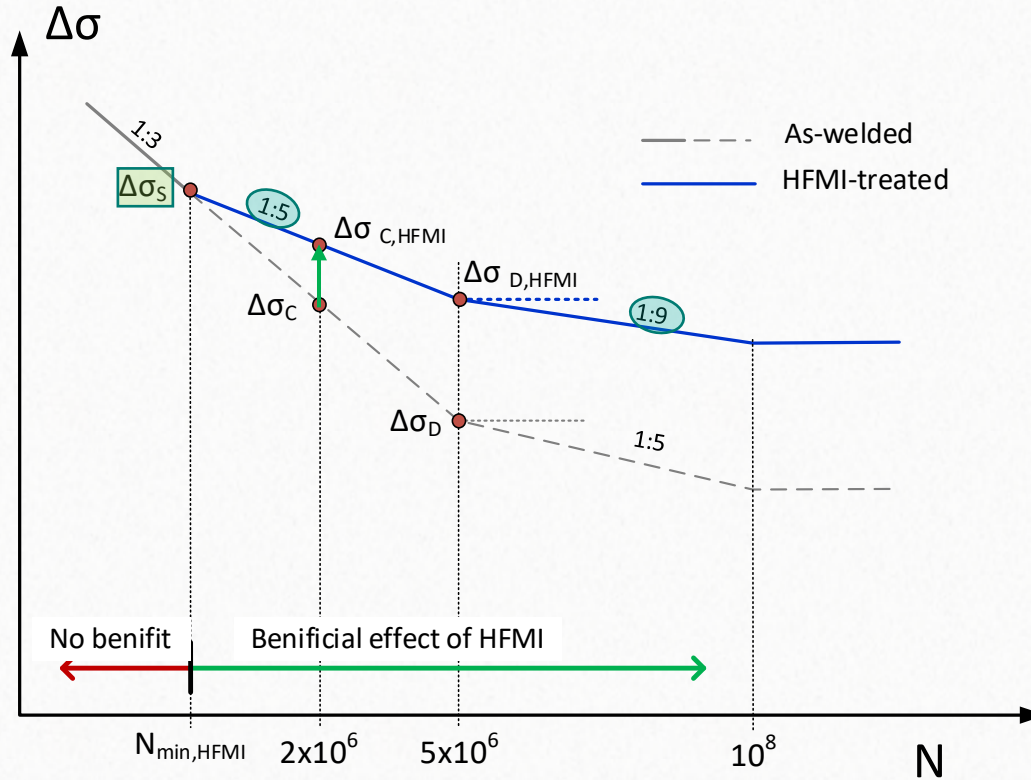
3 $\Delta\sigma_{Ed} \leq \Delta\sigma_{allowable}$

Other design recommendations:

Gary B. Marquis and Zuheir Barsoum (2016): IIW Recommendations for the HFMI Treatment for Improving the Fatigue Strength of Welded Joints. IIW collection, Springer.

Design of HFMI-treated welds (the upcoming Annex F of EN 1993-1-9)

0

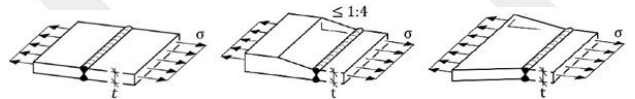
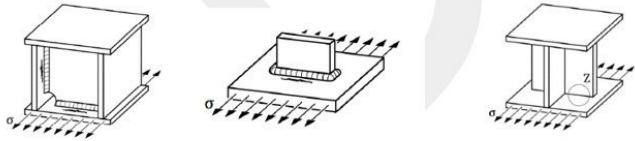
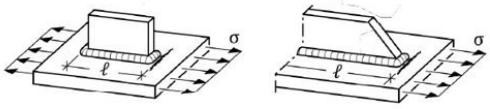


Design of HFMI-treated welds

(the upcoming Annex F of EN 1993-1-9) **0**

Fatigue strength & covered details

$$\Delta\sigma_{C, HFMI, ref} (f_y = 355 \text{ MPa} \ \& \ R = 0.1)$$

| Detail | Description | $\Delta\sigma_{C, HFMI, ref}$ |
|---|---|-------------------------------|
|  | Transverse K- and X-butt welds ^{1, 2)} | $k_S \times 160$ |
|  | Transverse non-load carrying attachments and stiffeners, with fillet or butt welds. | 140 |
|  | End of longitudinal non-load carrying attachments with fillet or butt welds | 100 |

- 1) A thickness correction factor $k_S = \left(\frac{25}{t}\right)^{0.2}$ is applied when plate thickness exceeds 25 mm.
- 2) Plates of equal dimensions or with tapering in width or thickness with slop $\leq 1:4$

Design of HFMI-treated welds

(the upcoming Annex F of EN 1993-1-9)

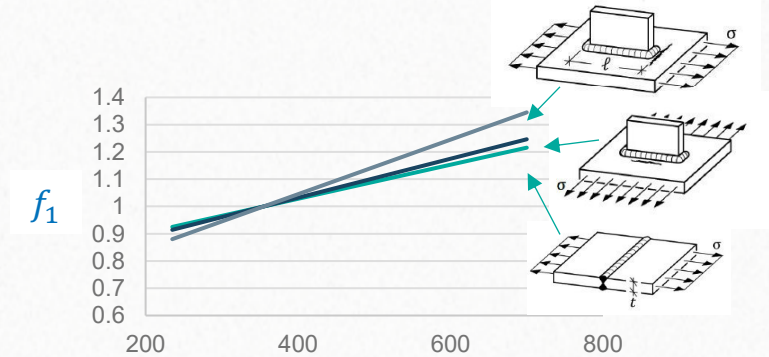


Modification of Fatigue strength – Derived from extensive fatigue testing

$$\Delta\sigma_{C, HFMI} = f_1 f_2 \Delta\sigma_{C, HFMI, ref}$$

1 Correction for f_y

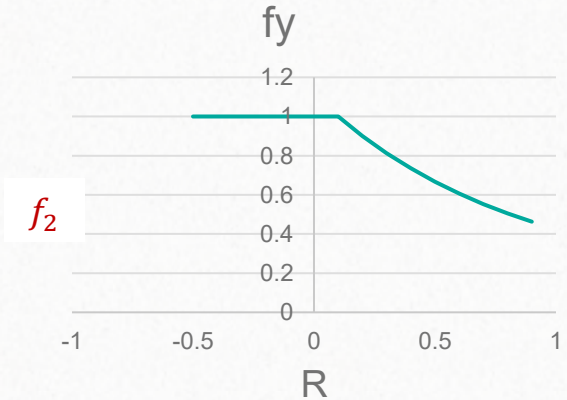
$$f_1 = 1 + \frac{0.1(f_y - 355)}{\Delta\sigma_{C, HFMI, ref}}$$



2 Correction for ***R – ratio***

[CAFL]

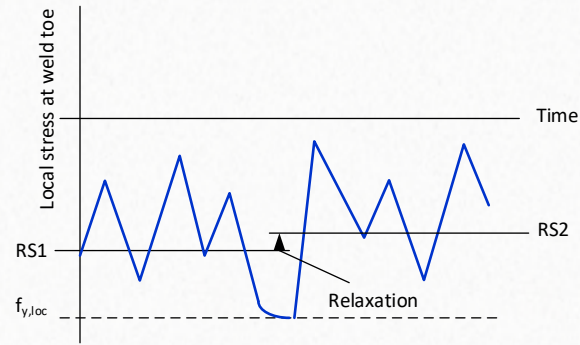
$$f_2 = \frac{1}{0.5R^2 + 0.95R + 0.9} \quad \text{if } 0.1 < R < 1.0, \text{ otherwise, } f_2 = 1.0$$



Design of HFMI-treated welds (the upcoming Annex F of EN 1993-1-9)

Limits for permissible (nominal) stresses

3



Based on extensive experimental work and numerical analysis

| Detail | Limits |
|--------|--------------------------------------|
| | $-0.9f_y \leq \sigma_{max} \leq f_y$ |
| | $-0.7f_y \leq \sigma_{max} \leq f_y$ |
| | $-0.5f_y \leq \sigma_{max} \leq f_y$ |

Design of HFMI-treated welds in bridges

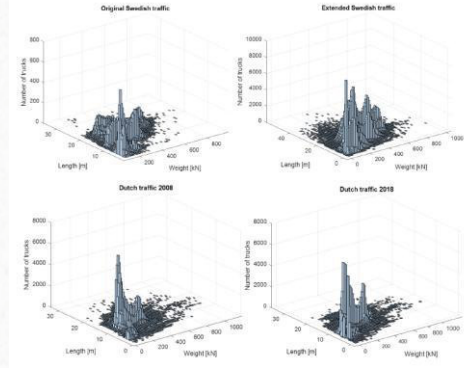


Two main questions remain:

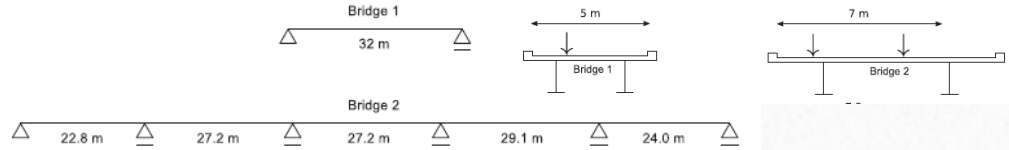
- What R-ratios are generated by “**Real**” traffic load (and permanent loads) on bridges?
- What load combination should be used to check max. permissible stresses?

Design of HFMI-treated welds

Derivation of **R-Ratio** effect for railway and road traffic



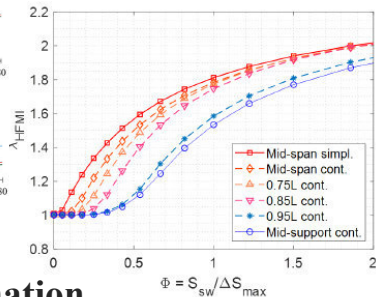
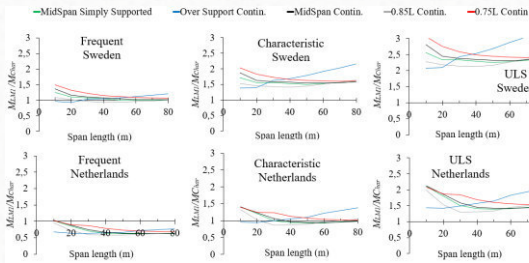
Simulations on bridges with different spans and influence lines



Measured traffic loads

873,000 lorries in Sweden

446,000 lorries in the Netherlands



Derivation of "correction" factors

$$\lambda_{HFMI}$$

FLM4

Moment response

$$f_i = 0.5R_i^2 + 0.95R_i + 0.9$$

$$\Delta S_{eq} = \sqrt{\frac{\sum (n_i \cdot \Delta S_i^m)}{\sum n_i}}$$

$$\Delta S_{eqR} = \sqrt{\frac{\sum (n_i \cdot (\Delta S_i \cdot f_i)^m)}{\sum n_i}}$$

$$\lambda_{HFMI} = \frac{\Delta S_{eqR}}{\Delta S_{eq}}$$

FLM3

Moment response

$$f = 0.5R^2 + 0.95R + 0.9$$

$$\lambda_{HFMI} = f$$

Correlation to fatigue load models

Use the **characteristic load combination**

Design of HFMI-treated welds

Fatigue verification method – Road bridges

$$\frac{\Delta\sigma_{e,2,HFMI,Ed}}{f_1 \times \Delta\sigma_{C,HFMI,ref} / \gamma_{Mf}} < 1.0$$

Reference fatigue strength
modified wrt f_y

$$\Delta\sigma_{e,2,HFMI,Ed} = \lambda_1 \lambda_2 \lambda_3 \lambda_4 \lambda_{HFMI} \Delta\sigma_{Ed}$$

Damage equivalent factor
to account for mean stress
effect (R)

Design of HFMI-treated welds

Fatigue verification method – Road bridges

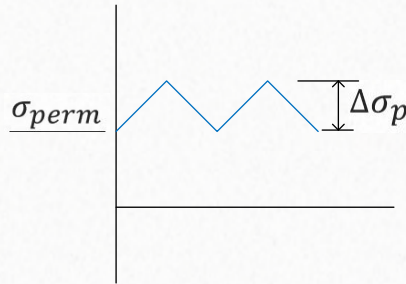


For road bridges

$$\lambda_{HFMI} = \frac{2.38\Phi + 0.64}{\Phi + 0.66} \geq 1.0 \quad \text{Midspan}$$

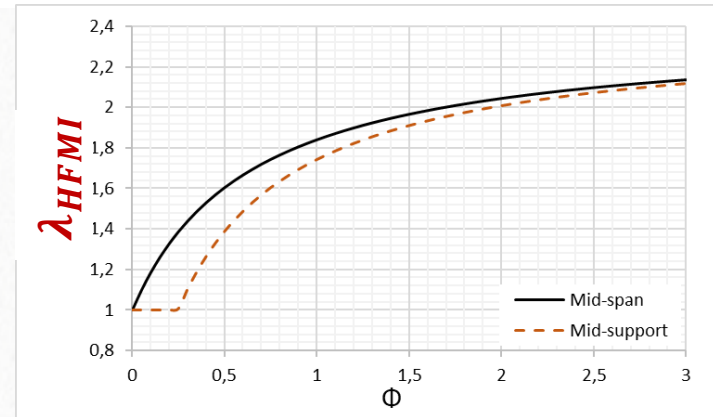
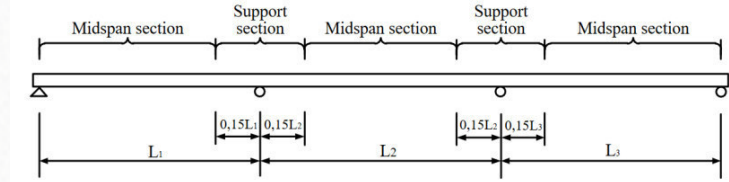
$$\lambda_{HFMI} = \frac{2.38\Phi + 0.06}{\Phi + 0.40} \geq 1.0 \quad \text{Mid-support}$$

$$\Phi = \frac{\sigma_{perm}}{2 \times \Delta\sigma_p}$$



$\Delta\sigma_p$ is the stress range generated by the passage of FLM3.

σ_{perm} is the stress from permanent loads.



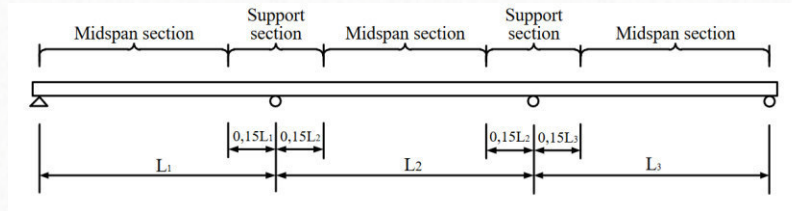
OBS! If HFMI-treatment is performed on-site (after bridge erection) λ_{HFMI} can be obtained from $\Phi = 0$, i.e. no mean stress effect.

Design of HFMI-treated welds

Fatigue verification method – Railway bridges



For railway bridges

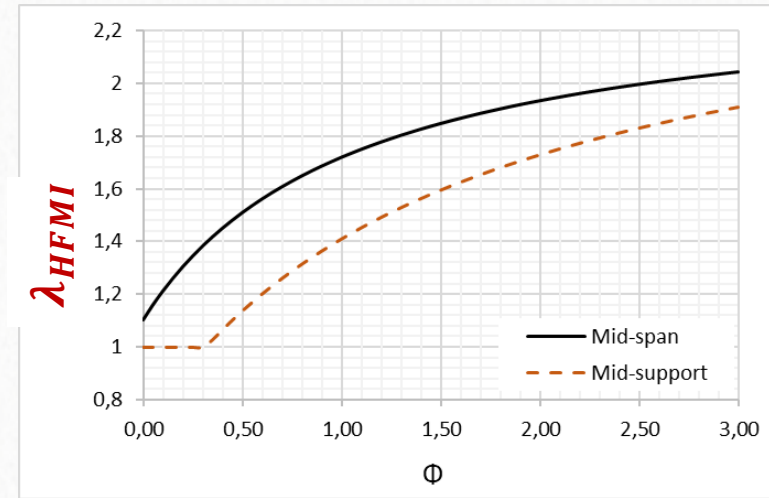


$$\lambda_{HFMI} = \frac{2.38\Phi + 1.18}{\Phi + 1.07} \geq 1.0 \quad \text{Midspan}$$

$$\lambda_{HFMI} = \frac{2.56\Phi + 1.12}{\Phi + 1.61} \geq 1.0 \quad \text{Mid-support}$$

$$\Phi = \frac{\sigma_{perm}}{0.73 \times \Delta\sigma_{LM71}} \quad \text{(with FLM71)}$$

$$\Phi = \frac{\sigma_{perm}}{0.90 \times \Delta\sigma_{max}} \quad \text{(with train mixes)}$$

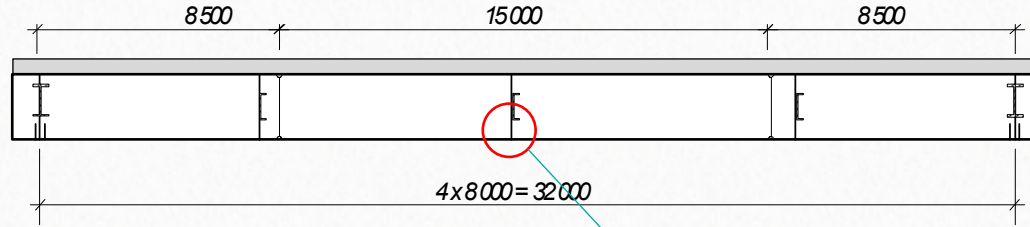
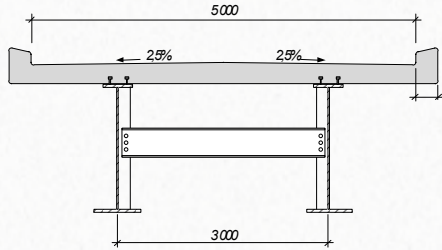


OBS! If HFMI-treatment is performed on-site (after bridge erection) λ_{HFMI} can be obtained from $\Phi = 0$, i.e. no mean stress effect.

Design example – Road bridge



CHALMERS

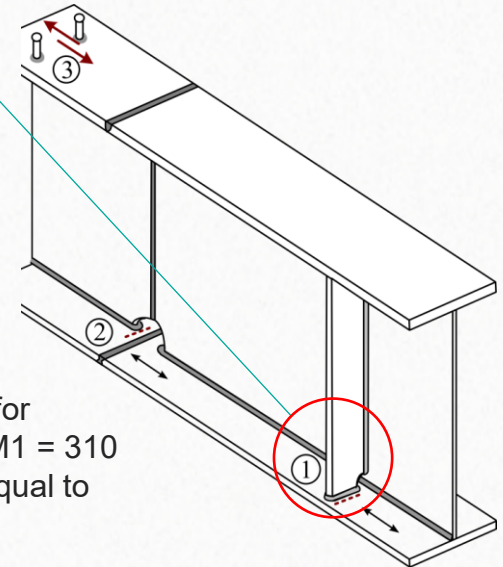


HFMI performed in workshop

Table 3: Bridge data for the calculation example

| | | | |
|--------------------------------------|------------------------|-------------------|---|
| Material yield stress | f_y [MPa] | 690 | ← |
| Bridge beam section modulus | W [mm ³] | 3.6×10^7 | |
| Stress from permanent load on detail | σ_{perm} [MPa] | 120 | ← |
| Partial factor on load | γ_{Ff} | 1.0 | |
| Partial factor on resistance | γ_{Mf} | 1.35 | |
| Design life | Years | 80 | |
| N_{obs} | Cycles | 50 000 | |

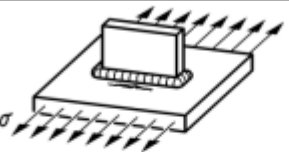
The average weight of the lorry for regional traffic is assumed to $QM1 = 310$ kN. The reference value $Q0$ is equal to 480 kN



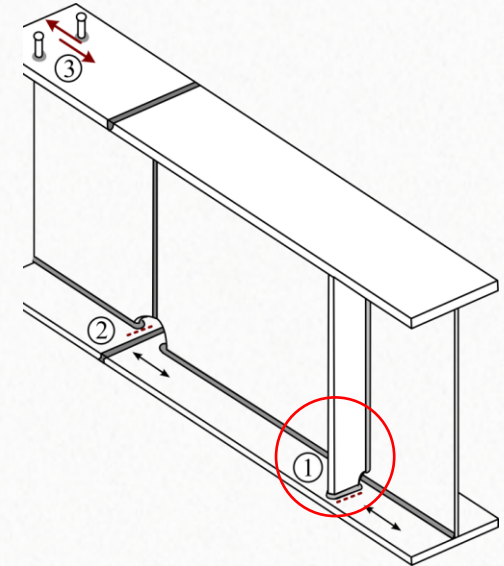
Design example – Road bridge

1) Find $\Delta\sigma_{C,HFMI,ref}$ for the governing detail ($f_y = 355$ och $R = 0.1$)

Table F.2 — Reference value of detail category $\Delta\sigma_{C,HFMI}$ of HFMI treated transverse stiffeners details due to qualified HFMI treatments for the nominal stress method

|  Steel grade according to EN 10025 | Detail category ^{a,b} | | |
|--|-----------------------------------|-----|-----|
| | Stress ratio R [-] ^c | | |
| | -1,0 | 0,1 | 0,5 |
| S235 ≤ S < S355 | 125 | 125 | 80 |
| S355 ≤ S < S650 | 160 | 140 | 90 |
| S650 ≤ S ≤ S700 | 160 | 160 | 125 |

^a Table applies for $\ell \leq 50\text{mm}$; if $50 < \ell \leq 80\text{mm}$, $\Delta\sigma_{C,HFMI}$ should be reduced by one detail category.
^b Limitations of applied stresses calculated according to F.3.1: $-0,7 f_y < \sigma \leq f_y$.
^c For other stress ratios R , linear interpolation is allowed.



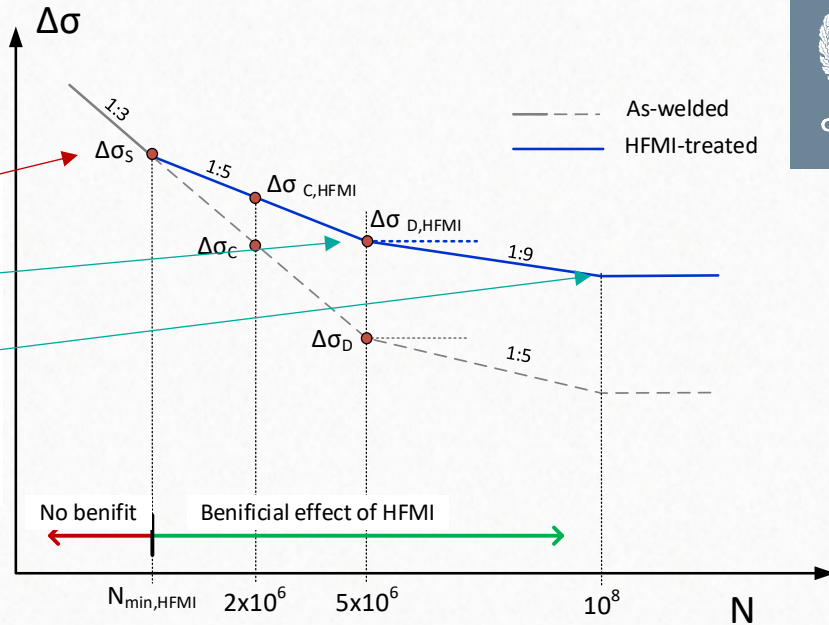
2) Construct the S-N curve

$$\Delta\sigma_{C,HFMI,ref} = 140 \text{ MPa}$$

$$\sigma_{D,HFMI} = 0.833 \times \Delta\sigma_{C,HFMI} = 116.6 \text{ MPa}$$

$$\Delta\sigma_{L,HFMI} = 0.717 \times \Delta\sigma_{D,HFMI} = 83.6 \text{ MPa}$$

$$\Delta\sigma_s = \left(\frac{\Delta\sigma_{C,HFMI}^5}{\Delta\sigma_C^3} \right)^{0.5} = 324.1 \text{ MPa}$$

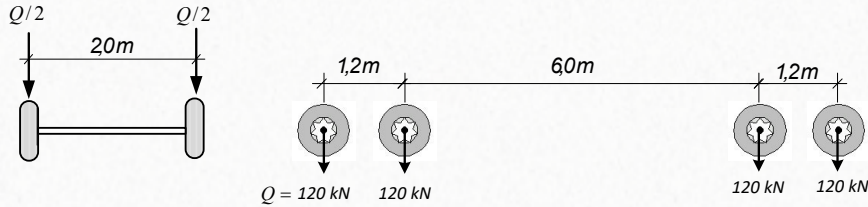


3) Correction for f_y

$$f_1 = 1 + \frac{0.1 (f_y - 355)}{\Delta\sigma_{C,HFMI,ref}} = 1 + \frac{0.1 (690 - 355)}{140} = 1,24$$

OBS! $1,24 \times 140 = 174 \text{ MPa} \rightarrow$ **Less than the fatigue strength of the base metal (DC180)**

(I) Fatigue verification with FLM3



$$M_{max} (FLM3) = 2976 \text{ kNm}$$

$$\Delta\sigma_{Ed} = \Delta\sigma_P = 82.7 \text{ MPa}$$

Table 3: λ -coefficients for fatigue verification with FLM3.

| λ coefficient | <u>Takes into account</u> | Value |
|-----------------------|---------------------------|---|
| λ_1 | Bridge length | $2.55 - 0.7(L-70)/70 = 2.33$ |
| λ_2 | Actual traffic flow | $(Q_{MI}/Q_0) \cdot (N_{obs}/N_0)^{1/5} = 0.407$ |
| λ_3 | Design fatigue life | $(t/100)^{1/5} = 0.956$ |
| λ_4 | Interaction of lanes | 1 (for single laned traffic) |
| λ_{max} | Maximum λ value | 2 (for bridge longer than 25 m) |
| λ | Damage equivalent factor | $\lambda_1 \lambda_2 \lambda_3 \lambda_4 < \lambda_{max} = 0.907$ |

(I) Fatigue verification with FLM3

$$\frac{\Delta\sigma_{e,2,HFMI,Ed}}{f_1 \times \Delta\sigma_{C,HFMI,ref} / \gamma_{Mf}} < 1$$

$\Delta\sigma_{e,2,HFMI,Ed} = 128 \text{ MPa}$ (from calculation below)
 $f_1 = 1,24$
 $\Delta\sigma_{C,HFMI,ref} = 140 \text{ MPa}$
 $\gamma_{Mf} = 1,35$

$$\Delta\sigma_{e,2,HFMI,Ed} = \lambda \times \lambda_{HFMI} \Delta\sigma_E \gamma_{Ff}$$

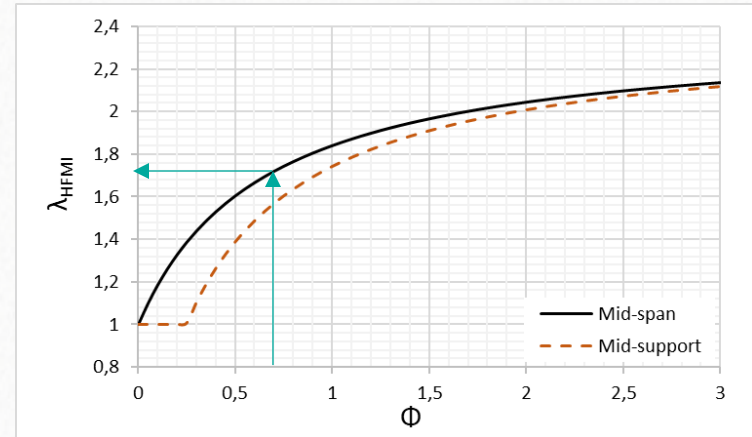
$\lambda = 0,907$
 $\lambda_{HFMI} = 1,71$
 $\Delta\sigma_E = 82,7 \text{ MPa}$
 $\gamma_{Ff} = 1,0$

$$\lambda_{HFMI} = \frac{2.38\Phi + 0.64}{\Phi + 0.66} = \frac{2.38 \cdot \left(\frac{120}{2 \times 82.7}\right) + 0.64}{\left(\frac{120}{2 \times 82.7}\right) + 0.66} = 1,71$$

$$\text{with } \Phi = \frac{\sigma_{perm}}{2 \times \Delta\sigma_p} = \frac{120}{2 \times 82.7} = 0,73$$

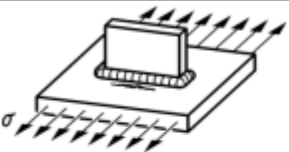
$$\Delta\sigma_{e,2,HFMI,Ed} = 0.907 \times 1.71 \times 82.7 \times 1.0 = 128 \text{ MPa}$$

$$\text{Verification: } \frac{128}{1,24 \times 140 / 1,35} = \mathbf{0.99} < 1.0$$



Verification of permissible stresses

Table F.2 — Reference value of detail category $\Delta\sigma_{C,HFMI}$ of HFMI treated transverse stiffeners details due to qualified HFMI treatments for the nominal stress method

|  | Detail category ^{a,b} | | |
|---|-----------------------------------|-----|-----|
| | Stress ratio R [-] ^c | | |
| Steel grade according to EN 10025 | -1,0 | 0,1 | 0,5 |
| S235 ≤ S < S355 | 125 | 125 | 80 |
| S355 ≤ S < S650 | 160 | 140 | 90 |
| S650 ≤ S ≤ S700 | 160 | 160 | 125 |

^a Table applies for $\ell \leq 50\text{mm}$; if $50 < \ell \leq 80\text{mm}$, $\Delta\sigma_{C,HFMI}$ should be reduced by one detail category.
^b Limitations of applied stresses calculated according to F.3.1: $-0,7 f_y < \sigma \leq f_y$.
^c For other stress ratios R , linear interpolation is allowed.

Characteristic load combination

$$SW + (1 \text{ or } 0) \times S + TS + UDL + 0.6 \times \max(F_w, T_k) = 300 \text{ MPa} = 0.45 f_y$$

(II) Fatigue verification with FLM4

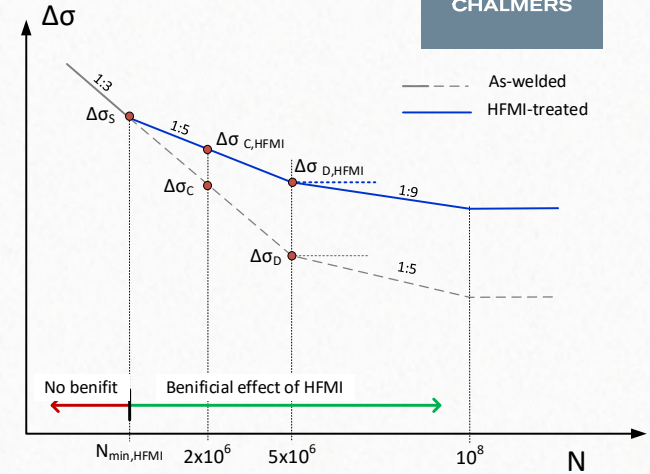
(Local traffic)



Table 4: Maximum moments and stress ranges at the location of the studied detail due to the passage of different vehicles of FLM4.

| Load | M_{max} (kN.m) | Stress range (MPa) |
|----------------|------------------|-------------------------|
| FLM4 (Lorry 1) | 1443 | $\Delta\sigma_1 = 40$ ✘ |
| FLM4 (Lorry 2) | 2255 | $\Delta\sigma_2 = 63$ |
| FLM4 (Lorry 3) | 3061 | $\Delta\sigma_3 = 85$ |
| FLM4 (Lorry 4) | 2380 | $\Delta\sigma_4 = 66$ |
| FLM4 (Lorry 5) | 2668 | $\Delta\sigma_5 = 74$ |

} slope 9



$$\frac{\Delta\sigma_{D,HFMI,ref}}{\gamma_{Mf}} = 86 \text{ MPa}$$

$$\frac{\Delta\sigma_{L,HFMI,ref}}{\gamma_{Mf}} = 62 \text{ MPa}$$

$$\frac{\Delta\sigma_S}{\gamma_{Mf}} = \frac{\left(\frac{140^5}{80^3}\right)^{0.5}}{\gamma_{Mf}} = 324 \text{ MPa}$$

(II) Fatigue verification with FLM4



$$\sigma_{eq} = \sqrt[9]{\frac{\sum_{j=1}^4 (n_j \times \Delta \sigma_j^9)}{\sum n_j}} = \sqrt[9]{\frac{(2500 \times 63^9 + 2500 \times 85^9 + 2500 \times 66^9 + 2500 \times 74^9)}{50,000}} = 63.5 \text{ MPa}$$

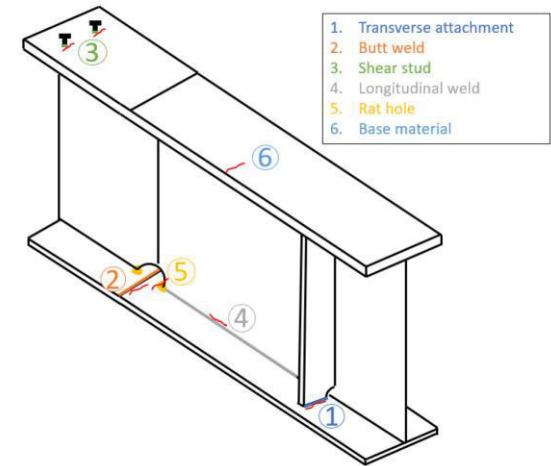
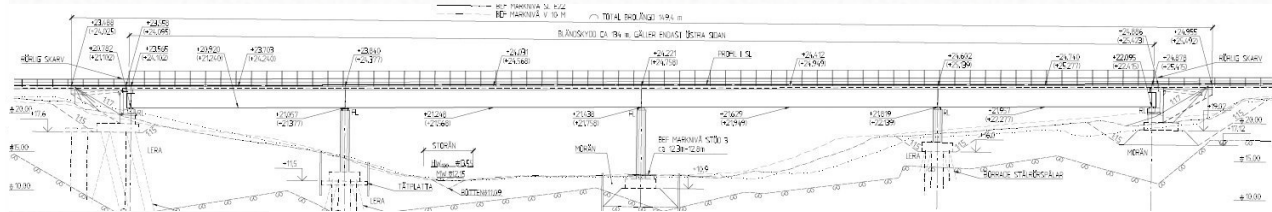
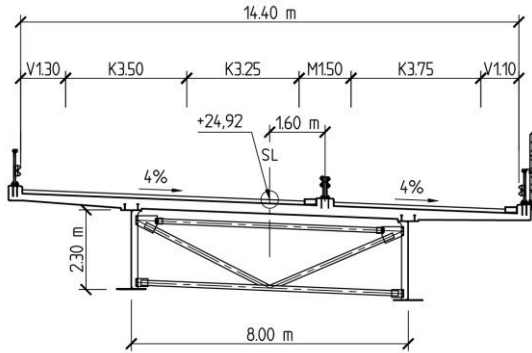
$$N_{eq} = 5.10^6 \left(\frac{f_1 \cdot \Delta \sigma_{D,HFMI.ref} / \gamma_{Mf}}{\lambda_{HFMI} \cdot \Delta \sigma_{eq} \cdot \gamma_{Ff}} \right)^{m_2} = 5.10^6 \left(\frac{\frac{1.24 \times 116.6}{1.35}}{1.71 \times 63.5 \times 1.0} \right)^9 = 4.4 \times 10^6 \text{ cycles}$$

$$D = \frac{n}{N_{EQV}} = \frac{50,000 \times 80}{4.4 \times 10^6} = \mathbf{0.9}$$

Compare to **0.99** with the simplified λ – method

Potential weight and cost saving employing HFMI

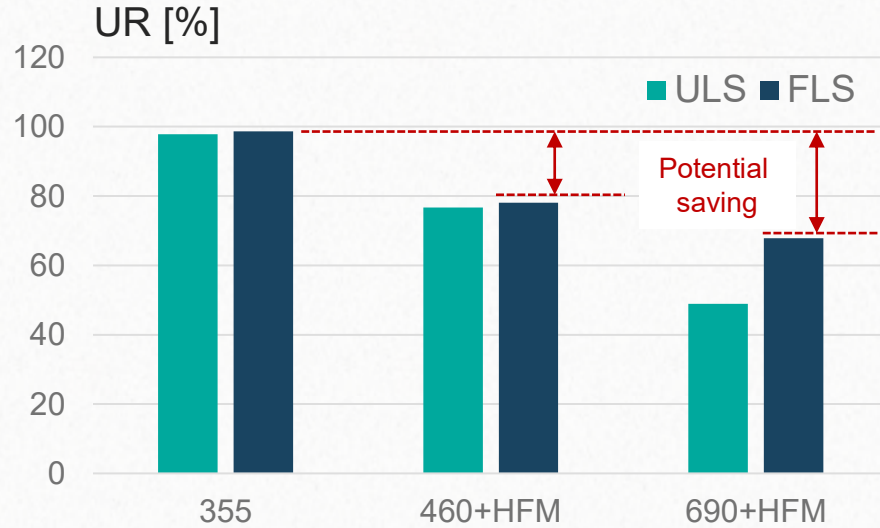
Case-study bridge (COWI)



| Details subjected to normal stresses: | $\Delta\sigma_c$ [MPa] | $\Delta\sigma_{c,HFMI,ref}$ [MPa] |
|---------------------------------------|------------------------|-----------------------------------|
| 1. Transverse attachment | 80 | 140 |
| 2. Butt-weld | 112 | 160 |
| 4. Longitudinal weld | 112 | - |
| 5. Rat-hole | 71 | 100 |
| 6. Base metal | 160 | - |
| Details subjected to shear stresses: | $\Delta\tau_c$ [MPa] | $\Delta\tau_{c,HFMI,ref}$ [MPa] |
| 3. Shear stud | 90 | - |

Potential weight and cost saving employing HFMI

Case-study bridge (COWI)



Utilization ratios with increased material strength and HFMI
(original bridge design / sections)

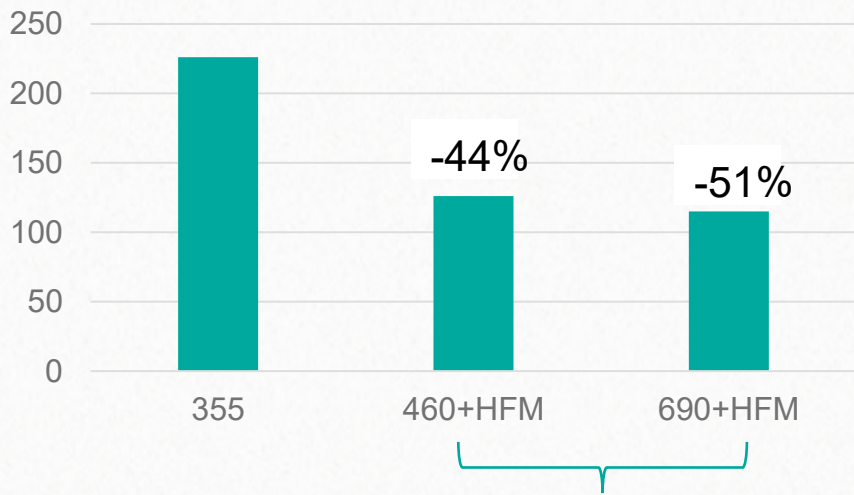
Potential weight and cost saving employing HFMI

Case-study bridge (COWI)



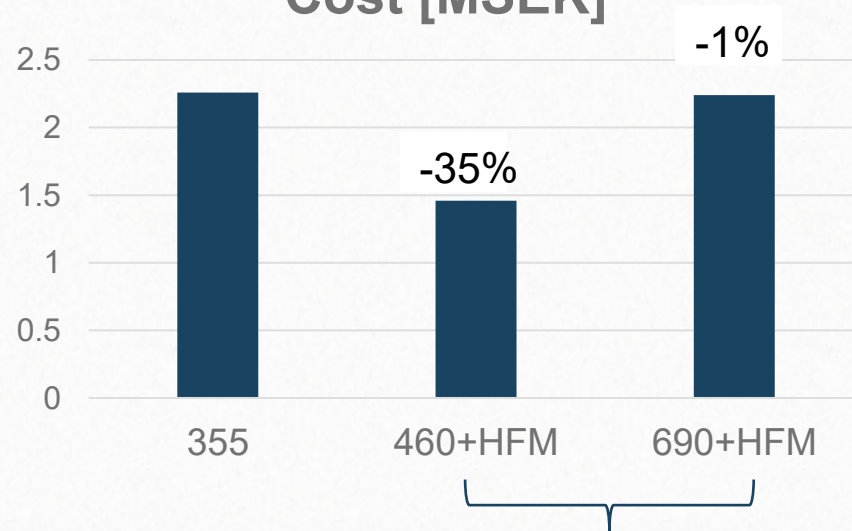
| Material | Unit price |
|----------|------------|
| S355 | 1.000 |
| S460 | 1.083 |
| S690 | 1.924 |

Weight [ton]



Re-designed bridge sections

Cost [MSEK]



Re-designed bridge sections

Some references

1. Hassan Al-Karawi, John Leander and Mohammad Al-Emrani (2023): Verification of the Maximum Stresses in Enhanced Welded Details via High-Frequency Mechanical Impact in Road Bridges. Buildings 13(2):364. DOI: [10.3390/buildings13020364](https://doi.org/10.3390/buildings13020364)
2. Poja Shams-Hakimi, Hassan Al-Karawi, and Mohammad Al-Emrani. "High-cycle variable amplitude fatigue experiments and design framework for bridge welds with high-frequency mechanical impact treatment." Steel Construction 15.3 (2022): 172-187. DOI: [10.1002/stco.202200003](https://doi.org/10.1002/stco.202200003)
3. Hassan Alkarawi, Poja Shams-Hakimi and Mohammad Al-Emrani (2022): Mean Stress Effect in High-Frequency Mechanical Impact (HFMI)-Treated Steel Road Bridges. Buildings 12(5):545. DOI: [10.3390/buildings12050545](https://doi.org/10.3390/buildings12050545)
4. Poja Shams-Hakimi, Fredrik Carlsson, Mohammad Al-Emrani and Hassan Al-Karawi (2021): Assessment of in-service stresses in steel bridges for high-frequency mechanical impact applications. Engineering Structures 241(3):112498. DOI: [10.1016/j.engstruct.2021.112498](https://doi.org/10.1016/j.engstruct.2021.112498).
5. Hassan Al-Karawi and Mohammad Al-Emrani (2021): The efficiency of HFMI treatment and TIG remelting for extending the fatigue life of existing welded structures. Steel Construction 14(4). DOI: [10.1002/stco.202000053](https://doi.org/10.1002/stco.202000053).
6. Hassan Al-Karawi, Mohammad Al-Emrani and R.U. Franz von Bock und Polach.(2021): Fatigue life extension of existing welded structures via high frequency mechanical impact (HFMI) treatment. Engineering Structures 239(4). DOI: [10.1016/j.engstruct.2021.112234](https://doi.org/10.1016/j.engstruct.2021.112234)
7. Aldén, R., Zuheir Barsoum, Z., Vouristo, T. Mohammad Al-Emrani. (2020). Robustness of the HFMI techniques and the effect of weld quality on the fatigue life improvement of welded joints. Weld World (2020). DOI: [10.1007/s40194-020-00974-4](https://doi.org/10.1007/s40194-020-00974-4).
8. Hassan Al-Karawi, Mohammad Al-Emrani and R.U. Franz von Bock und Polach.(2020): Fatigue crack repair in welded structures via tungsten inert gas remelting and high frequency mechanical impact. Journal of Constructional Steel Research 172:106200, September 2020. DOI: [10.1016/j.jcsr.2020.106200](https://doi.org/10.1016/j.jcsr.2020.106200).
9. Gary B. Marquis and Zuheir Barsoum (2016): IIW Recommendations for the HFMI Treatment for Improving the Fatigue Strength of Welded Joints. IIW collection, Springer.

Thank you for your attention

**3. HFMI FOR FATIGUE LIFE EXTENSION: GUIDELINES,
DIGITAL QA, AND BRIDGE APPLICATIONS (ZUHEIR
BARSOUM, KTH ROYAL INSTITUTE OF TECHNOLOGY)**



EUROPEAN CONVENTION FOR CONSTRUCTIONAL STEELWORK
Bridge Committee

HFMI for Fatigue Life Extension: Guidelines, Digital QA, and Bridge Applications

Professor **Zuheir Barsoum**

KTH Royal Institute of Technology, Stockholm, Sweden

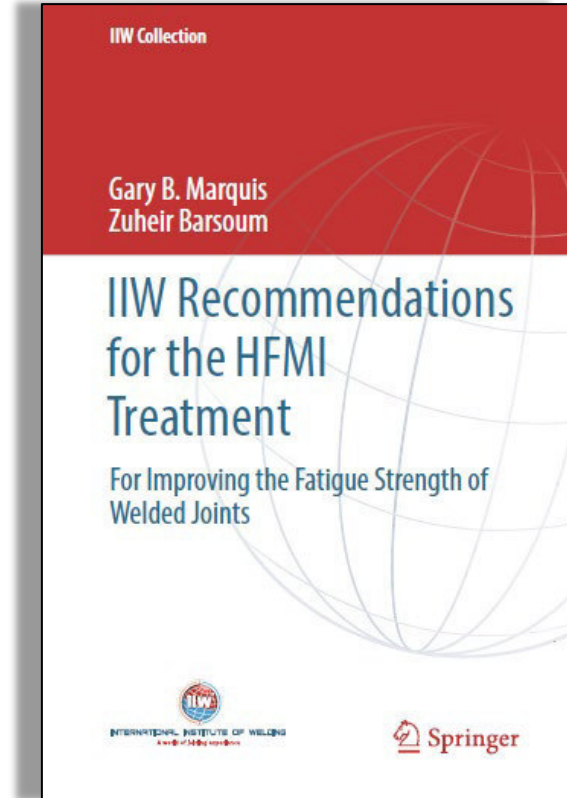
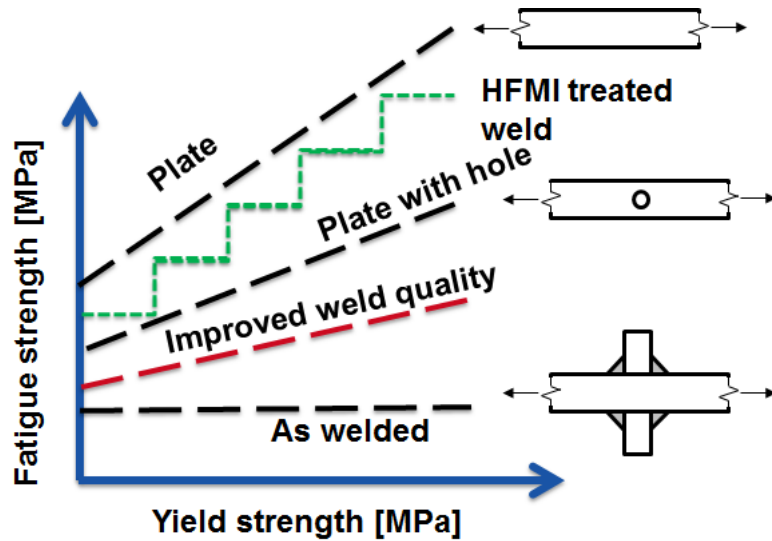




Research in collaboration with



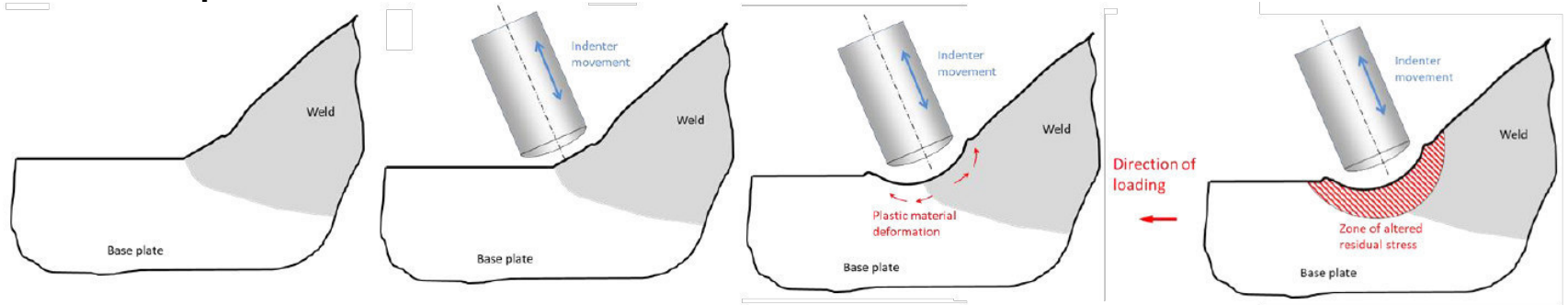
Fatigue Life Enhancement of Welded Structures



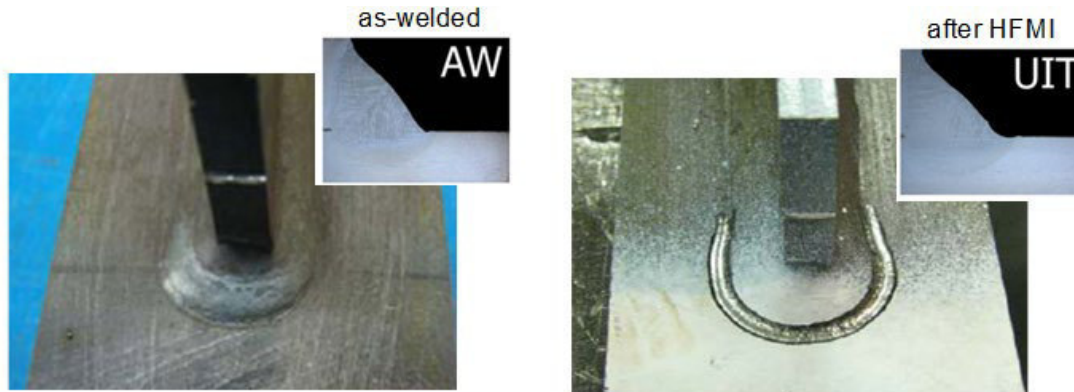
Marquis & Barsoum, Springer (2016)

High Frequency Mechanical Impact (HFMI)

Principle of the HFMI treatment



Typical weld toe profile in the as-welded condition and following HFMI treatment

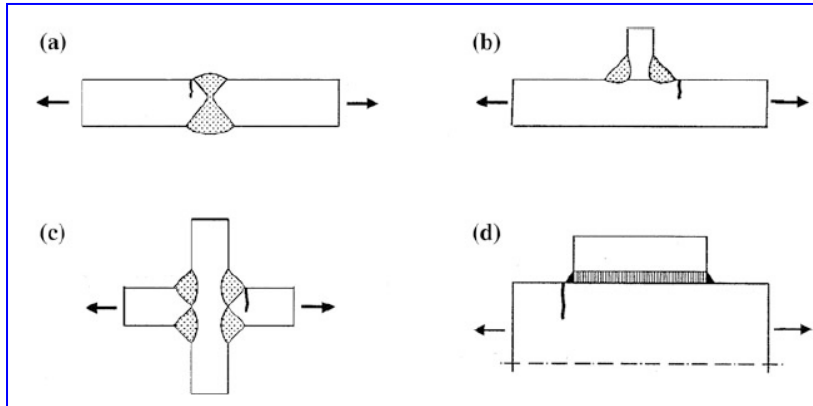


HFMI Fatigue Strength Improvement

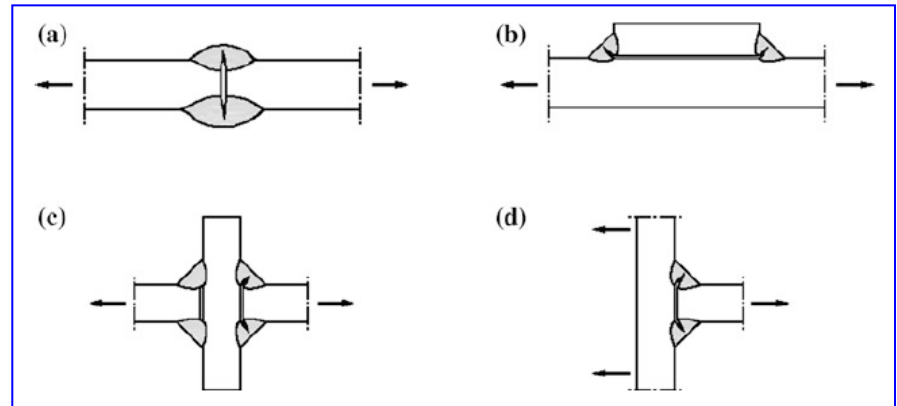
Some of Assumptions:

- The improvement method is applied to the **weld toe**
- The fatigue strength (HFMI improved) is based on an S-N slope of $m = 5$ defined at $N = 2 \cdot 10^6$ cycles

Examples of joints suitable for HFMI improvement

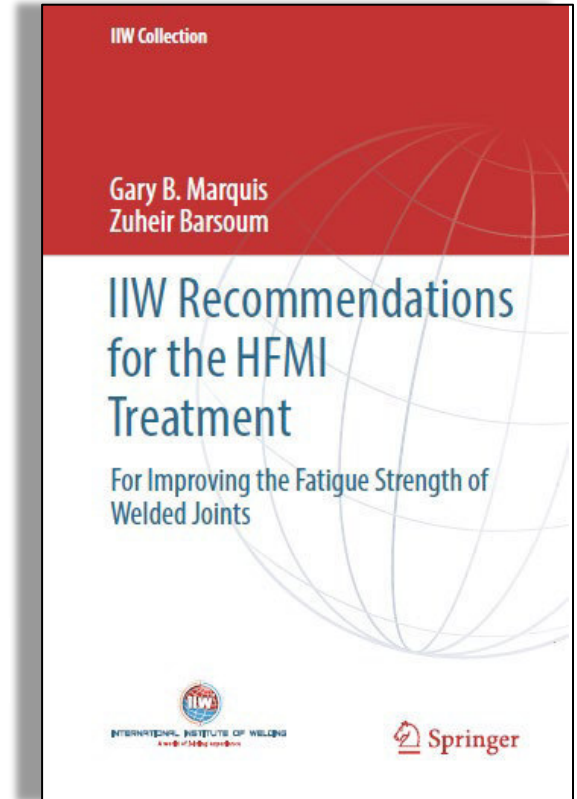
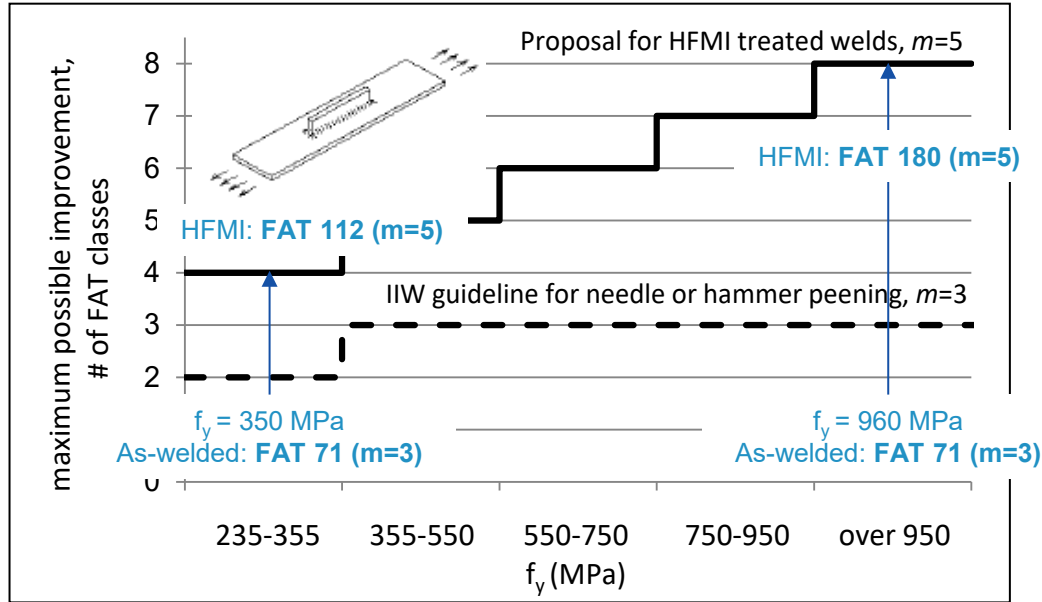


Examples of joints NOT suitable for HFMI improvement



HFMI Fatigue Strength Improvement

One fatigue class increase in strength (about 12.5%) for every 200 MPa increase in static yield strength is shown to be conservative with respect to available data.



Marquis & Barsoum, Springer (2016)



HFMI Treatment – challenges and opportunities

- How robust is the HFMI treatment?
- Is the improvement guaranteed, regardless of the weld quality (ISO 5817) prior the treatment?
- Are the induced compressive residual stresses stable during cyclic loading?
- How does the IIW HFMI recommendations apply for life extension of existing structures?
- How can we, fast and accurately, quality inspect HFMI treatments?

HFMI

Stability of compressive residual stress during cyclic loading

Analytical model for S355/S960

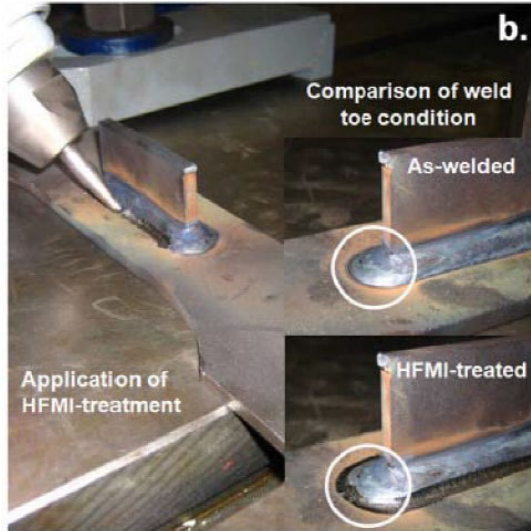
$$\frac{(\sigma_{res})_{relax}}{(\sigma_{res})_{1cycle}} = N^k$$

for $((\sigma_{res})_{ini} + \sigma_{app})/\sigma_y < 1$

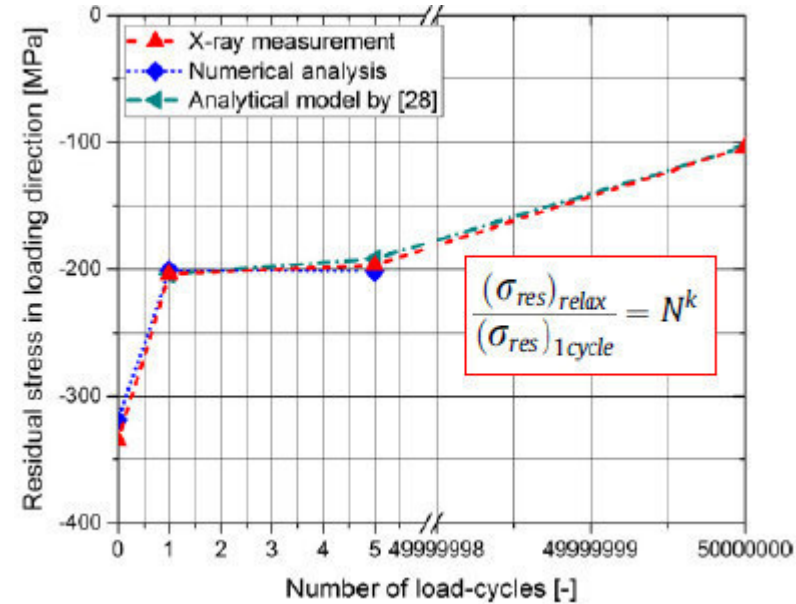
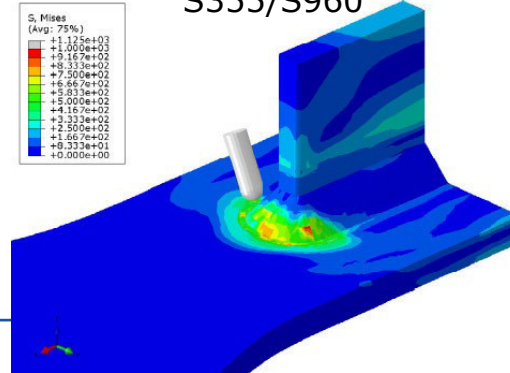
$$\frac{(\sigma_{res})_{relax}}{(\sigma_{res})_{ini}} = N^{-0.004}$$

for $((\sigma_{res})_{ini} + \sigma_{app})/\sigma_y \geq 1$

Experimental model for S355/S960



Numerical/FEM model for S355/S960

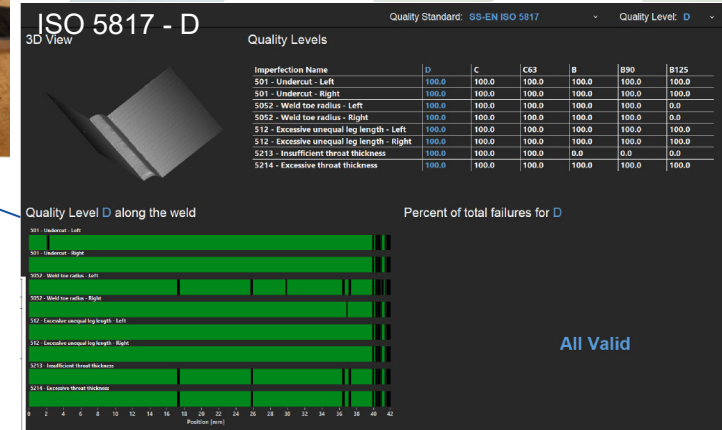
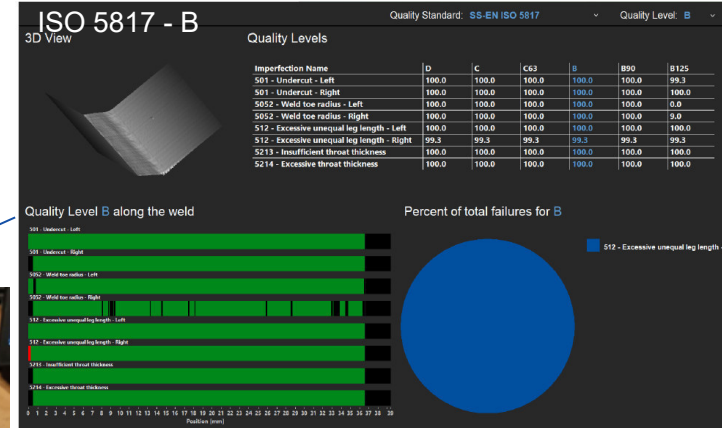
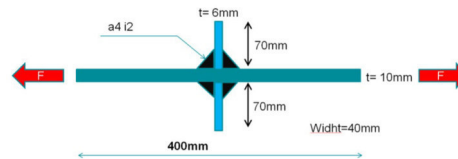
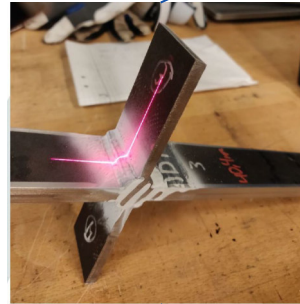


Leitner et al, Journal of Engineering Structures (2017)

HFMI Robustness of the treatment

Background

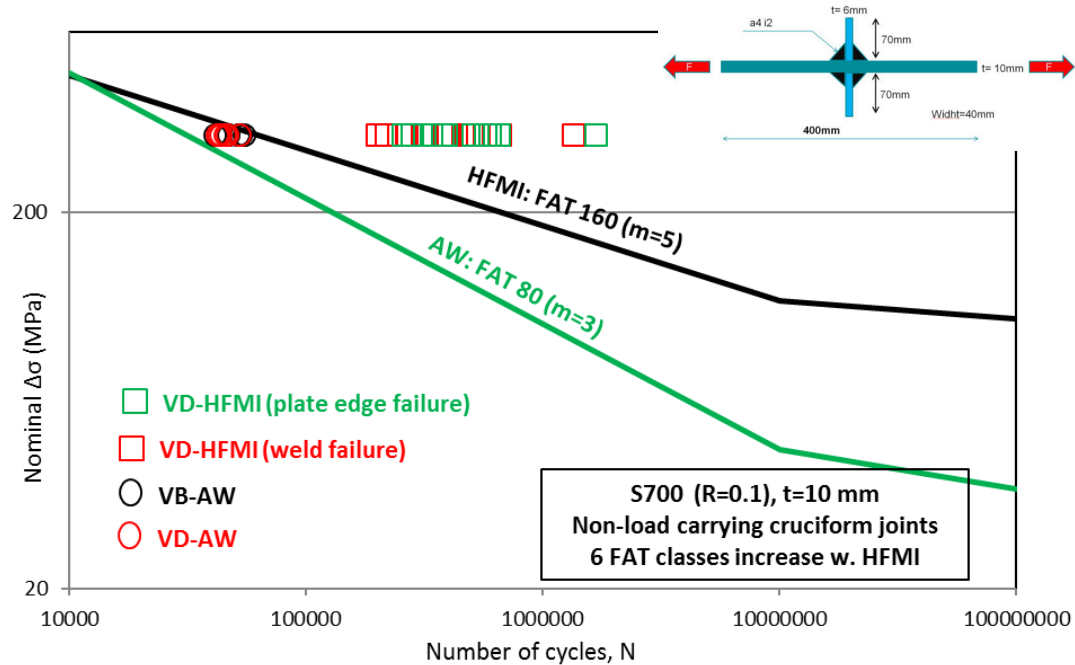
- IIW HFMI recommendations prior treatment: ISO5817-B
- What happens with the fatigue performance when HFMI treatment is performed on welds which do not fulfill the ISO 5817-B?
- As welded quality B (ISO 5817)
- As welded quality D (ISO 5817)
- HFMI treated



Aldén et al *Welding in the World* (2020)

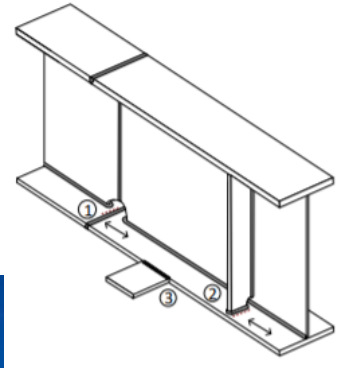
HFMI Robustness of the treatment

- HFMI FAT 160 , $m = 5$ ($S_Y = 700$ MPa)
- As-welded FAT 80, $m = 3$
- The fatigue test results, HFMI, are in good agreement with the recommendations
- HFMI treatment could be done on As-welded joints with **quality D** and still have the same benefits as recommended by IIW

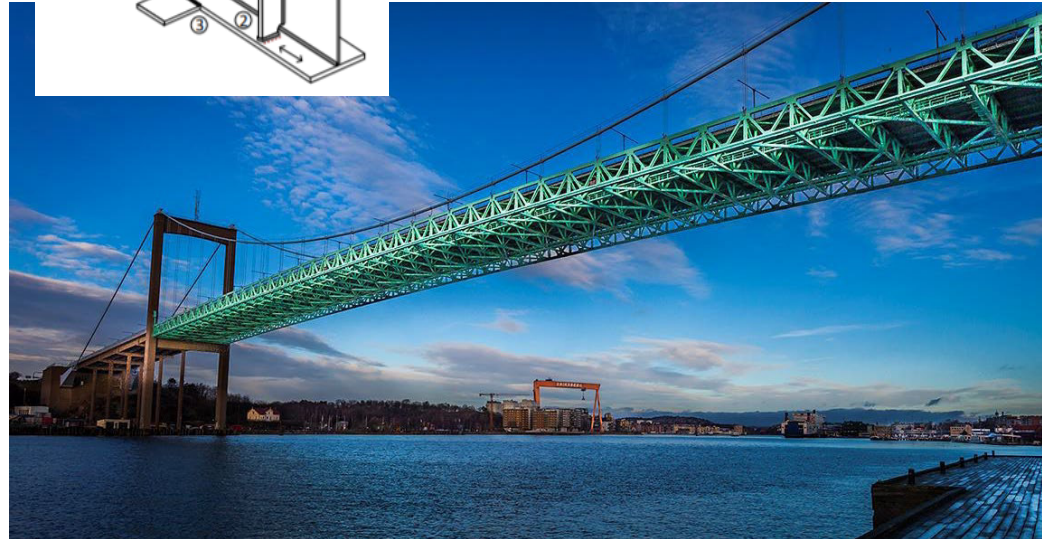


HFMI as a technique for Life extension of infrastructures

- Dramatical increase in the number of **aged steel bridge** is a societal challenge worldwide.
- Aged steel bridges has often **fatigue damages**.
- Utilization of **high-frequency mechanical impact (HFMI)** in fatigue-damaged steel bridges.
- However, **effect of HFMI treatment is sensitive to defect**

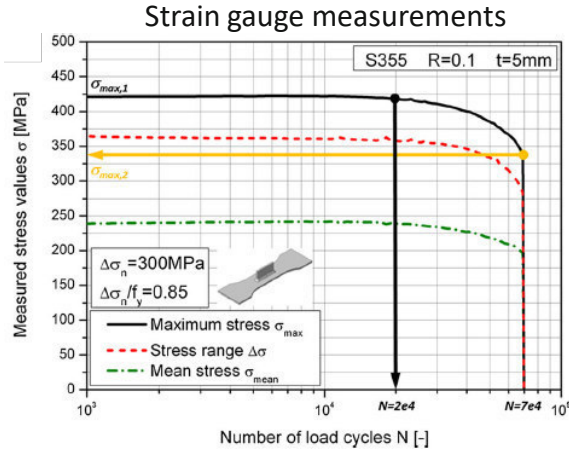


Typical fatigue critical structural details

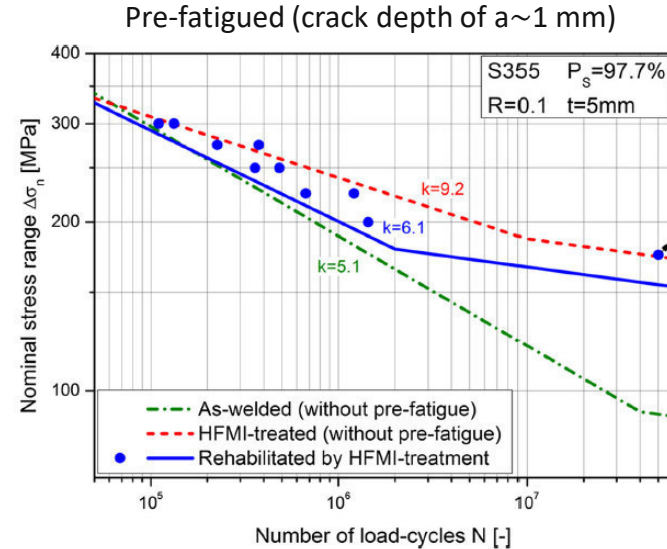
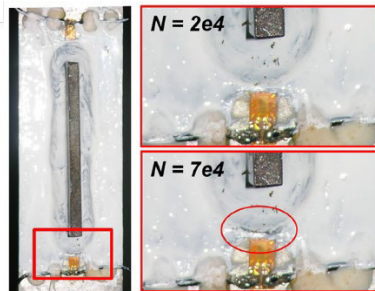


Rehabilitation by HFMI of pre-fatigued welded structures

At which crack sizes is HFMI effective for life extension?



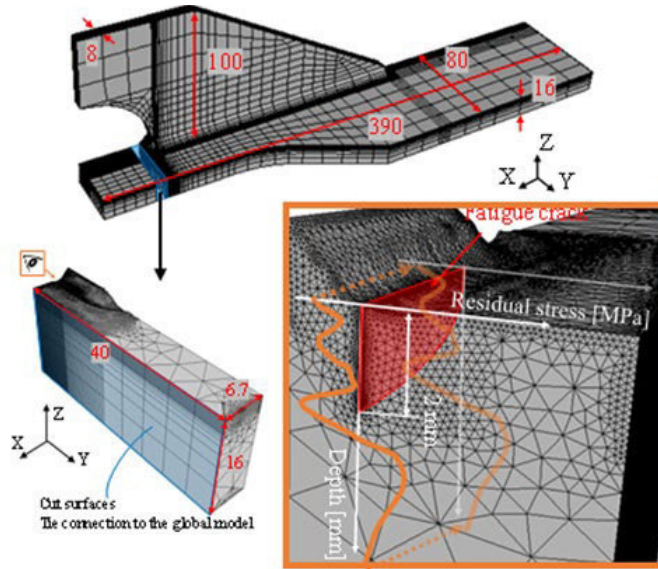
Surface crack detection



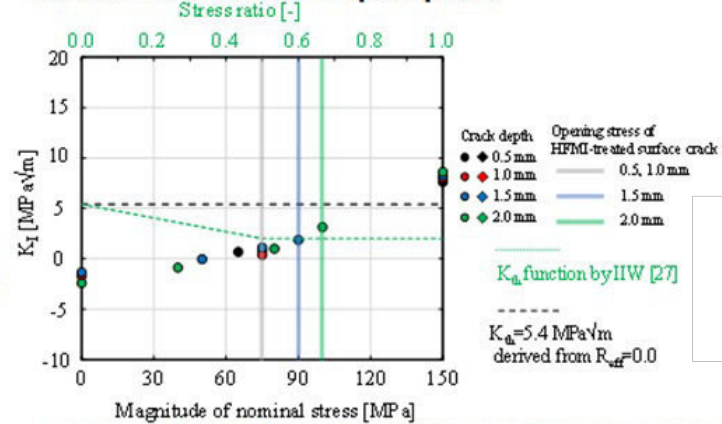
a conservative maximum crack depth of $a_{max} = 0.5$ mm can be recommended leading to an extension by the HFMI rehabilitation.

Defect tolerance of HFMI life extended welds in bridge application

Link 3D HFMI simulation and linear elastic fracture mechanics

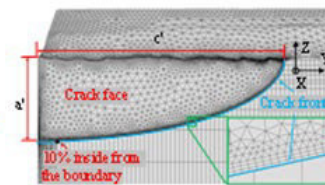
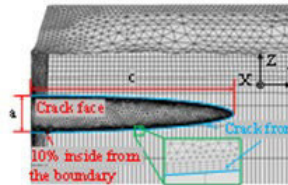


Investigation of crack opening and closing stress at the crack deepest point



Closed surface crack case

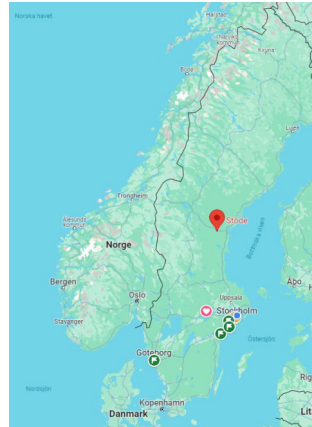
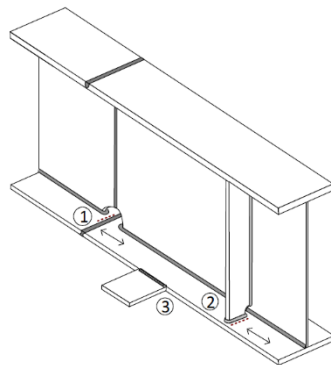
Opened surface crack case



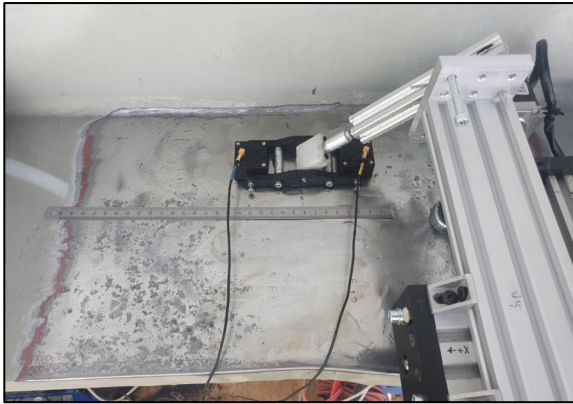
Fatigue cracks less than 1.5 mm in depth, HFMI could be applicable for an effective life extension, which also was experimentally demonstrated by the study

LifeExt Implementation – Prolonged life for existing steel bridges

- Life extension of welded details in a Bridge in Sweden, STÖDE.
- Cracks detection using UT- TOFD (Time of Flight Diffraction Technique)
- HFMI treatment of critical details
- Weld quality control using **Wintertia®** system
 - Prior and post HFMI



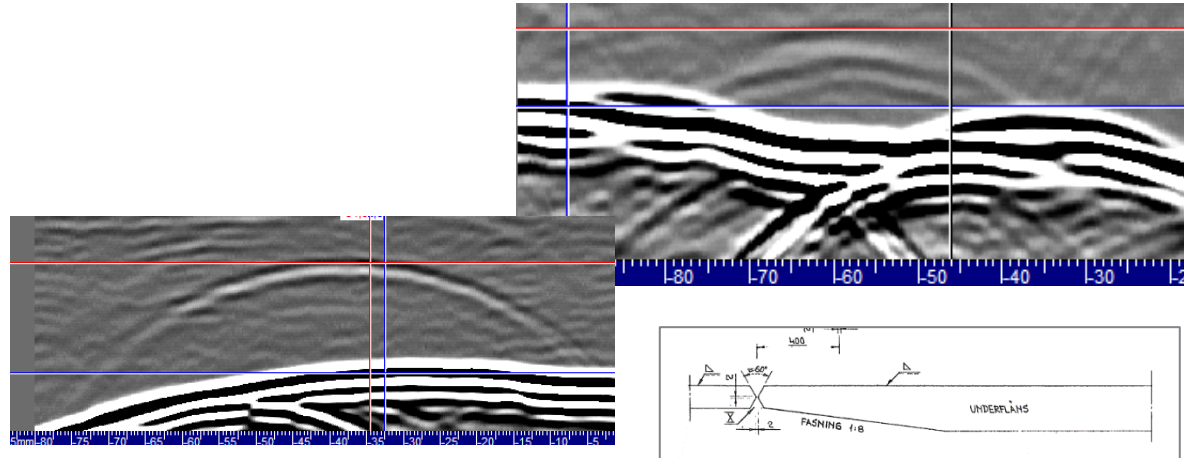
LifeExt Implementation – Prolonged life for existing steel bridges



- Estimation of crack depth – width/depth ratio – prior and after Life extension with HFMI

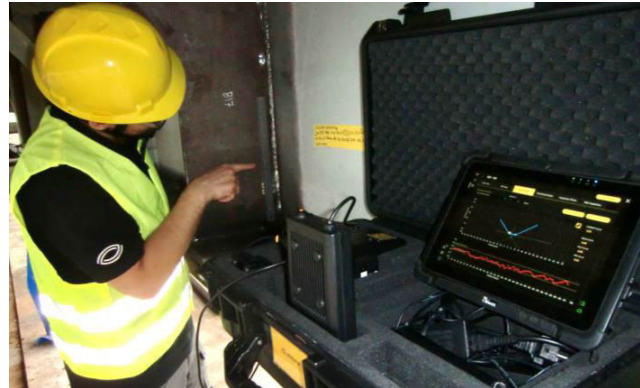
Example:

- Multiple positions with crack like indications, Possible crack – Depth 2.9 mm, width 15 mm

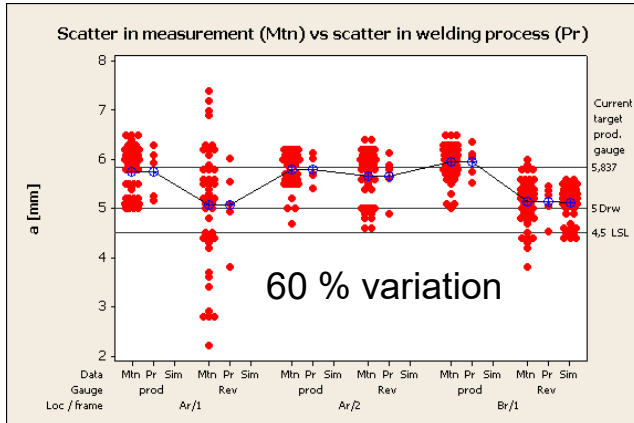


LifeExt Implementation – Prolonged life for existing steel bridges

- Life extension of welded details in a Bridge in Sweden, STÖDE.
- UT-TOFT Defect detection
- Winteria® weld quality inspection
- HFMI treatment
- Winteria® Post Weld quality inspection



Problems with manual weld quality inspection

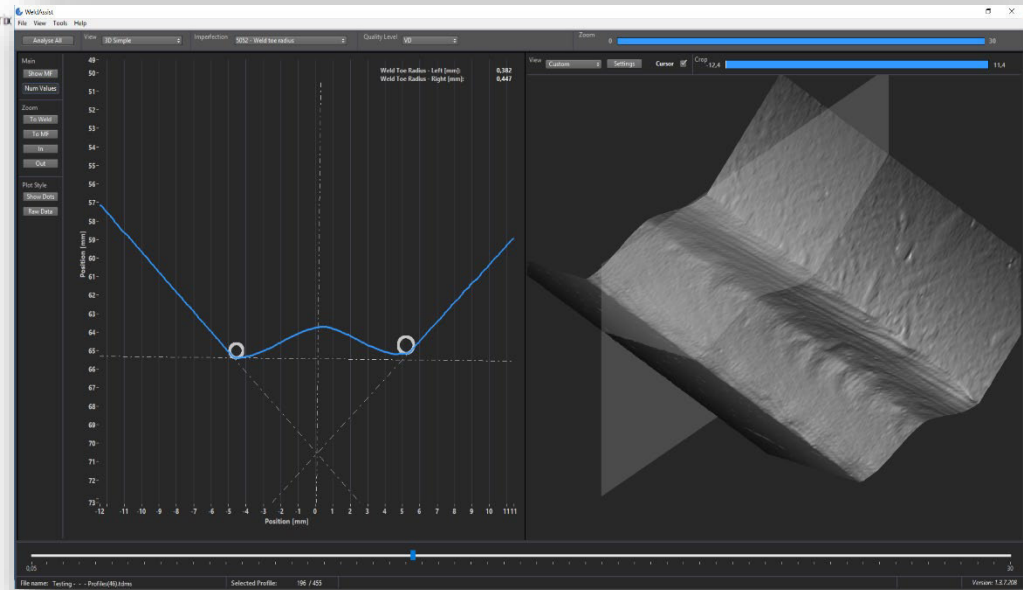
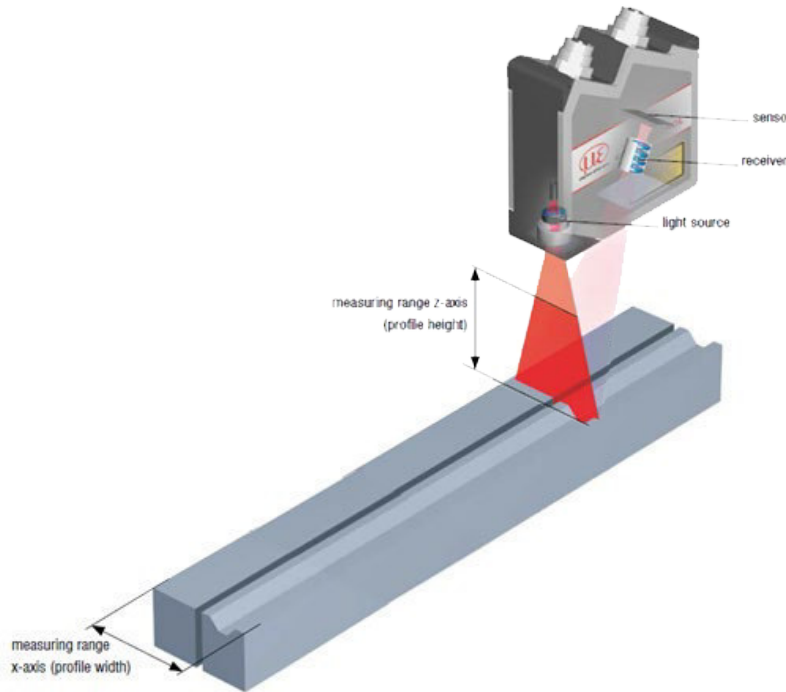


- Poor accuracy
- No data recording
- Slow process
- Repitative work
- Prone to human error

Stenberg et al, *Welding in the World* (2017)

Digitized Quality Assurance of Welded Structures

Winteria® solves the problem of automating digital weld inspection



Digitized Quality Assurance of Welded Structures

Winteria® solves the problem of automating digital weld inspection

Robot installation



Digitized Quality Assurance of Welded Structures

Winteria® solves the problem of automating digital weld inspection

Winteria® FLEX™

Flexible and Easy to Use



Mobility





Outlook (1)

- HFMI have proven to be a robust post weld treatment technique for both new structures and as life extension technique.
- The HFMI recommendations are incorporated in the general IIW recommendations and in EN 1993-3 (Eurocode 3).
- Digitized quality inspection (Winteria®) prior and after post weld treatment have proven to be a very effective tool, both robotized and handheld (FLEX™).
- IIW HFMI recommendations updated and will be published in 2026.

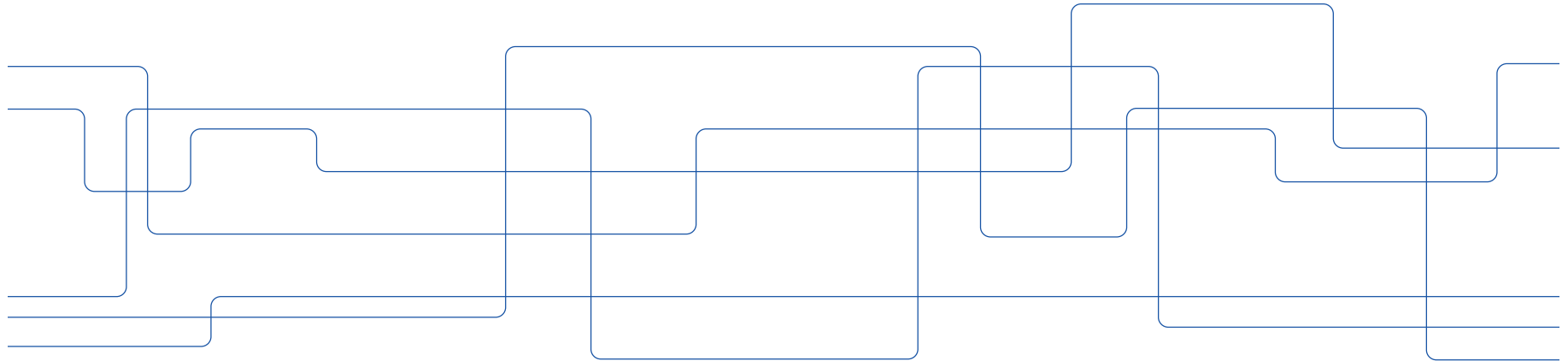


Outlook (2)

Recent Developments in HFMI Treatment Guidelines (Springer 2026), 2nd Edition

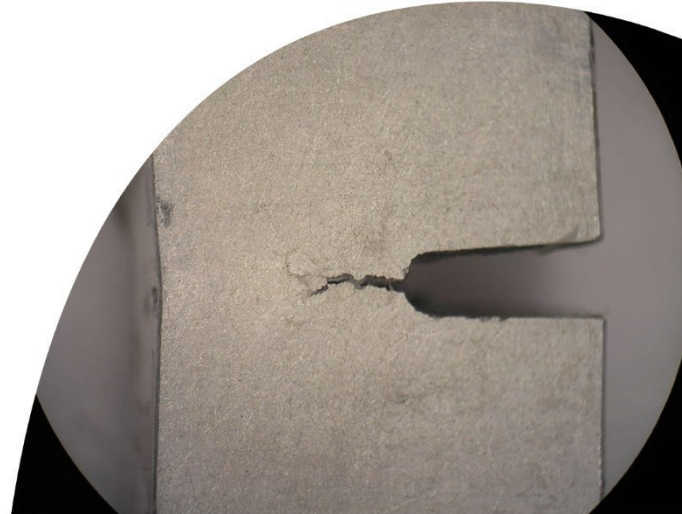
- Expanded material scope (yield strengths up to 1300 MPa, compared to 960 MPa in the previous edition)
- Relaxed weld geometry requirements (weld profiles prior to HFMI treatment, ISO 5817 Quality Level B, have been eased)
- Introduction of numerical fatigue strength modification factors
- Updated thickness correction factors
- Refined S–N curve descriptions
- Inclusion of guidelines for retrofitting pre-fatigued structures
- Standardized verification methodology for HFMI devices
- Updated recommendations for corrosive environments.

Thank you for your
kind attention!



**4. THE DEVELOPMENT AND CHANGES IN PREN1993-1-10
(BERTRAM KÜHN, TECHNISCHE HOCHSCHULE
MITTELHESSEN)**

EUROPEAN CONVENTION FOR CONSTRUCTIONAL STEELWORK Bridge Committee



How to properly select the steel grade concerning toughness requirements and how to extend the fatigue life of steel and steel composite bridges

The development and changes in prEN1993-1-10

Prof. Dr.-Ing. Bertram Kühn
(Chairman of the project team SC3.PT9)

Agenda

1. Initial Situation
2. Revision Process and Stakeholders
3. Key Innovations
4. Outlook

EUROPEAN STANDARD
NORME EUROPÉENNE
EUROPÄISCHE NORM

EN 1993-1-10

May 2005



ICS 91.010.30

Supersedes ENV 1993-1-1:1992

English version

Eurocode 3: Design of steel structures - Part 1-10: Material toughness and through-thickness properties

Eurocode 3 - Calcul des structures en acier vis-à-vis de la ténacité et des propriétés dans le sens de l'épaisseur - Partie 1-10 : Choix des qualités d'acier

Eurocode 3: Bemessung und Konstruktion von Stahlbauten - Teil 1-10 :Stahlsortenauswahl im Hinblick auf Bruchzähigkeit und Eigenschaften in Dickenrichtung

This European Standard was approved by CEN on 23 April 2004.

CEN members are bound to comply with the CEN/CENELEC Internal Regulations which stipulate the conditions for giving this European Standard the status of a national standard without any alteration. Up-to-date lists and bibliographical references concerning such national standards may be obtained on application to the Central Secretariat or to any CEN member.

This European Standard exists in three official versions (English, French, German). A version in any other language made by translation under the responsibility of a CEN member into its own language and notified to the Central Secretariat has the same status as the official versions.

CEN members are the national standards bodies of Austria, Belgium, Cyprus, Czech Republic, Denmark, Estonia, Finland, France, Germany, Greece, Hungary, Iceland, Ireland, Italy, Latvia, Lithuania, Luxembourg, Malta, Netherlands, Norway, Poland, Portugal, Slovakia, Slovenia, Spain, Sweden, Switzerland and United Kingdom.

CEN/TC 250

Date: 2024-01

FprEN 1993-1-10:2024

Eurocode 3 — Design of steel structures — Part 1-10: Material toughness and through-thickness properties

Eurocode 3: Bemessung und Konstruktion von Stahlbauten — Teil 1-10: Stahlsortenauswahl im Hinblick auf Bruchzähigkeit und Eigenschaften in Dickenrichtung

Eurocode 3 : Calcul des structures en acier — Partie 1-10 : Choix des qualités d'acier vis-à-vis de la ténacité et des propriétés dans le sens de l'épaisseur

1. Initial Situation

EUROPEAN STANDARD

EN 1993-1-10

NORME EUROPÉENNE

EUROPÄISCHE NORM

May 2005

ICS 91.010.30

Supersedes ENV 1993-1-1:1992

English version

Eurocode 3: Design of steel structures - Part 1-10: Material toughness and through-thickness properties

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see also Sedlacek, G. et al.:

<https://publications.jrc.ec.europa.eu/repository/handle/JRC47278>

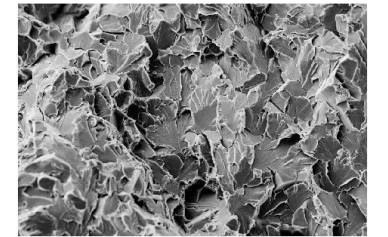
JRC Scientific and Technical Reports



COMMENTARY AND WORKED EXAMPLES to EN 1993-1-10 "Material toughness and through thickness properties" and other toughness oriented rules in EN 1993

G. Sedlacek, M. Feldmann, B. Kühn, D. Tschickardt, S. Höhler, C. Müller, W. Hensen, N. Stranghöner, W. Dahl, P. Langenberg, S. Münstermann, J. Brozetti, J. Raoul, R. Pope, F. Bijlaard

Background documents in support to the implementation, harmonization and further development of the Eurocodes



Joint Report
Prepared under the JRC – ECCS cooperation agreement for the evolution of Eurocode 3
(programme of CEN / TC 250)

Editors: M. Gérardin, A. Pinto and S. Dimova

First Edition, September 2008

EUR 23510 EN - 2008

1. Initial Situation 2. 3. 4.

- In Germany and other European countries valid since 2012
- Mostly positive experiences in the past 13 years all over Europe
- Further applications also outside Europe (e.g. similar design rules in Australia / New Zealand and 1st considerations in Canada)

| DEUTSCHE NORM | | Dezember 2010 |
|--|------------------------------|---------------|
| | DIN EN 1993-1-10 | DIN |
| ICS 91.010.30; 91.080.10 | Ersatzvermerk siehe unten | |
| Eurocode 3: Bemessung und Konstruktion von Stahlbauten – Teil 1-10: Stahlsortenauswahl im Hinblick auf Bruchzähigkeit und Eigenschaften in Dickenrichtung; Deutsche Fassung EN 1993-1-10:2005 + AC:2009 | | |
| Eurocode 3: Design of steel structures – Part 1-10: Material toughness and through-thickness properties; German version EN 1993-1-10:2005 + AC:2009 | | |
| Eurocode 3: Calcul des structures en acier – Partie 1-10: Choix des qualités d'acier vis à vis de la ténacité et des propriétés dans le sens de l'épaisseur; Version allemande EN 1993-1-10:2005 + AC:2009 | | |
| Ersatzvermerk | | |
| Ersatz für DIN EN 1993-1-10:2005-07; mit DIN EN 1993-1-1:2010-12, DIN EN 1993-1-1/NA:2010-12, DIN EN 1993-1-3:2010-12, DIN EN 1993-1-3/NA:2010-12, DIN EN 1993-1-5:2010-12, DIN EN 1993-1-5/NA:2010-12, DIN EN 1993-1-8:2010-12, DIN EN 1993-1-8/NA:2010-12, DIN EN 1993-1-9:2010-12, DIN EN 1993-1-9/NA:2010-12, DIN EN 1993-1-10/NA:2010-12, DIN EN 1993-1-11:2010-12 und DIN EN 1993-1-11/NA:2010-12 Ersatz für DIN 18800-1:2008-11; Ersatz für DIN EN 1993-1-10 Berichtigung 1:2010-05 | | |
| Gesamtumfang 22 Seiten | | |
| Normenausschuss Bauwesen (NABau) im DIN | | |

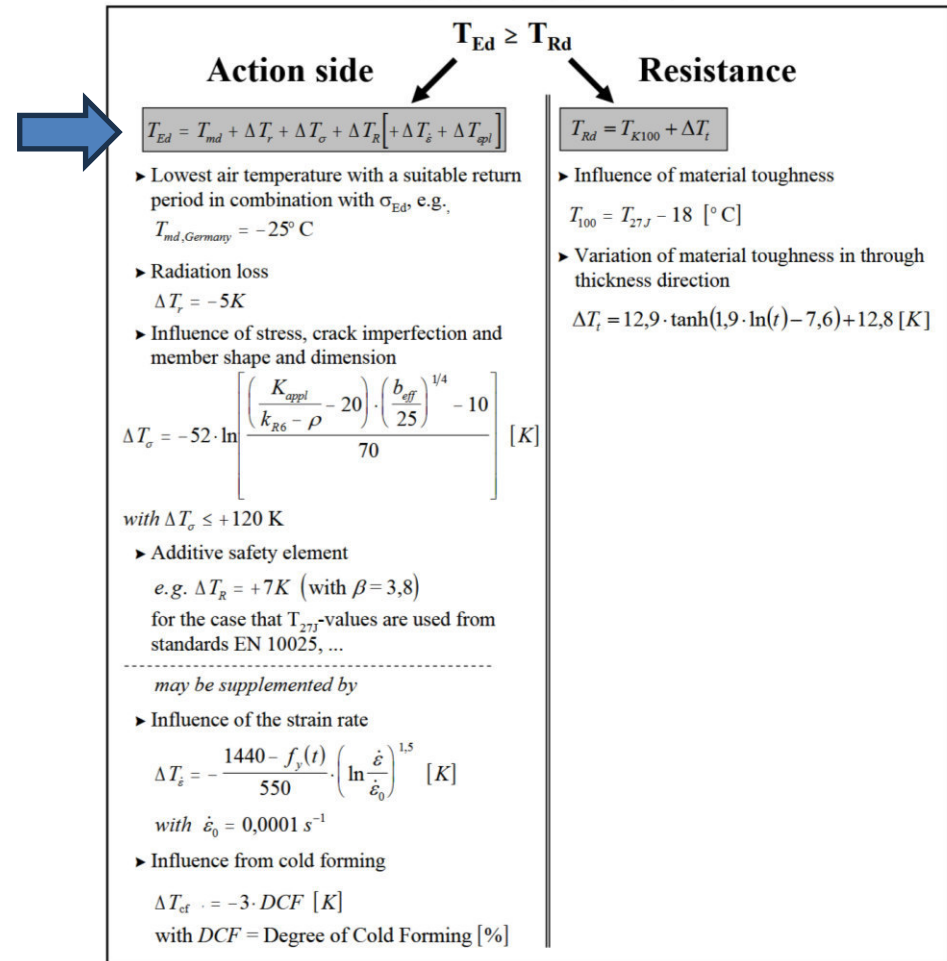
Fracture mechanics approach on temperature basis

$$K_{\text{appl,d}}^* \leq K_{\text{mat,d}} \rightarrow \text{Transformation} \rightarrow T_{\text{Ed}} \geq T_{\text{Rd}}$$

$$T_{\text{Ed}} = T_{\text{mdr}} + \Delta T_{\sigma} + \Delta T_{\text{R}} + \Delta T_{\dot{\epsilon}} + \Delta T_{\text{ecf}}$$

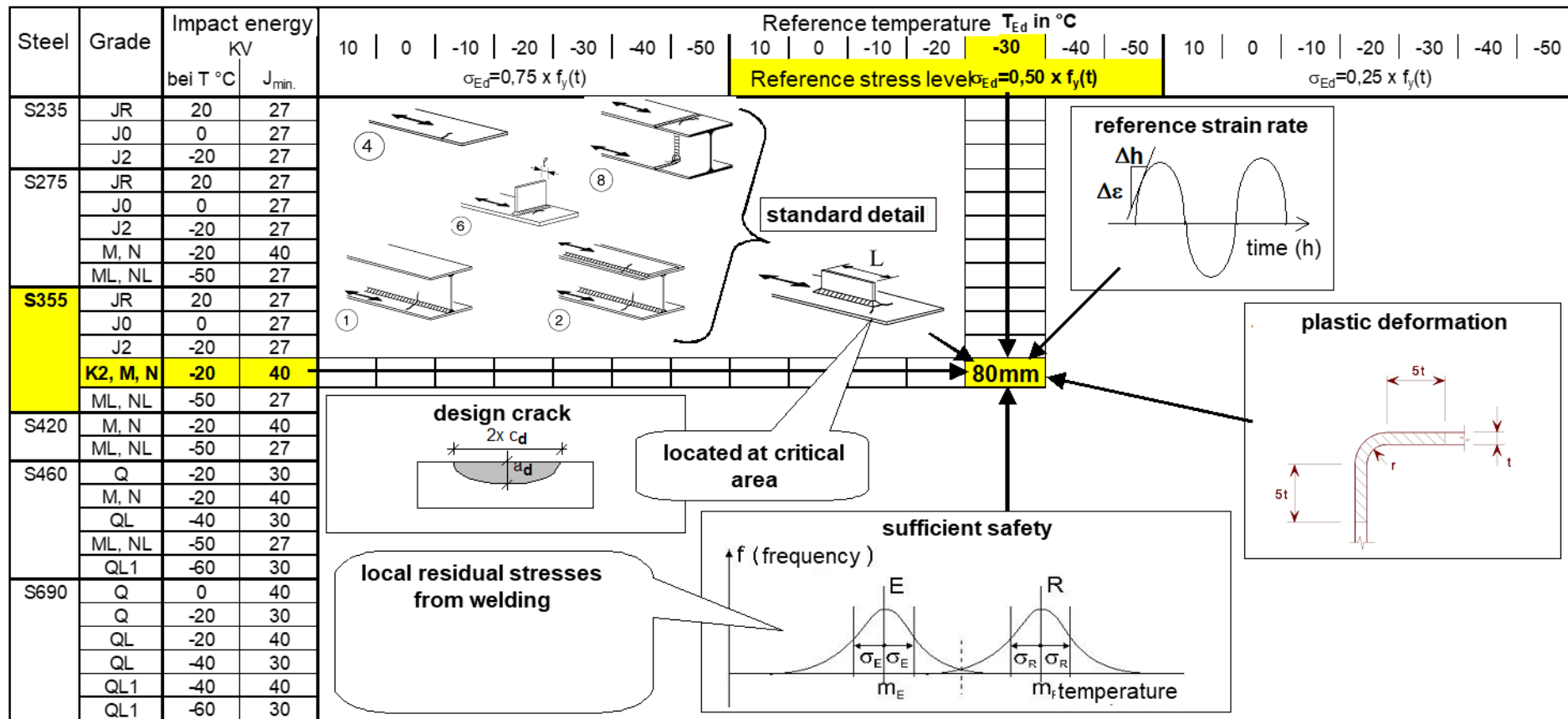
Reference temperature T_{Ed} also used as main input parameter for the simplified design approach using tabulated values.

Assessment scheme



1. Initial Situation 2. 3. 4.

Simplified design approach (in general)



2. Revision Process and Stakeholders



1. 2. Revision and Stakeholders 3. 4.

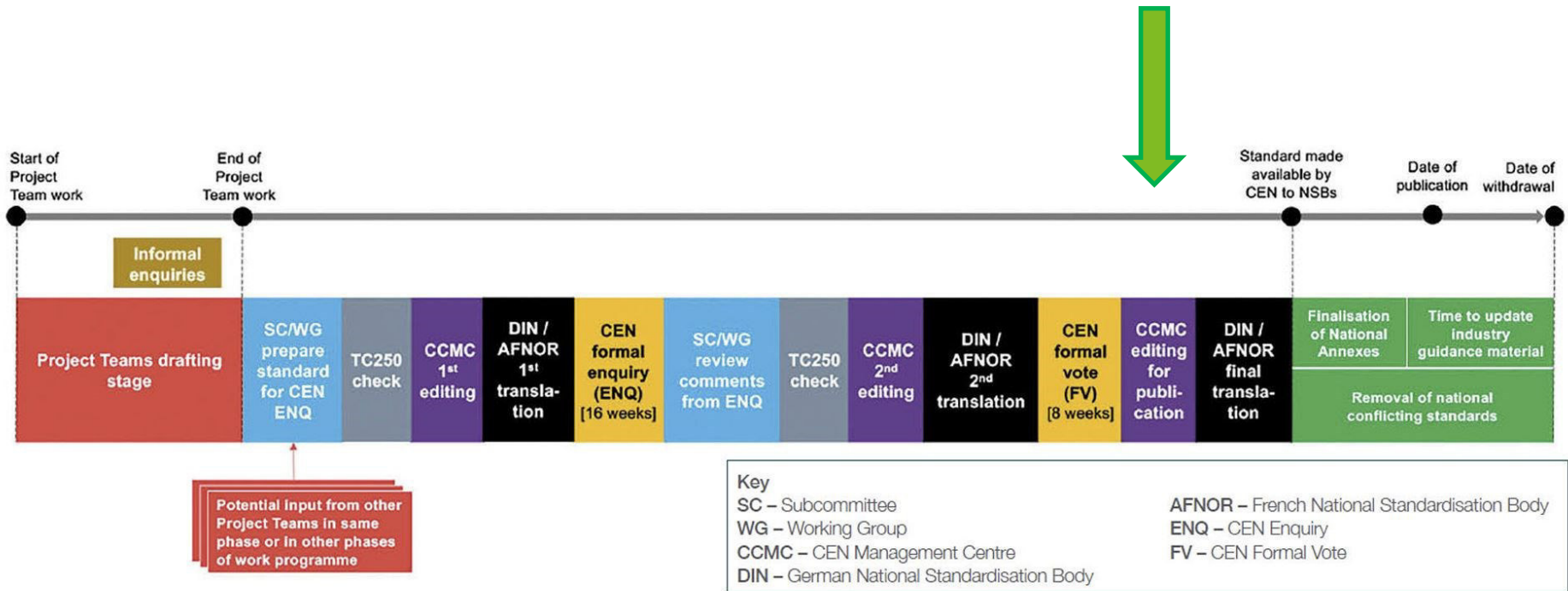
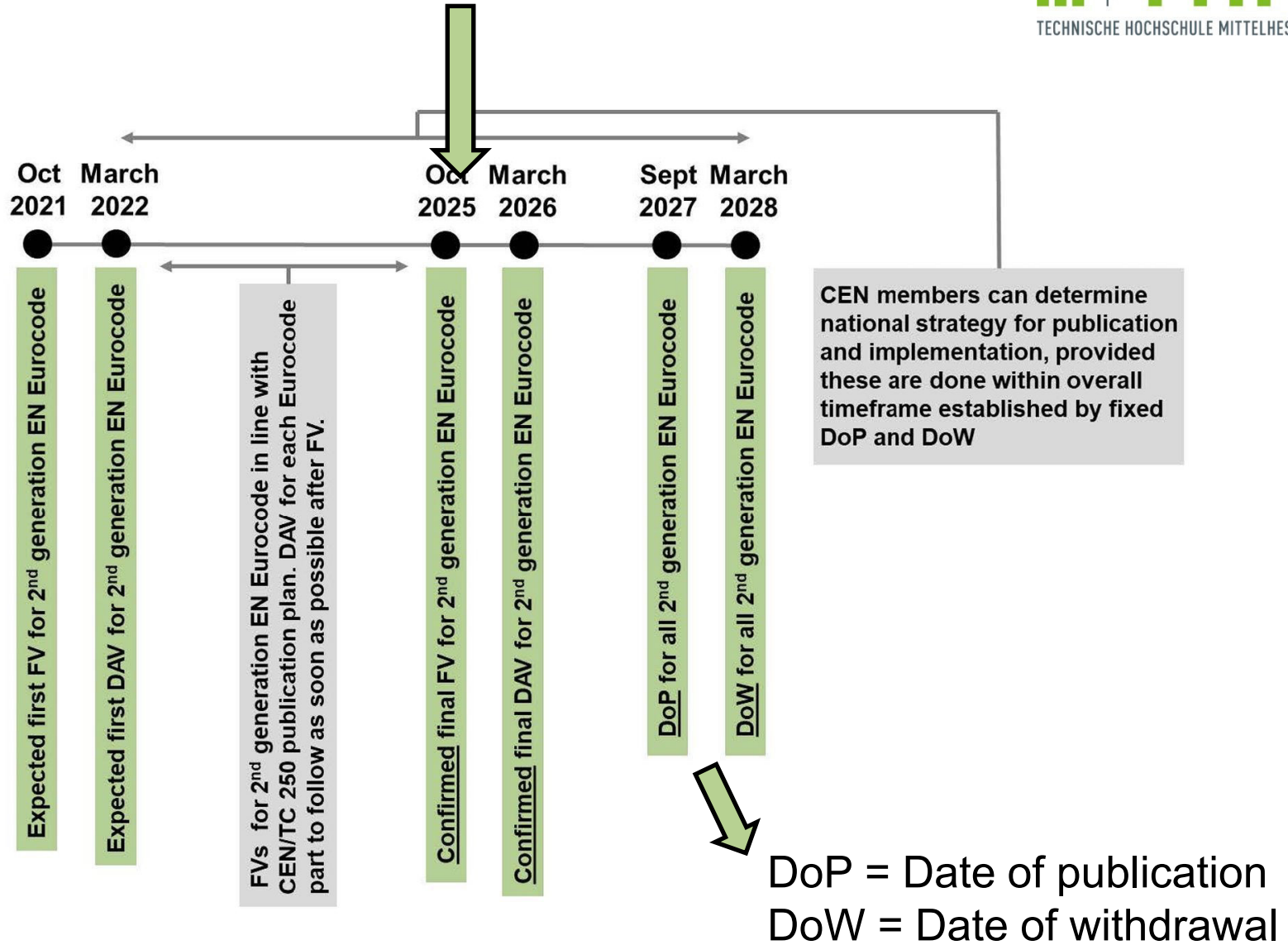


FIGURE 1 — Eurocode development process

1. 2. Revision and Stakeholders 3. 4.



Responsible for technical revisions ECCS / TC6 / WG10



1. 2. Revision and Stakeholders 3. 4.

Responsible for technical revisions ECCS / TC6 / WG10

| Role | Name | Send by |
|-----------|-------------------|-----------------|
| Convenor | Kühn, Bertram | CEN/TC 250/SC 3 |
| Secretary | Kempa, Susan | DIN (DE) |
| Expert | Abakanov, Mirken | KAZMEMST (KAZ) |
| Expert | Baudet, Philippe | AFNOR (FR) |
| Expert | Bespayev, Aliy | KAZMEMST (KAZ) |
| Expert | Borges, Luís | IPQ (PT) |
| Secretary | Braner, Jessica | DIN (DE) |
| Monitor | Burgos, Dag | SN (NO) |
| Expert | Candeias, Miguel | ILNAS (LUX) |
| Expert | Chirea, Cristina | ASRO (RO) |
| Secretary | Clunie, Sandra | DIN (DE) |
| Expert | Davaine, Laurence | AFNOR (FR) |
| Expert | Dest, Asky | BSI (GB) |
| Expert | Dubina, Dan | ASRO (RO) |
| Expert | Engelhardt, Imke | DIN (DE) |
| Expert | Feldmann, Markus | DIN (DE) |

1. 2. Revision and Stakeholders 3. 4.

Responsible for technical revisions ECCS / TC6 / WG10

| Role | Name | Send by |
|---------|----------------------------|----------------|
| Expert | Imam, Boulent | BSI (GB) |
| Expert | Kortelainen, Pasi | SFS (FI) |
| Expert | Krabbe-Christensen, Peter | DS (DK) |
| Expert | Kuhlmann, Ulrike | DIN (DE) |
| Expert | Lochte-Holtgreven, Stephan | DIN (DE) |
| Expert | Lukic, Mladen | AFNOR (FR) |
| Expert | Ogle, Martin | BSI (GB) |
| Monitor | Operations Support Centre | BSI (GB) |
| Expert | Pope, David | BSI (GB) |
| Expert | Sandon, Stefano | UNI (IT) |
| Expert | Shahnovich, Alexandr | KAZMEMST (KAZ) |
| Monitor | Skajaa, Birgitte | SN (NO) |
| Expert | Skejic, Davor | HZN (HR) |
| Expert | Sleczka, Lucjan | PKN (PL) |
| Expert | Solland, Gunnar | SN (NO) |

1. 2. Revision and Stakeholders 3. 4.

Responsible for technical revisions ECCS / TC6 / WG10

| Role | Name | Send by |
|--------|----------------------|----------------|
| Expert | Starr, Christopher | BSI (GB) |
| Expert | Stranghöner, Natalie | DIN (DE) |
| Expert | Tuleyev, Tursymbay | KAZMEMST (KAZ) |
| Expert | Wallin, Kim | SFS (FI) |
| Expert | Walter, Salvatore | UNI (IT) |
| Expert | Walters, Carey | NEN (NL) |

The group consists of:

1 chairperson

3 secretaries

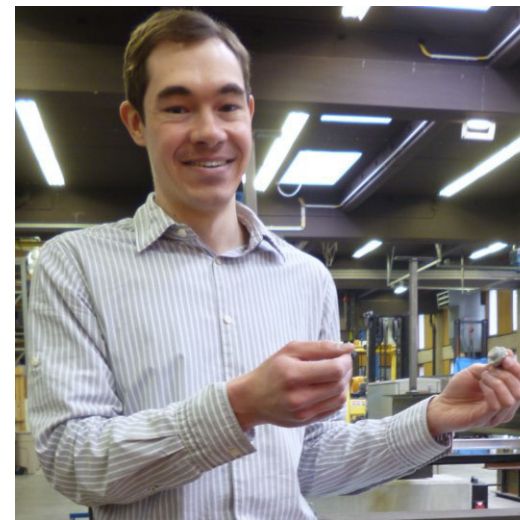
3 observers

30 experts

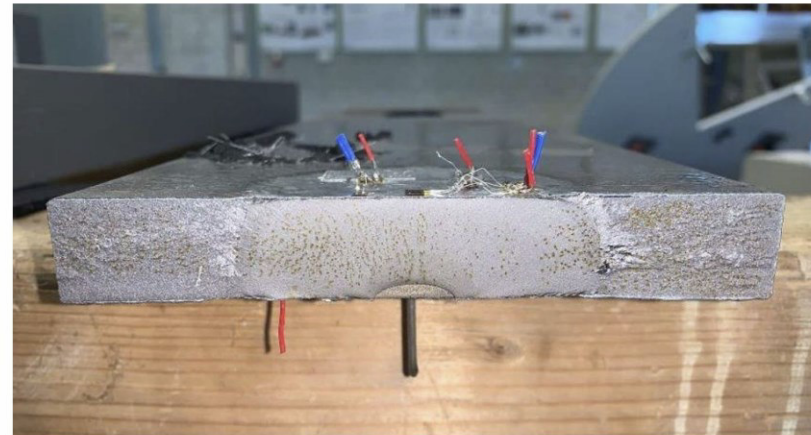
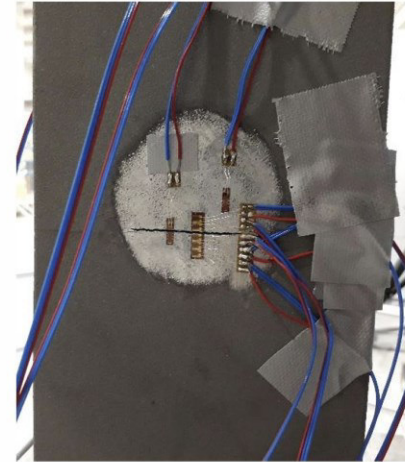
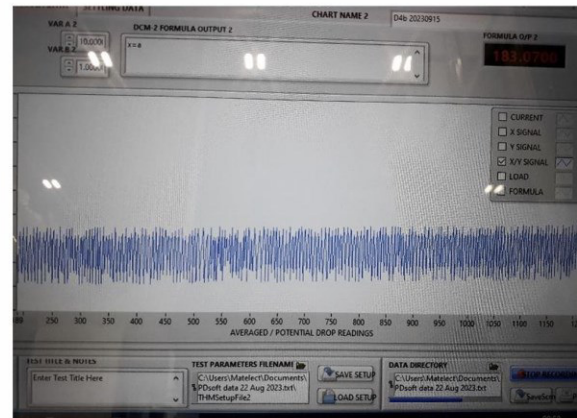
37 members from almost all over Europe

Responsible for text revisions TC250 / SC3 / PT9

| Position | Name | First name | Init. | Country |
|-----------|----------|------------|-------|---------|
| PT leader | Kühn | Bertram | Kue | DE |
| PT member | Cosgrove | Thomas | Cos | UK |
| PT member | Solland | Gunnar | GSol | NO |
| PT member | Borges | Luis | Bor | CH |
| PT member | Tibolt | Mike | Tib | LUX |



3. Key Innovations



1. 2. 3. Key Innovations 4.

Well known simplified design method remains

| Steel grade | Quality | KV | | Reference Temperature T_{Ed} [°C] | | | | | | | | | | | | | | | | | | | | | | | | | | | |
|-------------|----------|--------|-----------|-------------------------------------|-----|-----|-----|-----|-----|-----|-----|------|-----|-----|-----|---------------------------|-----|-----|-----|-----|------|-----|-----|----------------------------|-----|-----|-----|-----|-----|------|--|
| | | T [°C] | J_{min} | 10 | 0 | -10 | -20 | -30 | -40 | -50 | -80 | -120 | 10 | 0 | -10 | -20 | -30 | -40 | -50 | -80 | -120 | 10 | 0 | -10 | -20 | -30 | -40 | -50 | -80 | -120 | |
| | | | | $\sigma_{Ed} = 0,75f_y(t)$ | | | | | | | | | | | | $\sigma_{Ed} = 0,5f_y(t)$ | | | | | | | | $\sigma_{Ed} = 0,25f_y(t)$ | | | | | | | |
| S235 | JR | 20 | 27 | 60 | 50 | 40 | 35 | 30 | 25 | 20 | 10 | 5 | 90 | 75 | 65 | 55 | 45 | 40 | 35 | 20 | 15 | 135 | 115 | 100 | 85 | 75 | 65 | 60 | 40 | 30 | |
| | J0 | 0 | 27 | 90 | 75 | 60 | 50 | 40 | 35 | 30 | 15 | 10 | 125 | 105 | 90 | 75 | 65 | 55 | 45 | 30 | 15 | 175 | 155 | 135 | 115 | 100 | 85 | 75 | 50 | 35 | |
| | J2 | -20 | 27 | 125 | 105 | 90 | 75 | 60 | 50 | 40 | 25 | 10 | 170 | 145 | 125 | 105 | 90 | 75 | 65 | 40 | 20 | 200 | 200 | 175 | 155 | 135 | 115 | 100 | 65 | 40 | |
| S275 | JR | 20 | 27 | 55 | 45 | 35 | 30 | 25 | 20 | 15 | 10 | 5 | 80 | 70 | 55 | 50 | 40 | 35 | 30 | 20 | 10 | 125 | 110 | 95 | 80 | 70 | 60 | 55 | 40 | 25 | |
| | J0 | 0 | 27 | 75 | 65 | 55 | 45 | 35 | 30 | 25 | 15 | 5 | 115 | 95 | 80 | 70 | 55 | 50 | 40 | 25 | 15 | 165 | 145 | 125 | 110 | 95 | 80 | 70 | 45 | 30 | |
| | J2 | -20 | 27 | 110 | 95 | 75 | 65 | 55 | 45 | 35 | 20 | 10 | 155 | 130 | 115 | 95 | 80 | 70 | 55 | 35 | 20 | 200 | 190 | 165 | 145 | 125 | 110 | 95 | 60 | 40 | |
| | K2,M,N | -20 | 40 | 135 | 110 | 95 | 75 | 65 | 55 | 45 | 25 | 10 | 180 | 155 | 130 | 115 | 95 | 80 | 70 | 40 | 20 | 200 | 200 | 190 | 165 | 145 | 125 | 110 | 70 | 40 | |
| | ML,NL | -50 | 27 | 185 | 160 | 135 | 110 | 95 | 75 | 65 | 35 | 15 | 200 | 200 | 180 | 155 | 130 | 115 | 95 | 55 | 30 | 200 | 200 | 200 | 200 | 190 | 165 | 145 | 95 | 55 | |
| S355 | JR | 20 | 27 | 40 | 35 | 25 | 20 | 15 | 15 | 10 | 5 | 5 | 65 | 55 | 45 | 40 | 30 | 25 | 25 | 15 | 10 | 110 | 95 | 80 | 70 | 60 | 55 | 45 | 30 | 20 | |
| | J0 | 0 | 27 | 60 | 50 | 40 | 35 | 25 | 20 | 15 | 10 | 5 | 95 | 80 | 65 | 55 | 45 | 40 | 30 | 20 | 10 | 150 | 130 | 110 | 95 | 80 | 70 | 60 | 40 | 25 | |
| | J2 | -20 | 27 | 90 | 75 | 60 | 50 | 40 | 35 | 25 | 15 | 5 | 135 | 110 | 95 | 80 | 65 | 55 | 45 | 25 | 15 | 200 | 175 | 150 | 130 | 110 | 95 | 80 | 55 | 30 | |
| | J4 | -40 | 27 | 130 | 110 | 90 | 75 | 60 | 50 | 40 | 20 | 10 | 180 | 155 | 135 | 110 | 95 | 80 | 65 | 40 | 20 | 200 | 200 | 195 | 170 | 150 | 130 | 110 | 70 | 40 | |
| | K2,M,N | -20 | 40 | 110 | 90 | 75 | 60 | 50 | 40 | 35 | 20 | 5 | 155 | 135 | 110 | 95 | 80 | 65 | 55 | 30 | 15 | 200 | 200 | 175 | 150 | 130 | 110 | 95 | 60 | 35 | |
| | J5,ML,NL | -50 | 27 | 155 | 130 | 110 | 90 | 75 | 60 | 50 | 25 | 10 | 200 | 180 | 155 | 135 | 110 | 95 | 80 | 45 | 25 | 210 | 200 | 200 | 200 | 175 | 150 | 130 | 80 | 45 | |

Only table 2.1 will become table 4.1!

1. 2. 3. Key Innovations 4.

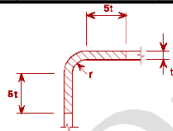
Harmonizations...

...with part 1-1

... with part 1-8

Table 6.3 — Conditions for welding cold-formed corners and adjacent material

| r/t | Strain due to cold forming (%) | Maximum thickness (mm) | | | |
|-------|--------------------------------|------------------------------|----------------------------|---|--|
| | | Generally | | Fully killed Aluminium-killed steel (Al ≥ 0,02 %) | |
| | | Predominantly static loading | Where fatigue predominates | | |
| ≥ 25 | ≤ 2 | any | any | any | |
| ≥ 10 | ≤ 5 | any | 16 | any | |
| ≥ 3,0 | ≤ 14 | 24 | 12 | 24 | |
| ≥ 2,0 | ≤ 20 | 12 | 10 | 12 | |
| ≥ 1,5 | ≤ 25 | 8 | 8 | 10 | |
| ≥ 1,0 | ≤ 33 | 4 | 4 | 6 | |



... with part 1-9

Von: Prof. Dr.-Ing. Bertram Kühn [mailto:bertram.kuehn@bau.thm.de]

Gesendet: Donnerstag, 4. April 2019 09:56

An: 'Kuhlmann, Ulrike' <ulrike.kuhlmann@ke.uni-stuttgart.de>

Cc: 'Bert Snijder' <hhsnijder@planet.nl>; 'Alain BUREAU' <ABUREAU@CTICM.com>; Luis Borges <lbo@structura.com>; Tom Cosgrove <tom.cosgrove@steelconstruction.org>

Betreff: AW: Comment on prEN1993-1-1, final draft

Dear Ulrike,
Dear Prof. Snijder,
Dear Mr. Bureau,

please excuse but we answer too fast yesterday. We do not recognized that also the clause (3) itself gives a val prEN1993-1-1:

Text as it is:

(3) For components under compression, a minimum toughness property should be selected using the appr NOTE For selection of toughness properties for members in compression, the values of maximum permissible elemen a country.

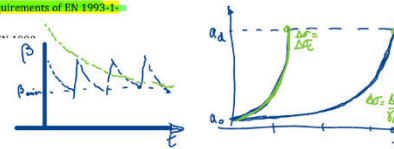
1.2 Scope of EN 1993-1-9

(1) EN 1993-1-9 gives design methods for the verification of fatigue resistance of steel structures.

NOTE Steel structures consist of members and their joints. Each member and joint can be represented as a structural detail or as several of the latter.

(2) These methods are derived from fatigue tests on constructional details with large scale specimens that include effects of geometrical and structural imperfections from material production and execution (e.g. the effects of tolerances and residual stresses from welding).

(3) This part only applies to materials which conform to the toughness requirements of EN 1993-1-10.



b) damage tolerant method

provision of readily inspectable details during regular in-service inspections, selecting details, materials and stress levels resulting in a fatigue life sufficient to achieve the acceptable level of reliability given in EN 1990 or level equal to that required for ultimate limit state verifications at the end of each inspection interval as required in EN 1993-1-10.

- so that in the event of the formation of cracks, low propagation rates and easily detectable cracks without failure would result,
- of multiple load paths are provided for,
- or crack-arresting details are provided for.

(5) For the purpose of verification of fatigue design situation using this part, an acceptable level of reliability may be achieved by adjustment of the partial factor for fatigue strength γ_{fat} taking into account the consequences of failure and the detection method used.

Handwritten notes: 1.2.10, 1.2.10, 1.2.10

Handwritten notes: ELU ok at the end of each in service inspection interval as required in EN 1993-1-10. dette cause it is confusing. It applies to both methods. The reference is already made at the beginning.

EN 1993-1-10
[30] = [27] + U [U]

1. 2. 3. Key Innovations 4.

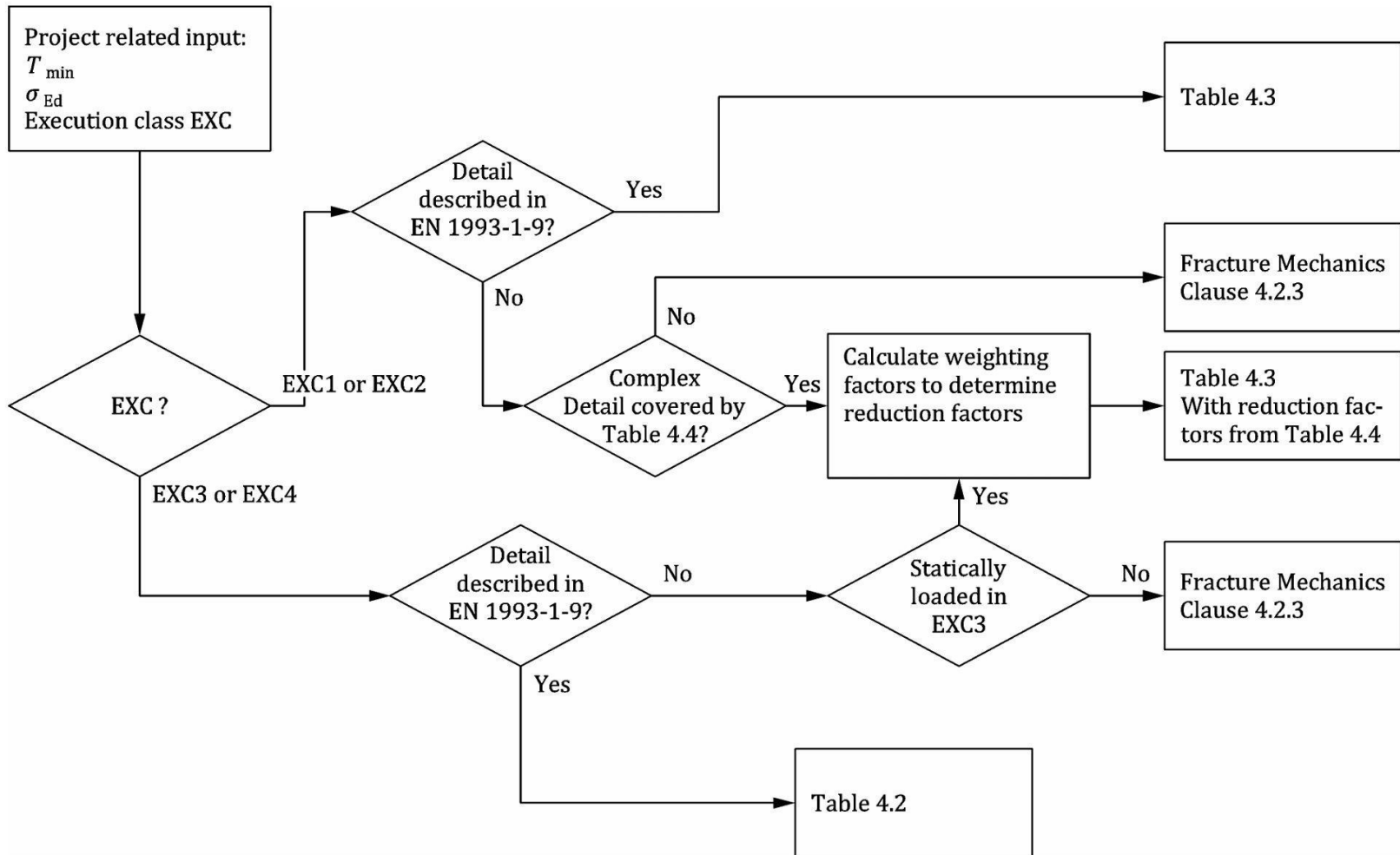
1. New rules for reused steel components

1.1 Scope of EN 1993-1-10 extended as follows:

(6) EN 1993-1-10 specifies rules that apply to steel materials covered by EN 1993-1-1:2022, 5.1(3), provided that each individual piece of steel is tested in accordance with the requirements of EN 1993-1-1:2022, 5.2.1 and EN 1090-2:2018, 5.1.

(7) This document does not apply to material salvaged from existing steelwork subjected to fatigue or fire.

2. Increasing Ease-of-use by adding a new flowchart



1. 2. 3. Key Innovations 4.

3. Extension to high-strength and new steel grades + cold formed hollow sections

| AM-1-12-2015-01 | |
|----------------------------------|---|
| Subject | Selection of materials for fracture toughness, determination of maximum permissible values of element thickness |
| Clause No./ Subclause No./ Annex | EN 1993-1-10, 2.3.2 (1) z.B. - EN 10149-2:2013 (S315MC – S960MC) - EN 10149-3:2013 (S260NC – S420NC) |
| Reason for amendment | It has been decided by SC3 that EN 1993-1-12 shall not be a separate document. Its content shall be included in the relevant parts of Eurocode 3. |



| Steel grade | Sub-grade | Charpy energy CVN | | Reference temperature T _{Ed} [°C] | | | | | | | | |
|-------------|-----------|---|------------------|--|-----|-----|-----|-----|-----|-----|-----|------|
| | | at T [°C] | J _{min} | 10 | 0 | -10 | -20 | -30 | -40 | -50 | -80 | -120 |
| | | σ _{Ed} = 0,75·f _y (t) | | | | | | | | | | |
| S235 | JR | +20 | 27 | 62 | 51 | 43 | 36 | 30 | 25 | 21 | 14 | 9 |
| | J0 | 0 | 27 | 89 | 74 | 62 | 51 | 43 | 36 | 30 | 18 | 11 |
| | J2 | -20 | 27 | 126 | 106 | 89 | 74 | 62 | 51 | 43 | 25 | 14 |
| S275 | JR | +20 | 27 | 54 | 45 | 37 | 31 | 26 | 22 | 18 | 11 | 7 |
| | J0 | 0 | 27 | 78 | 65 | 54 | 45 | 37 | 31 | 26 | 15 | 9 |
| | J2 | -20 | 27 | 112 | 94 | 78 | 65 | 54 | 45 | 37 | 22 | 11 |
| | M, N | -20 | 40 | 132 | 111 | 93 | 77 | 64 | 53 | 44 | 25 | 13 |
| | ML, NL | -50 | 27 | 185 | 158 | 134 | 112 | 94 | 78 | 65 | 37 | 18 |

1. 2. 3. Key Innovations 4.

4. Adding the option to consider consequence (execution) classes (EXC)

Table 4.2: Maximum permissible values of element thickness t in mm for Execution Class EXC3 and EXC4

| Steel grade | Quality | KV | | Reference Temperature T_{Ed} [°C] | | | | | | | | | | | | | | | | | | | |
|-------------|---------|-----------------------------------|-----------|-------------------------------------|-----|-----|-----|-----|-----|-----|----------------------------------|------|-----|-----|-----|-----|-----|-----|-----|-----|------|-----|---|
| | | T [°C] | J_{min} | 10 | 0 | -10 | -20 | -30 | -40 | -50 | -80 | -120 | 10 | 0 | -10 | -20 | -30 | -40 | -50 | -80 | -120 | 10 | |
| | | $\sigma_{Ed} = 0.75 \cdot f_y(t)$ | | | | | | | | | $\sigma_{Ed} = 0.5 \cdot f_y(t)$ | | | | | | | | | | | | |
| S235 | JR | 20 | 27 | 60 | 50 | 40 | 35 | 30 | 25 | 20 | 10 | 5 | 90 | 75 | 65 | 55 | 45 | 40 | 35 | 20 | 15 | 135 | 1 |
| | JO | 0 | 27 | 90 | 75 | 60 | 50 | 40 | 35 | 30 | 15 | 10 | 125 | 105 | 90 | 75 | 65 | 55 | 45 | 30 | 15 | 175 | 1 |
| | J2 | -20 | 27 | 125 | 105 | 90 | 75 | 60 | 50 | 40 | 25 | 10 | 170 | 145 | 125 | 105 | 90 | 75 | 65 | 40 | 20 | 200 | 2 |
| S275 | JR | 20 | 27 | 55 | 45 | 35 | 30 | 25 | 20 | 15 | 10 | 5 | 80 | 70 | 55 | 50 | 40 | 35 | 30 | 20 | 10 | 125 | 1 |

Table 4.3: Maximum permissible values of element thickness t in mm for EXC1 and EXC2

| Steel grade | Quality | KV | | Reference Temperature T_{Ed} [°C] | | | | | | | | | | | | | | | | | | | | | |
|-------------|---------|-----------------------------------|-----------|-------------------------------------|-----|-----|-----|-----|-----|-----|-----|----------------------------------|-----|-----|-----|-----|-----|-----|-----|-----|------|-----|-----|-----|---|
| | | T [°C] | J_{min} | 10 | 0 | -10 | -20 | -30 | -40 | -50 | -80 | -120 | 10 | 0 | -10 | -20 | -30 | -40 | -50 | -80 | -120 | 10 | 0 | -10 | |
| | | $\sigma_{Ed} = 0.75 \cdot f_y(t)$ | | | | | | | | | | $\sigma_{Ed} = 0.5 \cdot f_y(t)$ | | | | | | | | | | | | | |
| S235 | JR | 20 | 27 | 250 | 250 | 170 | 120 | 90 | 65 | 50 | 25 | 15 | 250 | 250 | 250 | 250 | 190 | 145 | 110 | 55 | 30 | 250 | 250 | 250 | 2 |
| | JO | 0 | 27 | 250 | 250 | 250 | 250 | 170 | 120 | 90 | 40 | 15 | 250 | 250 | 250 | 250 | 250 | 250 | 190 | 85 | 40 | 250 | 250 | 250 | 2 |
| | J2 | -20 | 27 | 250 | 250 | 250 | 250 | 250 | 250 | 170 | 65 | 25 | 250 | 250 | 250 | 250 | 250 | 250 | 250 | 145 | 55 | 250 | 250 | 250 | 2 |
| S275 | JR | 20 | 27 | 250 | 185 | 130 | 95 | 70 | 50 | 40 | 20 | 10 | 250 | 250 | 250 | 205 | 150 | 115 | 90 | 45 | 25 | 250 | 250 | 250 | 2 |
| | JO | 0 | 27 | 250 | 250 | 250 | 185 | 130 | 95 | 70 | 30 | 15 | 250 | 250 | 250 | 250 | 250 | 205 | 150 | 70 | 30 | 250 | 250 | 250 | 2 |
| | J2 | -20 | 27 | 250 | 250 | 250 | 250 | 250 | 185 | 130 | 50 | 20 | 250 | 250 | 250 | 250 | 250 | 250 | 250 | 115 | 45 | 250 | 250 | 250 | 2 |

1. 2. 3. Key Innovations 4.

5. Option for predominantly statically loaded components of EXC 1 and EXC 2 to take into account the complexity of connections

Table 4.4 — Selection of weighting values for detail complexity

| | Influencing parameter | Level of influencing parameters | W-value |
|----|---|---|---------------|
| a) | Degree of triaxiality due to nominal stresses | Uniaxial (Reference, see e.g. Figure 4.2) | $W_{ca} = 0$ |
| | | Biaxial | $W_{ca} = 2$ |
| | | Multiaxial, see e.g. Figure 4.3 | $W_{ca} = 5$ |
| b) | Degree of stress concentration | Medium (Reference, see e.g. Figure 4.2) | $W_{cb} = 0$ |
| | | High | $W_{cb} = 2$ |
| c) | Accessibility of NDT during fabrication | Normal (Reference, see e.g. Figure 4.2) | $W_{cc} = 0$ |
| | | Difficult, see e.g. Figure 4.3 | $W_{cc} = 1$ |
| d) | Fabrication of holes | No holes (Reference, see e.g. Figure 4.2) | $W_{cd} = 0$ |
| | | Drilled | $W_{cd} = 0$ |
| | | Punched (untreated) | $W_{cd} = 1$ |
| e) | Level of residual stresses σ_{res} | $\sigma_{res} < 0,5 f_y$ (after stress relieving due to heat treatment) | $W_{ce} = -1$ |
| | | $0,75 f_y < \sigma_{res} < 1,0 f_y$ (Reference, see e.g. Figure 4.2) | $W_{ce} = 0$ |

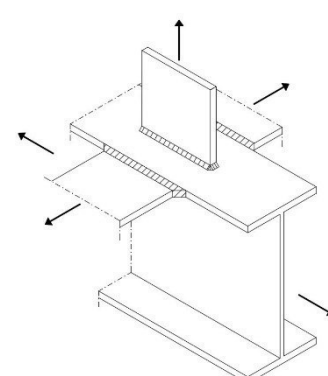
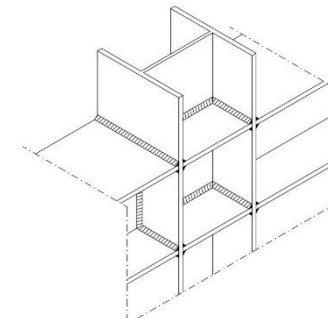


Table 4.5 — Reduction factors

| Weighting factors | $-1 \leq W_c \leq 2$ | $3 \leq W_c \leq 4$ | $5 \leq W_c \leq 6$ | $7 \leq W_c \leq 9$ |
|-------------------|----------------------|---------------------|---------------------|---------------------|
| Reduction Factors | 1,0 | 0,75 | 0,50 | 0,35 |

1. 2. 3. Key Innovations 4.

6. Additional rules for spaded details given in a new informative annex A

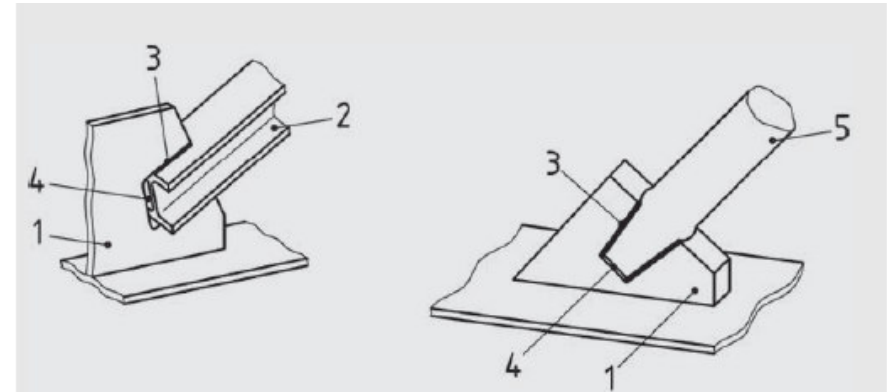


Table A.4 — Maximum permissible values of the width of the gusset plate on one side of the gap w_i^* [mm] for usual geometric parameters of spaded details according to Table A.1 and a thickness of the gusset plate $t \leq 40$ mm

| $L/w_i^* \geq 1,3$ $H/2w_i^* \leq 0,55$ $t \leq 40$ mm | | | | | | | | | | | | | | | | | | | | | | | | |
|--|---------|-----------------------------------|-----------|-------------------------------------|-----|-----|-----|----------------------------------|-----|-----|-----|-----|-----|-----------------------------------|-----|-----|-----|-----|-----|-----|-----|-----|-----|-----|
| Steel grade | Quality | KV | | Reference Temperature T_{Ed} [°C] | | | | | | | | | | | | | | | | | | | | |
| | | T [°C] | J_{min} | 10 | 0 | -10 | -20 | -30 | -40 | -50 | 10 | 0 | -10 | -20 | -30 | -40 | -50 | 10 | 0 | -10 | -20 | -30 | -40 | -50 |
| | | $\sigma_{Ed} = 0,75 \cdot f_y(t)$ | | | | | | $\sigma_{Ed} = 0,5 \cdot f_y(t)$ | | | | | | $\sigma_{Ed} = 0,25 \cdot f_y(t)$ | | | | | | | | | | |
| S235 | JR | 20 | 27 | 20 | - | - | - | - | - | - | 60 | 40 | 30 | 20 | - | - | - | 190 | 120 | 80 | 60 | 40 | 30 | 30 |
| | J0 | -0 | 27 | 50 | 30 | 20 | - | - | - | - | 150 | 90 | 60 | 40 | 30 | 20 | - | all | 280 | 190 | 120 | 80 | 60 | 40 |
| | J2 | -20 | 27 | 150 | 90 | 50 | 30 | 20 | - | - | 380 | 240 | 150 | 90 | 60 | 40 | 30 | all | all | all | 280 | 190 | 120 | 80 |
| S275 | JR | 20 | 27 | - | - | - | - | - | - | - | 40 | 30 | 20 | - | - | - | - | 150 | 100 | 70 | 50 | 30 | 30 | 20 |
| | J0 | -0 | 27 | 40 | 20 | - | - | - | - | - | 100 | 70 | 40 | 30 | 20 | - | - | 360 | 230 | 150 | 100 | 70 | 50 | 30 |
| | J2 | -20 | 27 | 110 | 60 | 40 | 20 | - | - | - | 290 | 180 | 110 | 70 | 40 | 30 | 20 | all | all | 360 | 230 | 150 | 100 | 70 |
| | M, N | -20 | 40 | 180 | 110 | 60 | 40 | 20 | - | - | all | 290 | 180 | 110 | 70 | 40 | 30 | all | all | all | 360 | 230 | 150 | 100 |
| | ML, NL | -50 | 27 | all | 300 | 180 | 110 | 60 | 40 | 20 | all | all | all | 290 | 180 | 110 | 70 | all | all | all | all | all | 360 | 230 |

7. Incorporation of new rules to ensure upper-shelf behaviour where needed

4.3 Materials with additional fracture toughness requirements in relation to upper shelf

(1) In addition to the provisions given in EN 1993-1-1:2022, 5.2 and in 4.2 of this document, for material thickness $t > 30\text{mm}$ for steel grade S275, S355 and for S235 where specified below, the following requirement should apply:

- For EXC1 the minimum toughness requirements should be JR.
- For EXC2 and for (quasi-)statically loaded steelwork in EXC3, the minimum toughness requirements should be J0.
- For fatigue and seismic loaded steelwork in EXC3, the minimum toughness requirements should be J2 for steel grade S235, S275 and S355.
- For EXC4, fine grain steels should be used.

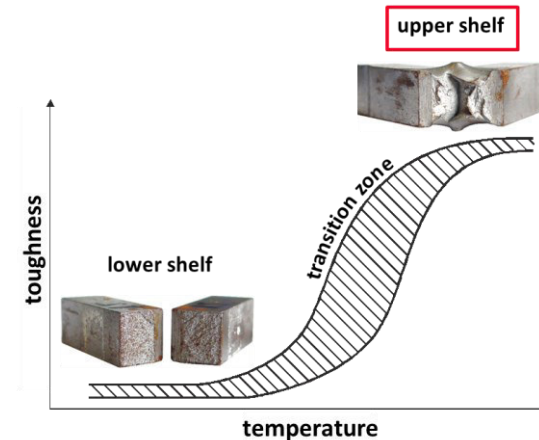
(2) The design against brittle fracture according to 4.2 could lead to higher minimum toughness requirements than mentioned in (1). In such cases the higher of both toughness requirements should be used.

(3) Where a further test is required, a minimum energy KV_{us} in a Charpy-V-notch impact test at room temperature should not be less than 100 J for static and fatigue design and 125 J for seismic design where details not covered by Annex E of prEN 1998-1-2:2023 are used. For fine grain steels, no test of KV_{us} is necessary.

NOTE 1 The National Annex can determine the scope of application, provide alternative requirements or alternative procedures for 4.3. This includes the conditions where the further test in paragraph (3) is required.

NOTE 2 Fine grain steels are steels with a ferritic grain size equivalent index of ≥ 6 , see EN ISO 643. Fine grain steels can be either ordered according to the corresponding product standards or fine grain structure size according to EN ISO 643 can be guaranteed by the producer upon request of the purchaser.

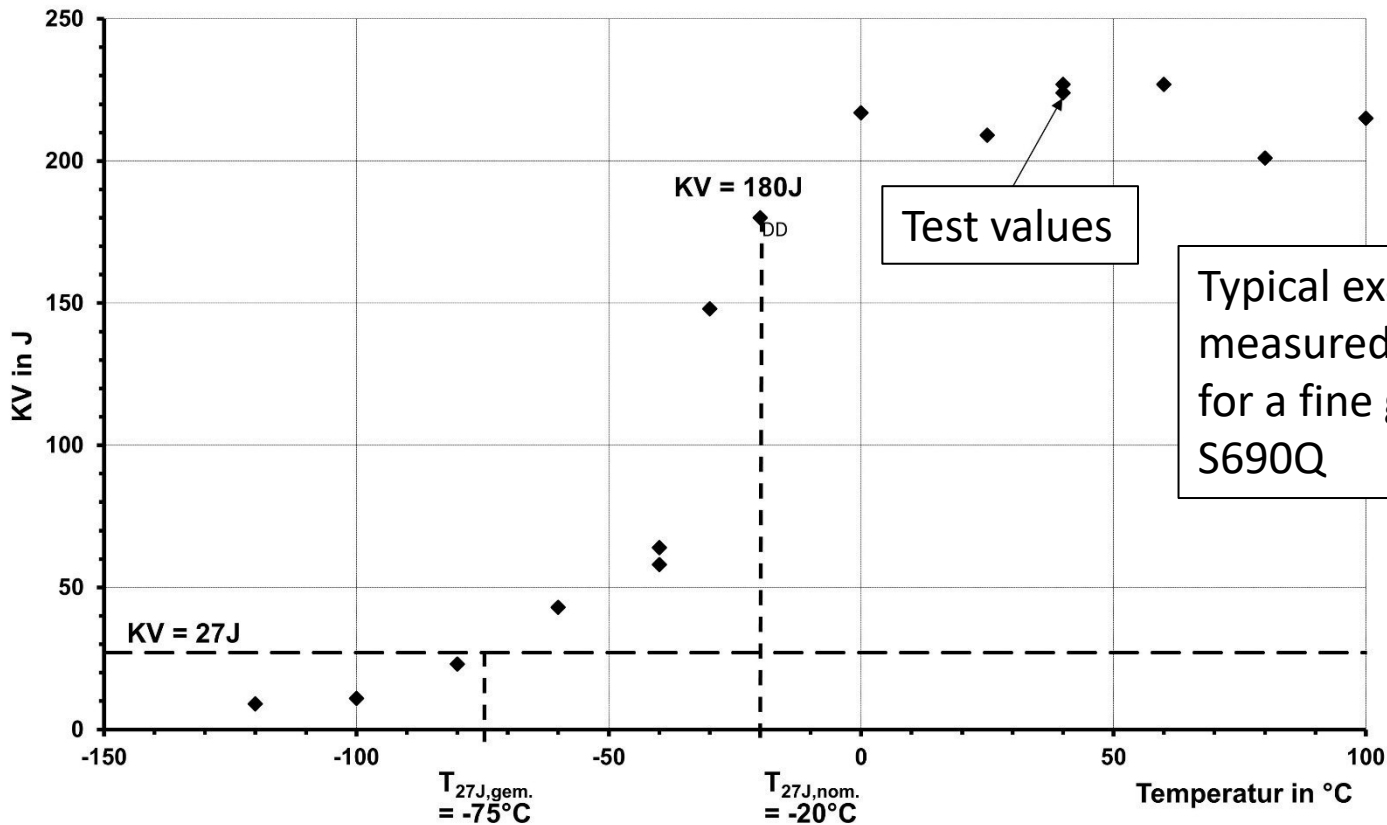
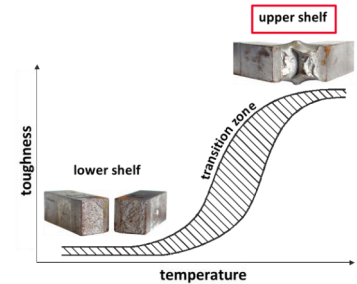
(4) For elements not subject to tension stresses, the provisions of 4.3 do not apply.



1. 2. 3. Key Innovations 4.

7. Incorporation of new upper-shelf rules

Need of such rules

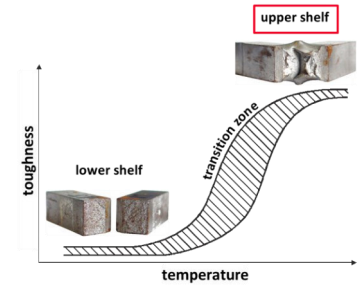
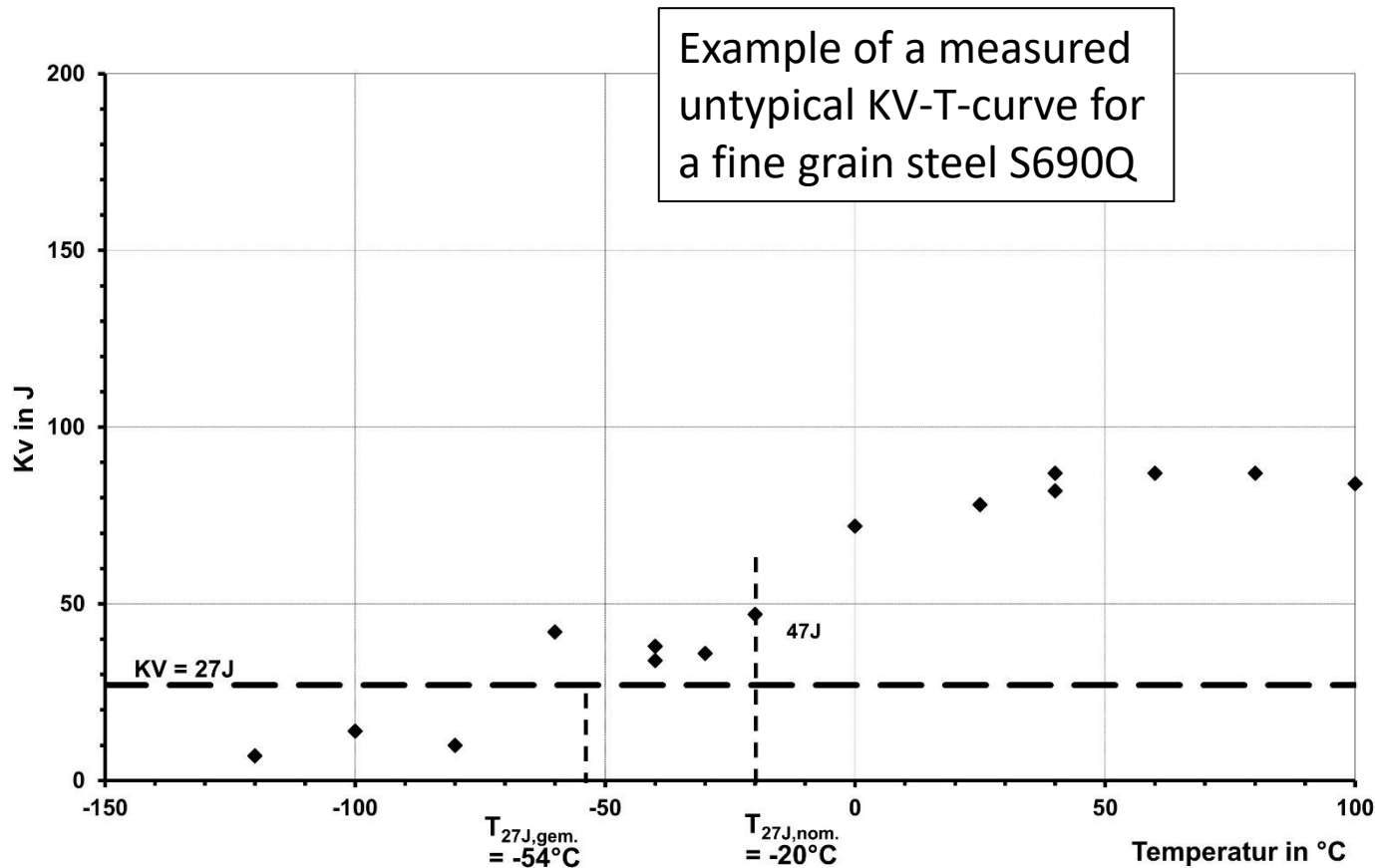


Typical example of a measured KV-T-curve for a fine grain steel S690Q

1. 2. 3. Key Innovations 4.

7. Incorporation of new upper-shelf rules

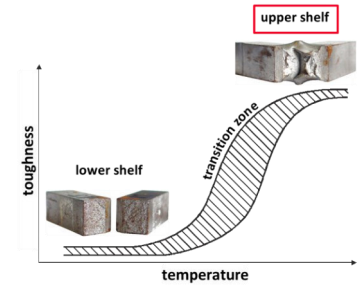
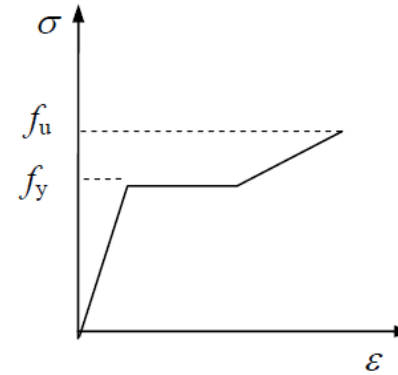
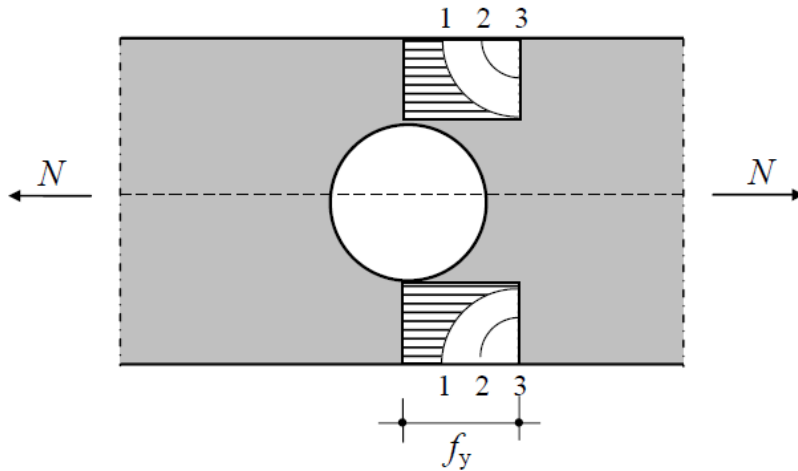
Need of such rules



1. 2. 3. Key Innovations 4.

7. Incorporation of new upper-shelf rules

Need of such rules

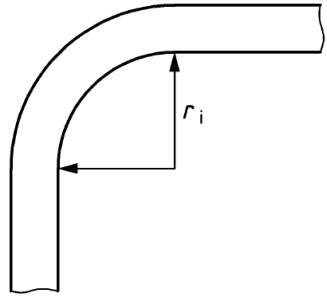
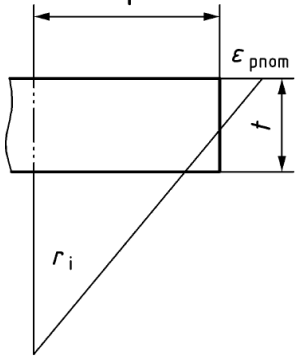
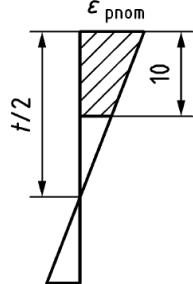


$$N_{u,2,Rd} = 0,9 \cdot A_{2,net} \cdot \frac{f_u}{\gamma_{M2}}$$

because some design rules rely on upper shelf behaviour!

8. Inclusion of hints for determining of ϵ_{pl} and ϵ_{eff} for sufficient consideration of the negative effect of cold forming

Table 4.1 — Definition of ϵ_{pnom} and ϵ_{eff} for radius r_i of cold bend

| | | |
|---|---|---|
|  |  | $\epsilon_{pnom} = \frac{t}{2r_i + t}$ |
| Determination of effective strain ϵ_{eff} | | |
| t | ϵ_{pnom} distribution | ϵ_{eff} |
| ≥ 20 |  | $\epsilon_{pnom} \left(1 - \frac{10}{t}\right)$ |

9. Simplification of the method for the selection of material to avoid lamellar tearing

Table 5.1 — Choice of quality option

| Option | Application of guidance | Note |
|--------|--|--|
| 1 | General application to all prefabricated components independently on the material and end use. | The specification of through thickness properties from EN 10164 is recommended. As an alternative, ultrasonic inspection prior to fabrication in accordance with EN 10160 is permitted. |
| 2 | Application restricted to cases of high risks associated to lamellar tearing. | Post fabrication inspection should be used to identify whether lamellar tearing has occurred in the weld zone as indicated in Figure 5.1 provided the minimum hold times after welding have been observed. |

4. Outlook

Review / refine of German and French translation

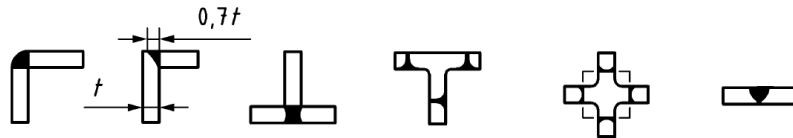
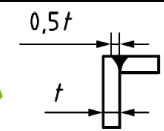
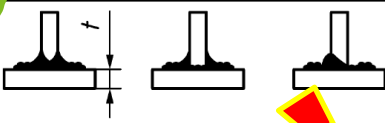
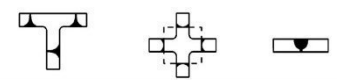
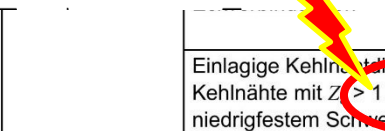


2024-10

FprEN 1993-1-10:2024

Titel de: Eurocode 3 — Bemessung und Konstruktion von Stahlbauten — Teil 1-10: Stahlsortenwahl im Hinblick auf Bruchzähigkeit und Eigenschaften in Dickenrichtung

Titel en: Eurocode 3 — Design of steel structures — Part 1-10: Material toughness and through-thickness properties

Titel fr: Eurocode 3 — Calcul des structures en acier — Partie 1-10: Ténacité du matériau et propriétés dans le sens de l'épaisseur

| | | | | | | |
|----|---|--|---|-------------|---|-------------|
| b) | Shape and position of welds in T- and cruciform- and corner-connections |  | $Z_b = -25$ | | | |
| | | corner joints |  | $Z_b = -10$ | | |
| | | single run fillet welds with $Z_a = 0$ or fillet welds with $Z_a > 1$ with buttering with low strength weld material |  | $Z_b = -5$ |  | $Z_b = -25$ |
| | | Einlagige Kehlnähte mit $Z_a = 0$ oder Kehlnähte mit $Z_a > 1$ mit Buttern mit niedrigfestem Schweißgut |  | $Z_b = -5$ |  | $Z_b = -10$ |
| | | Mehrlagige Kehlnähte |  | $Z_b = 0$ | | |

Support for the development of National Annexes

DIN EN 1993-1-10/NA:2016-04

Datum: Neu

DIN EN 1993-1-10/NA

Nationaler Anhang — National festgelegte Parameter — Eurocode 3: Bemessung und Konstruktion von Stahlbauten - Teil 1-10: Stahlsortenauswahl im Hinblick auf Bruchzähigkeit und Eigenschaften in Dickenrichtung

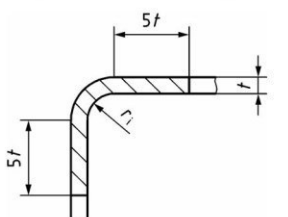
National Annex — Nationally determined parameters — Eurocode 3: Design of steel structures - Part 1-10: Material toughness and through-thickness properties

Annexe Nationale — Paramètres déterminés au plan national — Eurocode 3: Calcul des structures en acier - Partie 1-10: Choix des qualités d'acier vis à vis de la ténacité et des propriétés dans le sens de l'épaisseur

Final transmission error corrections

Table 4.6 — Conditions for welding cold-formed corners and adjacent material

| r_1/t | Strain due to cold forming [%] | Maximum thickness t [mm] | | |
|------------|--------------------------------|------------------------------|----------------------------|--|
| | | Generally | | Fully killed - / Aluminium-killed steel (Al \geq 0,02 %) |
| | | Predominantly static loading | Where fatigue predominates | |
| ≥ 25 | ≤ 2 | any | any | any |
| ≥ 10 | ≤ 5 | any | 16 | any |
| $\geq 3,0$ | ≤ 14 | 24 | 12 | 24 |
| $\geq 2,0$ | ≤ 20 | 12 | 10 | 12 |
| $\geq 1,5$ | ≤ 25 | 8 | 8 | 10 |
| $\geq 1,0$ | ≤ 33 | 4 | 4 | 6 |



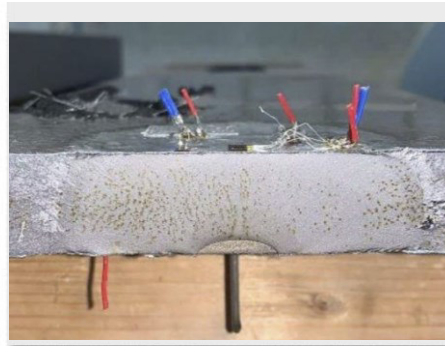
NOTE 1 For cold-formed hollow sections of steel grades up to and equal S460 which do not satisfy the inside corner-to-thickness (r_1/t) limits in Table 4.6, welding in the cold-formed corners and adjacent distances of $5t$ from the corners can be carried out if the following is satisfied:

- thickness $t \leq 12,5$ mm;
- the steel is aluminium killed;
- the steel quality is J2H, K2H, MH, MLH, NH or NLH;
- the chemical analysis meets the following limits: C \leq 0,18 %, P \leq 0,02 % and S \leq 0,012 %.

NOTE 2 Fine grained steels according to EN10219-1 can have less than 0,02% Al content, but can be considered to follow the requirements for fully killed steels.

Limitation on \leq S460
may not correct

Thank you for your attention!



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